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ARBITRARY LAGRANGIAN-EULERIAN FINITE ELEMENT ANALYSIS OF STRAIN LOCALIZATION IN TRANSIENT PROBLEMS

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SUMMARY

Nonlocal models guaranty that finite element computations on strain softening materials remain sound up to failure from a theoretical and computational viewpoint. The non locality prevents strain localization with zero global dissipation of energy, and consequently finite element calculations converge upon mesh refinements. One of the major drawbacks of these models is that the element size needed in order to capture the localization zone, must be smaller than the internal length. Hence, the total number of degrees of freedom becomes rapidly prohibitive for most engineering applications and there is an obvious need for mesh adaptivity. This paper deals with the application of the arbitrary Lagrangian-Eulerian (ALE) formulation, well known in hydrodynamics and fluid-structure interaction problems, to transient strain localization in a nonlocal damageable material. It is shown that the ALE formulation which is employed in large boundary motion problems, can also be well suited for nonlinear transient analysis of softening materials where localization bands appear. The remeshing strategy is based on the equi-distribution of an indicator that quantifies the interelement jump of a selected state variable. Two well known one-dimensional examples illustrate the capabilities of this technique: the first one deals with localization due to a propagating wave in a bar, and the second one studies the wave propagation in a hollow sphere.

Keywords: Nonlinear Computational Mechanics, Arbitrary Lagrangian-Eulerian, Mesh Adaptivity, Strain-Softening, Localization, Damage Mechanics, Wave Propagation.

INTRODUCTION

Prediction of strain localization in transient dynamics is of importance for the design and verification of concrete and reinforced concrete structures subjected to impact loads or explosions, for which the response of the structure up to failure is required. For materials like concrete, failure is triggered by localization caused by strain-softening, a feature which is necessary in order to describe progressive cracking. However, before these phenomena are modelled, theoretical and computational problems should be addressed; first, a mathematically consistent model, in agreement with the mechanical behavior of the material, must be developed for the description of the localization and the post-localization response; second, an efficient computational strategy must be used in order to capture the large local variations of the material parameters. This paper is focused on the later problems.

From a mathematical point of view, the appearance of localization in classical rate independent continuum mechanics is associated with a change of type of the governing equations. In dynamics, the equations of motion change from hyperbolic to elliptic at the inception of localization. It has been pointed out on numerous occasions [2,6] that strain-softening in classical continuum mechanics produces failure without energy dissipation (it occurs over a set of measure zero) and this is totally unrealistic. As a consequence, finite element calculations suffer from pathological mesh dependence. To overcome these difficulties, localization limiters must be implemented. Among these are the micropolar constitutive models [7], nonlocal models [6,27], gradient dependent models [8,19], and rate dependent models [29]. These constitutive relations bear the same fundamental property: an internal length, also called intrinsic characteristic length, is introduced in the constitutive equations to limit the thickness of the shear band. Actually, this implies that two well-differentiated length scales are present in the problem: one associated to the macroscopic behavior of the solid and one micro-scale related to the localization zone.

From a computational point of view, there is a basic underlying issue common to every constitutive model or resolution technique: the spatial interpolation of the primitive variables. The solution is desired at the micro-scale level: shear bands or, generally speaking, high localized gradients must be accurately described. This implies that the measure of the discretization (loosely speaking: the element size) must be in the order of, or even smaller than, the internal length in static and dynamic analyses. In the later, for instance, semi-discrete studies of harmonic wave propagation in a bar with a

uniform mesh have shown that element sizes must be one order of magnitude smaller than the internal length if errors under five percent are desired [15,16]. On the other hand, the macroscopic scale analysis of the structure is also of interest, and it should be noticed that the localization band may not be known *a priori*. Thus, adaptive remeshing strategies, in a general sense, seem the natural solution.

In most academic tests a uniform fine mesh is employed because it has the advantage that no prediction of the localization zone is needed *a priori*. The computer cost of the fine uniform mesh in real engineering problems is, however, totally prohibitive because the entire domain is discretized with elements designed for the localized area. Adaptive mesh refinement seems therefore an excellent solution [24,25,31]. It is based in reducing the element size on the localization band and has proved its effectiveness in steady-state analyses. However, it needs to modify the mesh connectivity at each iteration and the elaboration of sophisticated algorithms to preclude an increase in number of elements, algorithms which are usually restricted to triangular elements. Ortiz and Quigley [24] present an excellent discussion on the difficulties that have deterred an extensive application of the adaptive methods in the context of strain localization: path-dependent constitutive relations and error estimation relying, in statics, on the ellipticity of the equations which is lost at the inception of localization. Finally, methods based in spectral overlays [4] have the advantage of increasing the richness of the interpolation in a particular zone without the reduction in the element size. This allows higher rates of convergence but requires an *a priori* knowledge of the localization area.

In this paper, another generalized spatial interpolation technique is suggested: the arbitrary Lagrangian-Eulerian (ALE) formulation. This technique which is now well established in the fluid mechanics field, has recently been extended to nonlinear continuum mechanics in transient analyses [13] increasing the field of applicability of this formulation in the context of explicit codes. However, the ALE formulation has remained up to now restricted to the problems that naturally suggested its need: boundary motion analyses. Here, the ALE is reinterpreted as a simple numerical technique to modify the spatial interpolation introducing more degrees of freedom where they are needed.

ALE FORMULATION

The arbitrary Lagrangian-Eulerian formulation is briefly reviewed in this section. A complete presentation may be found in [13,14,20-22]. The material, spatial and referential coordinates denoted

by \mathbf{X} , \mathbf{x} and $\boldsymbol{\chi}$, are respectively related to the particles, laboratory and the kinematic reference employed. A Lagrangian formulation is employed when the reference, $\boldsymbol{\chi}$, coincides with the material particles, \mathbf{X} ; and an Eulerian formulation imposes a reference fixed in the laboratory or spatial domain. An ALE formulation allows an arbitrary motion of the reference with respect to the material and spatial coordinates, see Figure 1. This is perfectly suited for large boundary motion problems in fluid [5,9,14,17] and solid [12,13,20-22] mechanics; here the same formulation is used not to account for boundary motion but to enhance (refine) the spatial interpolation in order to capture the localization areas.

The kinematics in the ALE description that link it to the classical Lagrangian and Eulerian descriptions are designed to relate material time derivatives, with referential time derivatives and spatial gradients. The fundamental equation in ALE, originally devised in [17] is

$$\frac{\partial \psi}{\partial t} \Big|_{\mathbf{X}} = \frac{\partial \psi}{\partial t} \Big|_{\boldsymbol{\chi}} + c_i \frac{\partial \psi}{\partial x_i} \quad (1)$$

where ψ is any physical property, and \mathbf{c} is the convective velocity which relates the material, \mathbf{v} , and mesh, $\hat{\mathbf{v}}$, velocities

$$\mathbf{c} = \mathbf{v} - \hat{\mathbf{v}} \quad (2)$$

with

$$\mathbf{v} = \frac{\partial \mathbf{x}}{\partial t} \Big|_{\mathbf{X}} \quad \hat{\mathbf{v}} = \frac{\partial \mathbf{x}}{\partial t} \Big|_{\boldsymbol{\chi}} \quad (3)$$

Notice that \mathbf{v} is the unknown particle velocity but $\hat{\mathbf{v}}$ is an arbitrarily chosen mesh velocity. The remeshing technique, described in future sections, is based on an automatic procedure to compute $\hat{\mathbf{v}}$.

With an ALE formulation, *i.e.* using Eq. (1), the equilibrium equation is rewritten in strong form as

$$\rho \frac{\partial v_i}{\partial t} \Big|_{\boldsymbol{\chi}} + \rho c_j \frac{\partial v_i}{\partial x_j} = \frac{\partial \sigma_{ij}}{\partial x_j} \quad (4)$$

where ρ is the density, $\boldsymbol{\sigma}$ is the Cauchy stress tensor, no body forces are included, and standard indicial notation is adopted.

Apart from the conservation equations the constitutive law is also modified to account for the relative motion between mesh and particles. Any of the frequently employed rate type constitutive equations may be written after substitution of Eq. (1) as

$$\frac{\partial \sigma_{ij}}{\partial t} \Big|_{\mathbf{x}} + c_k \frac{\partial \sigma_{ij}}{\partial x_k} = r_{ij}(\rho, \mathbf{v}, \boldsymbol{\sigma}) \quad (5)$$

where \mathbf{r} is the usual Lagrangian rate of the Cauchy stress tensor. The integration of Eq. (5) is readily done in a Lagrangian formulation because quadrature points at which stresses are evaluated coincide with the same material points throughout the deformation. In ALE this integration is more difficult but it is important to observe that Eq. (5) is a scalar equation for each stress component. Moreover, all the “stress related variables”, *i.e.* variables or constant fields needed for the evaluation of \mathbf{r} , have an evolution similar to Eq. (5). Consequently the difficulties of implementing the ALE formulation in the context of nonlinear path dependent materials are concerned with the accurate time integration scheme for

$$\frac{\partial \tau}{\partial t} + c_j \frac{\partial \tau}{\partial x_j} = q \quad (6)$$

where τ is any stress or related variable and q is a general nonlinear source term.

A split-step algorithm [13] is used to integrate in time this last equation. First of all, a pseudo-Lagrangian stress is obtained by simply integrating the Lagrangian stress rate, q . Then, the pseudo-Lagrangian stress is convected. This step is equivalent to the interpolation or projection (between old and new meshes) needed in any adaptive mesh refinement technique. Any numerical formulation for first order hyperbolic or conservation equations may be implemented. Here a full upwind technique is used because it can be adapted to the particular nature of the stress fields (discontinuous element to element) and to the numerical constraints (explicit code), see [13]. The detailed equations are presented in next sections.

NONLOCAL DAMAGE MODEL

The localization limiter used in this study is the nonlocal damage model proposed by Pijaudier-Cabot and Bažant [25]. The stress strain relation is identical to that of a scalar continuous damage model:

$$\sigma_{ij} = (1 - D) C_{ijkl} \varepsilon_{kl} \quad (7)$$

in which σ_{ij} and ε_{ij} are the ij components of the stress and strain tensors, respectively; the scalar D is the damage variable which ranges from 0 initially to 1 when the material cannot sustain any stress; and C_{ijkl} are the components of the elastic fourth-order tensor for the undamaged material. The growth of damage is defined by a loading function f and its associated evolution equation. The loading function is:

$$f(\bar{Y}(\mathbf{x}), D) = \int_0^{\bar{Y}(\mathbf{x})} F(\mathbf{z}) d\mathbf{z} - D \quad (8)$$

where F is a function which describes the growth of damage, and $\bar{Y}(\mathbf{x})$ is the average energy release rate due to damage at point \mathbf{x} . In fact, this quantity (the average energy release rate \bar{Y}) introduces the nonlocal nature of the model because it is defined as a weighted average of the local energy release rate, Y , over the entire material domain denoted as \mathcal{V} , namely

$$\bar{Y}(\mathbf{x}) = \int_{\mathcal{V}} \Psi(\mathbf{s} - \mathbf{x}) Y(\mathbf{s}) d\mathbf{s} \quad (9)$$

with

$$Y(\mathbf{s}) = \frac{1}{2} \varepsilon_{ij}(\mathbf{s}) C_{ijkl} \varepsilon_{kl}(\mathbf{s}) \quad (10)$$

The weighting function Ψ is a normalized bell-shaped function, that is

$$\Psi(\mathbf{s} - \mathbf{x}) = \frac{e^{-\frac{\|\mathbf{s} - \mathbf{x}\|^2}{2\ell^2}}}{\int_{\mathcal{V}} e^{-\frac{\|\mathbf{z} - \mathbf{x}\|^2}{2\ell^2}} d\mathbf{z}} \quad (11)$$

where ℓ is the internal length of the nonlocal continuum which can be estimated experimentally as a function of the size of the heterogeneities in the material [2]. Finally, the evolution law of damage is prescribed as usually in non associated plasticity or damage:

$$\dot{D} = \delta \frac{\partial g}{\partial \bar{Y}} \quad (12)$$

with the Kuhn-Tucker conditions $\delta \geq 0$, $f \leq 0$, and $\delta f = 0$, where δ is the damage multiplier, f is defined in Eq. (8), and g is the evolution potential which is simply the average energy release rate, $g = \bar{Y}$, in the present model. In the applications, function F is of the form:

$$F(\mathbf{x}) = \frac{B}{[1 + B(\bar{Y}(\mathbf{x}) - Y_0)]^2} \quad (13)$$

where B and Y_0 are material parameters.

REMESHING STRATEGY

In the ALE description the mesh nodes move arbitrarily in the Lagrangian coordinate system. The mesh motion is determined once the convective velocity or the mesh velocity is chosen. In hydrodynamics and fluid-structure interaction problems the objective of the ALE formulation is to maintain regular shaped elements; consequently, the remeshing algorithms are based in geometrical considerations. In this paper, the remeshing strategy must be adequate to capture correctly the localized region, and also to introduce in it “enough” finite elements to minimize the discretization errors whose influence in the phase velocity is demonstrated in [15].

In adaptive remeshing techniques an error indicator and a mesh optimality criterion are defined. From a general point of view, error indicators could be divided into *i*) those based on the properties of the equations under study, see [1,18,32] among others, and *ii*) those purely related to the interpolation of the unknown variables, see [24-26]. This extremely simple classification is independent of their ability to quantify the error or to control the quality of the computations. On the other hand, the optimality criterion is designed to provide the desired element size. Among the several mesh optimality criteria the most usual one is based on the equidistribution of the error indicator, but others may be also used, see for instance [23].

In the context of localization problems several error indicators have been employed. Among those based in the properties of the equations, the success of the Zienkiewicz-Zhu [32] *a posteriori* error estimator initially developed for elastic problems induced at first its direct application to localization problems [31]. However, since it is based on a standard energy norm, the loss of ellipticity of the underlying equations precludes efficient results, as Ortiz and Quigley [24] demonstrated. Nevertheless, the standard energy norm or a norm based in the plastic dissipation can be used when a localization limiter is implemented, see the comparisons presented in [30].

Other authors [25,24] use interpolation-based estimators. While in [24] the variation of the velocity field (related to the displacement interpolation error bounds in Sobolev spaces) is the error indicator, in [25] the estimator used in compressible flow computations by [26] is modified for localization problems. The later estimator is based on the quadratic departure from the piecewise linear interpolation assuming exact nodal values. Therefore, a “key” variable for the problem, such as density in compressible flow, is chosen and the error estimator detects the ability of the continuous

piecewise linear interpolation to approximate it. Second derivatives of such “key” variable are needed. In [25] instead of a single variable two functionals of field variables and their derivatives are used, one related to the plastic deviatoric strain and the other related to the determinant of the acoustic tensor. In spite of the fact that both functionals do not have a continuous piecewise linear interpolation and that a numerical estimation of the second order derivatives of internal variables such as plastic strains is questionable for triangular linear elements, the results are encouraging.

The remeshing strategy in the arbitrary Lagrangian-Eulerian formulation follows a similar line of thought: first an error indicator must be defined and then a mesh optimality criterion is chosen. With respect to the error indicator, it is important to observe that in the present problem, the interest is focused on obtaining the best possible description of the damage localization zone which is characterized by propagating damage fronts, because the solution in this highly nonlinear problem is largely dependent on the damage localization zone. Therefore, it is assumed that the optimum mesh must concentrate elements in the neighborhood of sharp variations of damage. Consequently, instead of using the bounded variation of the velocity field as a suitable norm of the solution, see [24], the variation of a dependent variable, damage, is taken as the relative indicator of the error estimate. Damage is a bounded field, recall that it varies between 0 initially and 1 when the material can not sustain any stress (failure). The numerical evaluation of the mesh indicator is, in this case, extremely efficient in contrast to [25] where second derivatives of the indicator involving sophisticated schemes, such as projections or variational recoveries, are needed. On the other hand, as the other error estimators based on the interpolation errors, the indicator only provides a control of the quality of the mesh. That is, it simply detects when the mesh is too coarse and a minimum admissible element size must be given and related to the maximum value of the indicator to determine the desired element sizes in the rest of the domain. Nevertheless, the ALE formulation is perfectly suited for such relative error estimates, because it does not introduce new elements, in fact the motion of the computational frame may be also based on a relative measure of the element size.

In one dimensional problems, damage is known at the Gauss points but the mesh indicator, denoted κ , is defined at each nodal point as a function of the absolute value of the jump of damage across elements; in higher dimensions, generalizations similar to those presented in [24] could be used. That is,

$$\kappa(x_i) = a \llbracket D(x_i) \rrbracket + b \quad \text{where} \quad \llbracket D(x_i) \rrbracket = |D^{i+1} - D^i| \quad (14)$$

where a and b are two parameters, x_i is the coordinate of node i and D^i denotes the damage at element i .

In order to equi-distribute the estimator κ over the entire domain the following equation must be solved:

$$\frac{\partial}{\partial \chi} \left[\kappa(x) \frac{\partial x}{\partial \chi} \right] = 0 \quad (15)$$

It is based on the classical elliptic mapping of the coordinate lines, see [11] and it is automatically generalizable to higher dimensions. In Eq. (15) the estimator κ may be regarded as a diffusion coefficient known at each nodal point. In fact, Eqs. (14) and (15) with the known position of the domain boundary define a Dirichlet problem which has a simple physical interpretation: the error indicator κ times the element measure must be kept constant over the domain. The parameters a and b in Eq. (14) control the relative element size. It can be easily shown that the ratio of the maximum over the minimum length of the elements throughout the mesh is bounded by $1 + a/b$.

Equation (15) induces a nonlinear system which must be solved iteratively at each time step, once the error indicator is computed. This is certainly a major drawback because fast transient problems are usually solved with explicit schemes which induce a large number of time steps. The resolution of Eq. (15) must not be too much time consuming. In fact, for one dimensional problems, explicit techniques could be devised for a faster remeshing procedure. However, in multidimensions the remeshing algorithms are time consuming. On the other hand, the numerical solution of Eq. (15) can be quite efficient and easily generalizable to higher dimensions. First of all, the tolerance in the convergence scheme can be quite large (typically 1% to 10% for the relative error of the solution with the usual Euclidean norm) because the unknowns are the nodal positions and its accuracy is not extremely important. Unlike the momentum equation, it is not necessary to satisfy Eq. (15) with the highest possible accuracy. Second, it is important to notice that this is an evolutionary problem, thus the initial approximation of the solution in the nonlinear system of equations is relatively close to the final solution because the time steps are small. And third, efficient techniques such as inverse Broyden method with line search can be implemented. In all the applications, the number of iterations is relatively small: usually two and always less than five. Nevertheless, the development of an adequate explicit remeshing algorithm which will serve the same purpose is the subject of further research.

When the error indicator is only a function of damage, remeshing starts after damage has started

to grow in one element. If damage and localization occur at time steps close to each other, the inception of the localization will be controlled by the initial mesh. Indeed, since damage has not grown sufficiently, the mesh may be still too coarse. This initial deficiency of the discretization may perturb significantly the subsequent distributions of damage. In order to avoid this problem, remeshing can also be performed starting at the beginning of the computation. For this, the indicator is taken as a function of the second invariant of strain. In a one dimensional formulation it simply reduces to

$$\kappa(x_i) = a \llbracket \hat{\varepsilon}(x_i) \rrbracket + b \quad \text{where} \quad \llbracket \hat{\varepsilon}(x_i) \rrbracket = |\hat{\varepsilon}^{i+1} - \hat{\varepsilon}^i| \quad (16)$$

where $\hat{\varepsilon}^i$ is the second invariant of the strain at element i . This indicator is used up to the inception of damage when the previously defined one, Eq. (14), is activated. In the applications, both cases are implemented.

FINITE ELEMENT IMPLEMENTATION – EXAMPLES

In order to illustrate the capabilities of this computational method, two well-known problems which deal with localization due to wave propagation are considered. The first one is a rod subjected to constant velocity boundary conditions [2]. The second example deals with the propagation of a pressure wave in a spherically symmetric structure. This example was proposed by Belytschko *et al.* [6], and further studied in [19]. These two problems are one dimensional computations which are aimed at exhibiting the main properties of the continuum description and the computational strategy.

The one dimensional weak form of the momentum equation is:

$$\int_0^L \rho \delta u \frac{\partial v}{\partial t} \Big|_{\mathbf{x}} dx + \int_0^L \rho \delta u c \cdot \frac{\partial v}{\partial x} dx + \int_0^L \frac{\partial \delta u}{\partial x} \sigma dx = 0 \quad (17)$$

Where the first integral is classical, it is associated to the local increment of velocity; the second one is related to the convective effects due to the relative motion between mesh and particles; and the last one is the usual internal force term. Note that according to this equation, accelerations are the time derivatives of the particle velocity, v , holding the reference frame fixed (not the particle, nor the laboratory). However, the spatial gradient of the velocities corresponds exactly to the classical Lagrangian strain (under the small strain assumption).

An explicit central difference scheme has been used for the time integration of the momentum equation. Thus, after the finite element analysis of the previous equation, the system of algebraic ordinary differential equations to be integrated, is

$$\mathbf{M}\mathbf{a} = \mathbf{f}_{ext} - \boldsymbol{\eta}(\mathbf{v}) - \mathbf{f}_{int} \quad (18)$$

where \mathbf{M} is the mass matrix which is taken lumped, \mathbf{M}^L , in the time integration scheme; \mathbf{a} is the nodal acceleration vector (nodal values of $\left. \frac{\partial v}{\partial t} \right|_{\mathbf{x}}$); \mathbf{f}_{ext} is the vector of externally applied loads; $\boldsymbol{\eta}(\mathbf{v})$ is the convective vector, it is related to the second term of Eq. (17); and \mathbf{f}_{int} is the usual internal force vector associated to the last term in (17).

The convective term depends on the arbitrarily imposed mesh velocity and also on the unknown particle velocity: it is nonlinear. This nonlinearity which is easily overcome implementing an explicit algorithm, precludes second order accuracy of the time integration scheme for constant time steps. However, the relative influence of the convective term in Eq. (18) for all the studied cases, indicates that almost second order is achieved.

The time integration algorithm is detailed in Box 1. It shows that very simple modifications are needed in a classical explicit code to introduce the ALE formulation. Apart from the implementation of the convective term already discussed, it is necessary to evaluate the new nodal coordinates and to update the strain and damage.

In order to compute the convective velocities, the new nodal coordinates are evaluated according to the remeshing strategy. Here, this is done at each time step but remeshing can be activated whenever the user desires. The finite element integration of Eq. (15), a nonlinear Laplace equation with Dirichlet boundary conditions (the known positions of the contour nodes), induces the following nonlinear system of equations

$$\mathbf{L}(\mathbf{x}) \mathbf{x} = \mathbf{b} \quad (19)$$

where the unknowns are the new nodal positions. As cited previously, an inverse Broyden algorithm with one iteration per line search has been implemented.

Once the new mesh coordinates are obtained, the convective nodal velocities are computed

$$\mathbf{c}(t + \frac{1}{2}\Delta t) = \mathbf{v}(t + \frac{1}{2}\Delta t) - \left[\frac{\mathbf{x}(t + \Delta t) - \mathbf{x}(t)}{\Delta t} \right] \quad (20)$$

Then, the state variables, strain and damage, can be transported to the new mesh coordinates. In fact, the integration of Eq. (6) is done by a split step technique. The Lagrangian part is computed with a usual central difference scheme for strains

$$\boldsymbol{\varepsilon}(t^*) = \boldsymbol{\varepsilon}(t) + \Delta t \dot{\boldsymbol{\varepsilon}}(t + \frac{1}{2} \Delta t) \quad (21)$$

where $\dot{\boldsymbol{\varepsilon}}$ is simply the vector of spatial gradients of the particles velocities, \mathbf{v} , at the Gauss points; and t^* is related to a middle instant between t and the end of step $t + \Delta t$. In order to obtain the final values, the convection due to the mesh motion must be evaluated, see Eq. (6). On the other hand, for the damage variable the Lagrangian update is directly evaluated knowing the strain field at the desired instant and the damage field at the previous step [27].

Then, the convection part which is related to the right hand side of Eq. (6), is computed in both cases using a full upwind first order time integrator. A second order Lax-Wendroff technique can also be implemented, but the results are almost identical [13]. Both take into account the discontinuous nature of the state variables and require, for stability, the enforcement of the local Courant condition

$$\frac{c_e \Delta t}{h_e} \leq 1.0 \quad (22)$$

where c_e and h_e are the average convection velocity and the length at element e . Note that the numerical difficulties associated to the Riemann problem are drastically reduced because the convection part of Eq. (6) has a wave speed equal to the convection velocity which is usually much smaller than the stress wave speeds. The full upwind technique, which is generalizable to higher dimensions and takes into account the incoming and outgoing information along the characteristic lines, has a simple expression in the one dimensional case. Denote by $\tau_e(t)$ the state variable at the Gauss point of element e of length h_e whose nodes are identified by the local indices 1 and 2. The full upwind integration is simply

$$\tau_e(t + \Delta t) = \tau_e(t^*) - \frac{\Delta t}{h_e} \left\{ c_1 \frac{1 + \text{sign}(c_1)}{2} [\tau_e - \tau_{e-1}] + c_2 \frac{1 - \text{sign}(c_2)}{2} [\tau_{e+1} - \tau_e] \right\} \quad (23)$$

where the adjacent elements are referred by $e - 1$ and $e + 1$. Note that it is not mandatory to do first the Lagrangian integration and then transport the state variables to the new mesh. In fact, here, damage is first convected and then updated, see the algorithm in Box 1.

Bar Problem

The implementation of the nonlocal damage model in a one-dimensional finite element code is presented in [27] where the objectivity of the discretization for this constitutive model was demonstrated. Figure 2 shows the response of the material under uniaxial tension. The material constants are taken as follows: the elastic modulus E is 32 000 MPa, the poisson ratio ν is 0.2, the density ρ is 2.510^3 kg/m^3 , and the material parameters Y_0 and B are taken as $1.50 \cdot 10^{-4} \text{ MPa}$ and $9.27 \cdot 10^3 \text{ MPa}^{-1}$ respectively. This set of parameters corresponds to concrete with an average strength. The bar has a cross section of 1mm^2 and a characteristic length $\ell = 1\text{mm}$. The velocity at the extremities is such that a constant strain wave of amplitude $0.75\varepsilon^*$ (where ε^* is the strain measured at the peak stress) is propagated. These waves meet at t^* in the middle of the bar where localization occurs. Figure 3 shows a schematic problem statement.

Two bars of different lengths, 40mm and 160mm, are discretized in 9 elements of constant and uniform lengths and their results are compared with a “reference solution”. This reference is obtained with a sufficient number of elements (199 elements) of constant length and fixed throughout the calculation. These two cases are studied because the ratio between the localization width and the initial mesh size are very small, in the order of 1.6 and 0.4 respectively, while the reference case has ratios of 33 and 9. Notice that in the second case, the localization width is less than half the size of the initial discretization.

Figure 4 compares the damage profiles obtained at $t = 1.5t^*$ with the reference solution, the fixed mesh, an ALE computation with remeshing starting at t^* , and an ALE computation with remeshing from $t = 0$. The length of the bar is 40mm. When remeshing starts at t^* , the mesh indicator is a function of the jump of damage, Eq. (14). However, when remeshing starts at $t = 0$, the mesh indicator depends on the strain jump, Eq. (16), until damage starts to grow, then the indicator is defined by Eq. (14). The remeshing parameters in Eqs. (14) and (16) are $a = 13$ and $b = 1$. Compared with the reference solution the fixed mesh clearly overestimates the localization width. On the other hand, both ALE solutions are close to the reference one.

It can be noticed that the size of the element where localization starts has some influence in the results. In fact, this difference between both ALE solutions decreases and becomes unnoticeable as the number of elements is augmented. The influence of remeshing starting before or after localization

is better seen in Figure 5, where the same cases are shown at $t = 1.2t^*$. In this figure both the ALE from t^* and the fixed mesh clearly overpredict the localization width. The nodal points remain fixed if damage is constant; upon a sudden variation of the damage field, that is at the inception of localization, the finite element mesh is too coarse for a good approximation. The motion of the ALE mesh corrects this error in the following time-steps but the history-dependent constitutive equation (the rate of damage is always positive or zero) and the explicit solver for the equilibrium equation precludes a better approximation shortly after localization appears. Figure 6 presents at $t = 1.5t^*$ and for the same cases the element length along the bar corresponding to the damage profiles on Figure 4. As expected, the element size is reduced (half of the initial mesh) near the damage front, and it is augmented accordingly away from the localized area. Finally, Figure 7 shows the damage profiles for the 160mm bar. In this extreme case where the localization width is smaller than the initial element length the influence of the remeshing is clear. The ALE computation with remeshing starting at $t = 0$ captures adequately the localized area.

Spherically Symmetric Problem

The second test case deals with the propagation of a pressure wave in a hollow sphere. The internal radius of the sphere is $R_2 = 10\text{mm}$ and the external one is $R_1 = 100\text{mm}$. Figure 8 shows the problem statement. A uniform pressure is applied on the exterior surface of the hollow sphere. Due to the spherical symmetry, the stresses increase as the stress wave propagates towards the center of the sphere. When the peak strength is reached, localization occurs over an interior surface of the sphere. The material constants used for this problem are: $E = 1\text{MPa}$, $G = 1 \times 10^6\text{MPa}$, $\rho = 1\text{kg/mm}^3$. They are identical to the constants chosen in [19]. The governing equations for this problem are standard and can be found in the same reference. The constants for the evolution law of damage are $B = 2\text{MPa}^{-1}$, $Y_0 = 0.5\text{MPa}$, $\ell = 2.5\text{mm}$, and the applied external pressure is $p = 0.6\text{MPa}$.

The implementation of the nonlocal damage model in the finite element code requires the calculation of the average of the damage energy release rate \bar{Y} at every Gauss point using Eq. (9). This equation is, in fact, an integration over the entire three-dimensional domain. However, it can be simplified to a one-dimensional computation if R (distance to the center of symmetry of any Gauss point where \bar{Y} is evaluated) is sufficiently large compared to the internal length ℓ of the continuum,

then it is well approximated by:

$$\bar{Y}(R) = \int_{R_2}^{R_1} \frac{e^{-\frac{(R-r)^2}{2l^2}}}{\int_{R_2}^{R_1} e^{-\frac{(R-\xi)^2}{2l^2}} d\xi} Y(r) dr \quad (24)$$

This approximation has been used in the finite element calculations. Following the previous discussion, remeshing starts at $t = 0$ and it is related to the volumetric strain, Eq. (16), up to the inception of damage when remeshing is then controlled by damage itself, Eq. (14).

The reference solution has been obtained with a uniform mesh of 80 elements and a time step $\Delta t = 0.5s$. Localization appears at $t^* = 52s$. Figure 9 compares the distribution of damage at $t = 80s$ obtained with the reference solution and with a fixed uniform mesh of 30 elements. Notice in the reference solution that two localized areas are clearly developed and a third one is appearing. Several localized areas were expected: as the stress wave propagates to the center it is amplified, therefore when it reaches the peak stress localization is triggered; nevertheless, loading waves in this dispersive medium can still propagate after localization has started, consequently the stresses are again amplified as they pass to the center and trigger the formation of additional localized surfaces. Therefore several localization zones may appear before the wave is reflected on the interior surface at $R = R_2$. Note that the exact position of the localization zones is not known *a priori* by symmetry considerations as in the bar problem. It is important to observe that the damage distribution obtained with the coarser mesh is quite different from the reference solution. Three localization areas are clearly developed and none coincides with the positions of the reference ones. This proves the inherent difficulties of this particular problem with respect to mesh distribution.

The ALE analysis uses the same initial uniform mesh with 30 elements. Figure 10 shows a comparison between the ALE results and the reference solution at $t = 60s, 70s$ and $80s$. Although better agreement between both calculations is observed, several qualitative differences still persist. The most important one being a slight translation of the distribution of damage towards the center of the sphere. In view of [19] it is important to remark that in the present explicit time integration no artificial damping is introduced in the fixed mesh computations; therefore, in the reference solution, spurious oscillations appear from one element to the next. On the other hand, in the ALE analysis the full upwind scheme employed for the integration of the convection term, Eq. (23), introduces some

numerical damping. Therefore in the fixed mesh studies, such as the one with the refined mesh of 80 elements, the oscillations associated to high frequencies tend to trigger localization sooner, that is, farther from the center of the sphere. In order to suppress these oscillations, a stiffness proportional damping, similar to [19], is introduced only in the non ALE computations. The last term in Eq. (17) associated to the internal force is modified and the equilibrium equation is rewritten as

$$\int_0^L \rho \delta u \frac{\partial v}{\partial t} \Big|_{\mathbf{x}} dx + \int_0^L \rho \delta u c \cdot \frac{\partial v}{\partial x} dx + \int_0^L \frac{\partial \delta u}{\partial x} \sigma dx + \beta \int_0^L \frac{\partial \delta u}{\partial x} \dot{\sigma} dx = 0 \quad (25)$$

where β controls the amount of damping.

The reference solution is reevaluated with added damping ($\beta = 0.1$), as well as the analysis with the fixed 30 element mesh, the differences between both computations are similar to those shown in Figure 9. On the other hand, a very good agreement is obtained between the ALE results and the reference solution with $\beta = 0.1$. This shows the effectiveness of the ALE formulation. It is important to observe that this example is particularly demanding for the ALE formulation. Note that damage is detected, for every localized area, in coarse elements compared to the reference solution. This explains the lower maximum values of damage during the generation of the bands (in a similar manner as in the bar problem). For instance the second band in Figure 11b just appeared at $t = 66s$ (eight time-steps before the representation). Moreover, the remeshing tends to concentrate elements around the first localized area until the second band is detected. Consequently the mesh motion must be reversed to concentrate elements also on the second band which is correctly captured, see Figure 11c. Finally, Figure 11 also shows that the ALE damage distributions have a small bias towards the center of the sphere compared to the reference solution, this is due to the numerical smoothing of the sharp damage fronts introduced by the updating algorithm.

Figure 12 shows the evolution of the mesh with time. In these plots each element is represented by a rectangle whose side is proportional to the actual length. At $t = 40s$ the mesh indicator is still computed according to the volumetric strain, the ratio between the lengths of the smallest element to the largest one is approximately 3. At $t = 80s$ damage is well developed and this ratio is almost 11. The agreement between ALE and reference solutions shows that the strain and damage distributions are not distorted by the variations in the mesh indicator or in the size of the elements.

CONCLUSIONS

The Arbitrary Lagrangian-Eulerian formulation is one of the techniques that can be applied to strain localization analysis in fast-transient problems. In the ALE formulation, the mesh nodes move independently of the material particles, hence the element size can be conveniently adjusted in order to capture strain localization. This formulation has several interesting properties: the mesh connectivity and the number of degrees of freedom remain constant during the computation. Furthermore, the remeshing itself is imbedded in the equation of motion which is not the case in other adaptive methods. In fact, in this paper the ALE formulation has been used as an adaptive remeshing technique to better capture the localization area. Although only one-dimensional examples are presented, the formulation is developed to allow a further generalization to problems in two and three dimensions.

A simple remeshing strategy based in the equidistribution of the jump of material state variables (strain or damage) has been chosen in order to obtain an accurate solution. The remeshing equation reduces to a nonlinear system of equations. The resolution of these equations, in spite of the effectiveness of the algorithm and the relaxed convergence requirements, is time consuming and requires further study in the context of fast-transient codes. On the other hand the strain and damage update, necessary to account for the mesh motion, is very simple and efficient.

Two test cases presented where good approximations are achieved with meshes whose element length is initially much larger than the internal length of the material (which scales the localization zone). When remeshing starts at the beginning of the calculation, the element size is automatically adjusted in order to obtain strain localization at the correct location. Accurate capturing of the propagating damage fronts is achieved in spite of the small amount of elements used in the discretization. Moreover the results are clearly comparable with the reference solutions and always obtained with a drastic reduction in the computer time. The quality of the solution at the inception of localization depends on the mesh configuration at this specific time-step. Even though the ALE solution is not very accurate at this precise time because the mesh is too coarse, the formulation corrects this error at subsequent time steps.

In the ALE formulation, it is clear that the initial number of degrees of freedom is the important parameter that must be adjusted in order to achieve an accurate description of strain localization. If it is too low, the interpolation will be too coarse despite element size optimization. An elegant solution

would be to combine the ALE formulation with mesh adaptive techniques available in the literature when the number of degrees of freedom needs to be increased.

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FIGURE CAPTIONS

Box 1: Flow chart of the ALE finite element program.

Figure 1: Schematic representations for the Lagrangian, Eulerian and ALE formulations.

Figure 2: Concrete response under uniaxial tension.

Figure 3: Wave propagation in a bar: problem statement.

Figure 4: Wave propagation in a bar: damage profiles at $t = 1.5t^*$.

Figure 5: Wave propagation in a bar: damage profiles at $t = 1.2t^*$.

Figure 6: Wave propagation in a bar: element lengths at $t = 1.5t^*$.

Figure 7: Wave propagation in a bar: damage profiles at $t = 1.5t^*$ for the 160mm bar.

Figure 8: Wave propagation in a hollow sphere: problem statement.

Figure 9: Wave propagation in a hollow sphere: comparison of the damage profiles between the reference solution and the fixed mesh computation at $t = 80s$.

Figure 10: Wave propagation in a hollow sphere: comparison of the damage profiles between the reference solution and the ALE computation at a) $t = 60s$, b) $t = 70s$, and c) $t = 80s$.

Figure 11: Wave propagation in a hollow sphere: comparison of the damage profiles between the reference solution with damping and the ALE computation at a) $t = 60s$, b) $t = 70s$, and c) $t = 80s$.

Figure 12: Wave propagation in a hollow sphere: evolution of the mesh in the ALE solution.

Algorithm for the ALE explicit code

1.– Initialization and initial conditions

2.– Loop over time steps

At time t^n :

- a) Given ε^n , compute σ^n and D^n use Eqs. (7) to (13).
- b) Given the spatial coordinates \mathbf{x}^n , compute the internal forces and the mass matrix at t^n .
- c) From the particle, $\mathbf{v}^{n-\frac{1}{2}}$, and convective velocities, $\mathbf{c}^{n-\frac{1}{2}}$, compute the convective forces, $\boldsymbol{\eta}^{n-\frac{1}{2}}(\mathbf{v})$, at $t^{n-\frac{1}{2}}$.
- d) Solve the equation of motion with the lumped mass matrix and compute the accelerations at t^n .

$$\mathbf{M}^L \mathbf{a}^n = \mathbf{f}_{\text{ext}}^n - \boldsymbol{\eta}^{n-\frac{1}{2}}(\mathbf{v}) - \mathbf{f}_{\text{int}}^n$$

e) Compute new velocities and strains

o Loop over nodes

$$\mathbf{v}^{n+\frac{1}{2}} = \mathbf{v}^{n-\frac{1}{2}} + \Delta t \mathbf{a}^n + \mathcal{O}(\Delta t^2)$$

$$\mathbf{v}^n = \mathbf{v}^{n-\frac{1}{2}} + \frac{1}{2} \Delta t \mathbf{a}^n + \mathcal{O}(\Delta t)$$

o Loop over elements, compute $\dot{\varepsilon}^{n+\frac{1}{2}}$ as the gradient of the velocities, $\mathbf{v}^{n+\frac{1}{2}}$ at each element, then

$$\boldsymbol{\varepsilon}^* = \boldsymbol{\varepsilon}^n + \Delta t \dot{\boldsymbol{\varepsilon}}^{n+\frac{1}{2}} + \mathcal{O}(\Delta t^2)$$

f) Compute the mesh indicator, κ , using Eq. (14) or (16) from D^n or $\boldsymbol{\varepsilon}^*$.

g) Solve Eq. (19) and get new nodal coordinates, \mathbf{x}^{n+1} , then compute the convective velocities at $t^{n+\frac{1}{2}}$ using

$$\mathbf{c}^{n+\frac{1}{2}} = \mathbf{v}^{n-\frac{1}{2}} - \left(\frac{\mathbf{x}^{n+1} - \mathbf{x}^n}{\Delta t} \right) + \mathcal{O}(\Delta t^2)$$

$$\mathbf{c}^n = \mathbf{v}^n - \left(\frac{\mathbf{x}^{n+1} - \mathbf{x}^n}{\Delta t} \right) + \mathcal{O}(\Delta t)$$

h) Update strain and damage at the new nodal positions using Eq. (23).

3.– Compute the next time step, go to 2.–

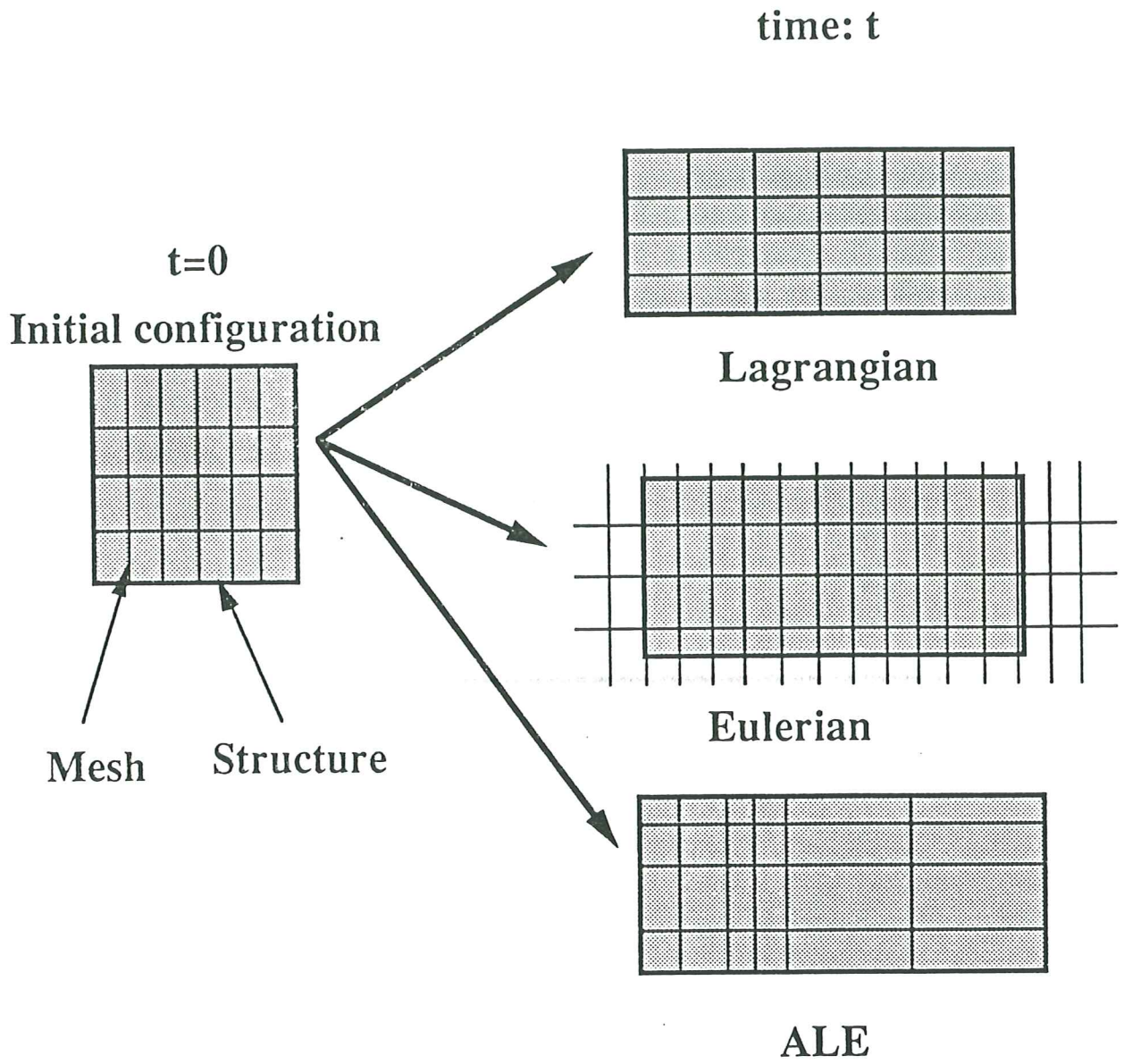


fig 1

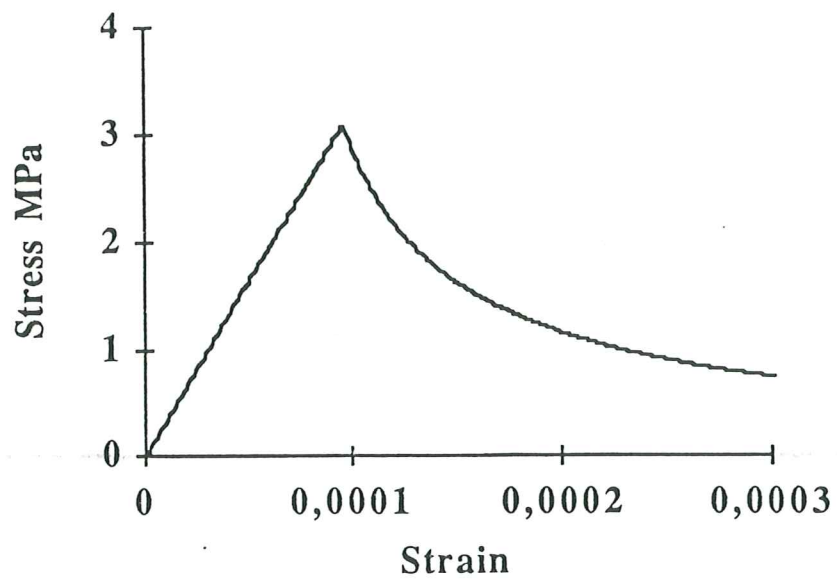


Fig 2

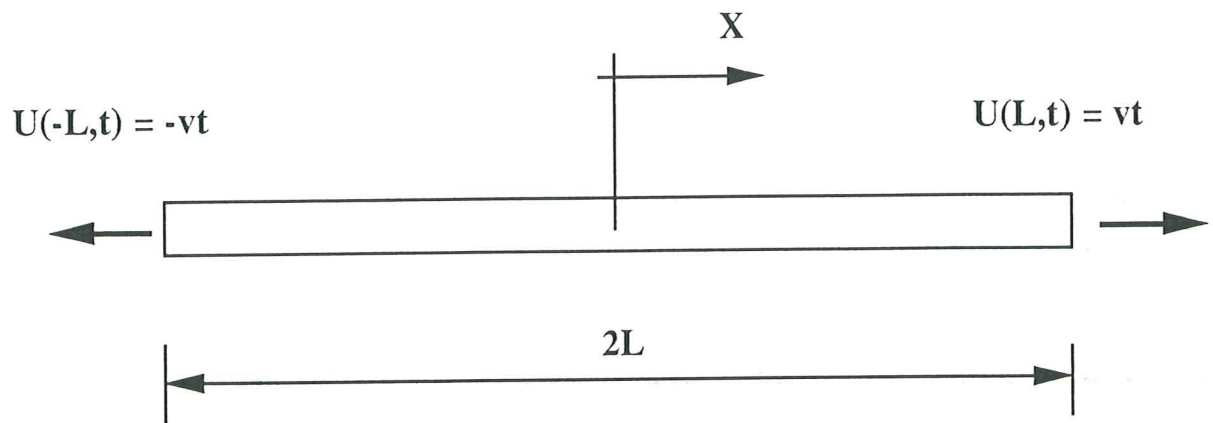


fig. 3

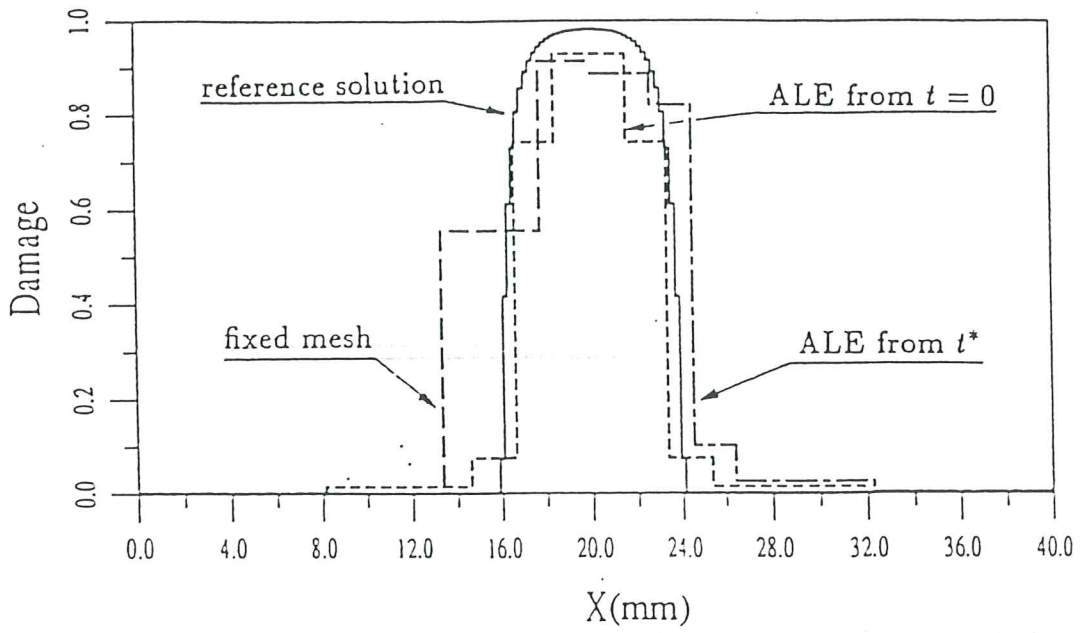


Fig 4

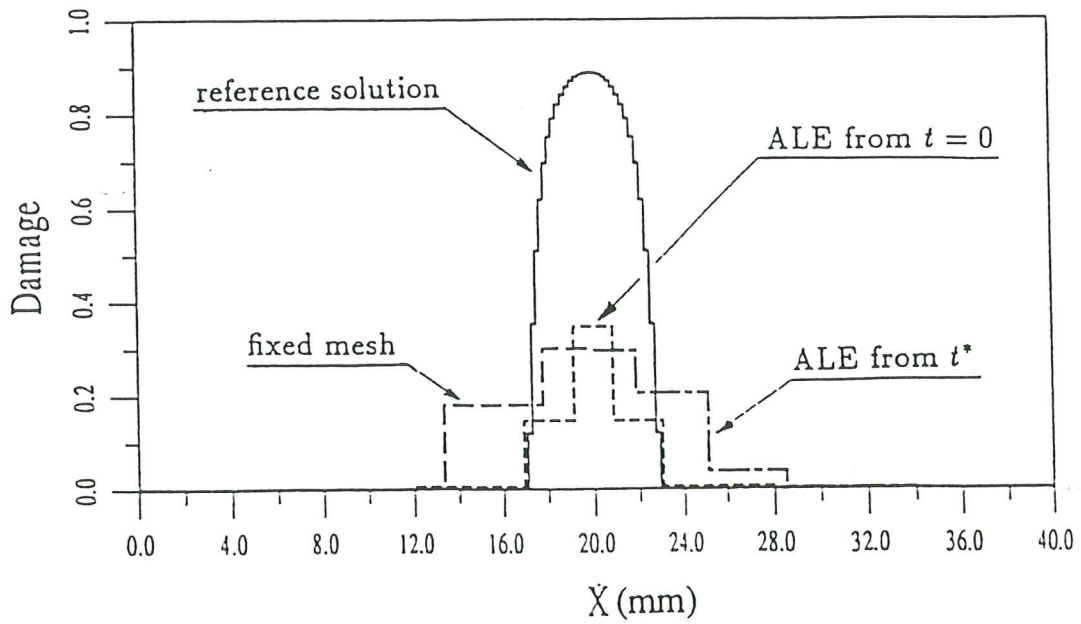


Fig 5

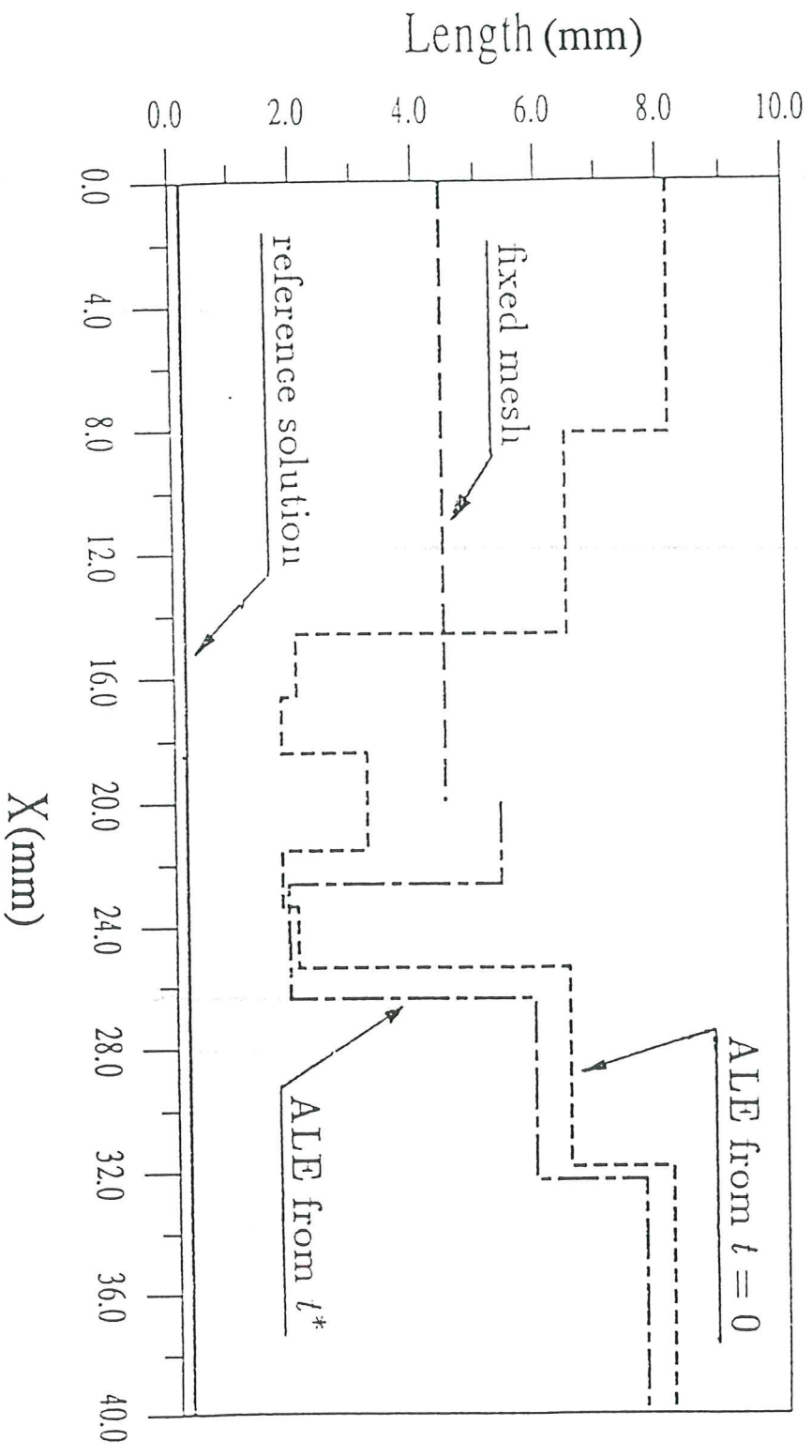


Fig. 6 : Wave Propagation in a Bar : Element Lengths at $t = 1.5 t^*$.

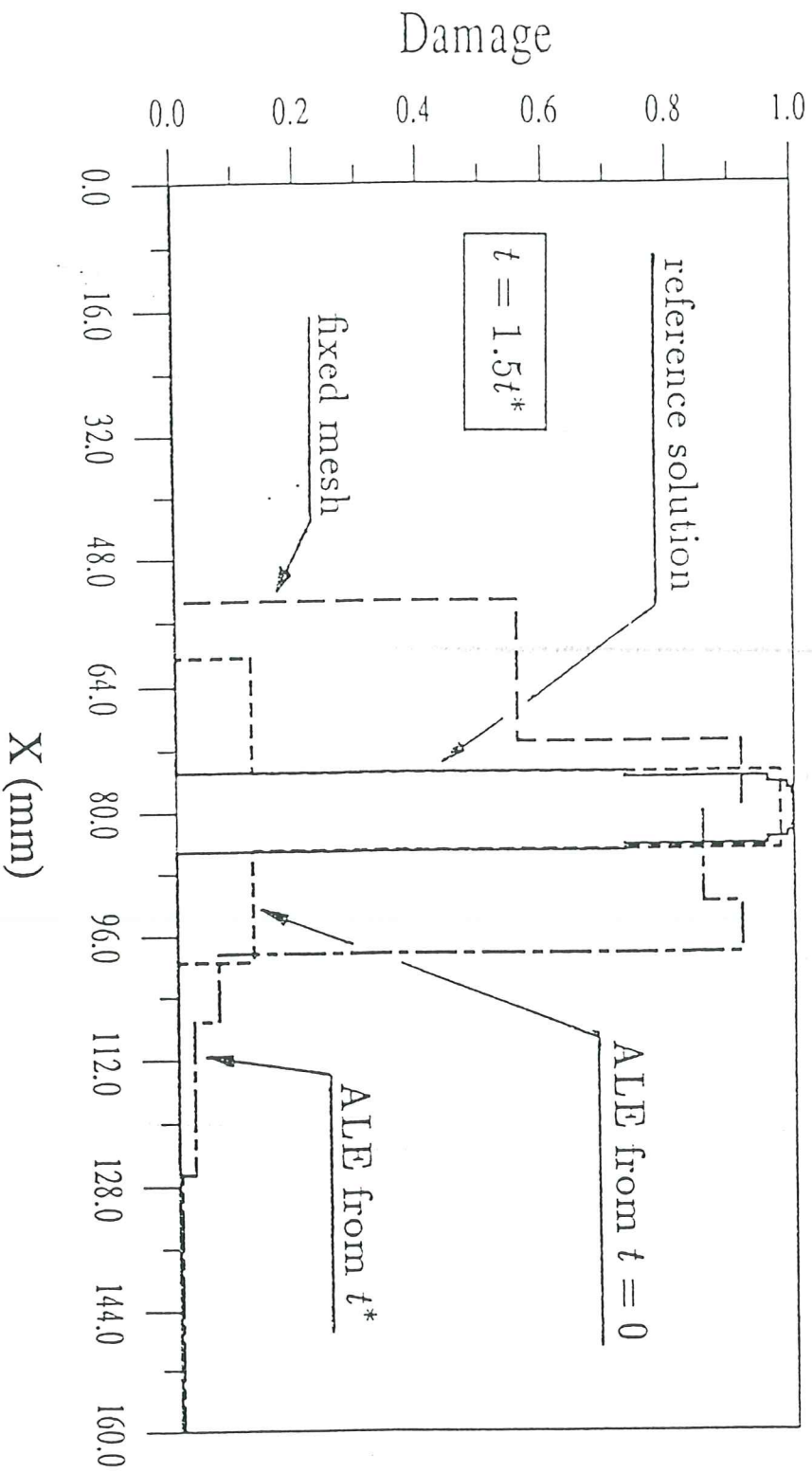


Fig. 7 : Wave Propagation in a Bar : Damage Profiles at $t = 1.5 t^*$ for the 160 mm Bar.

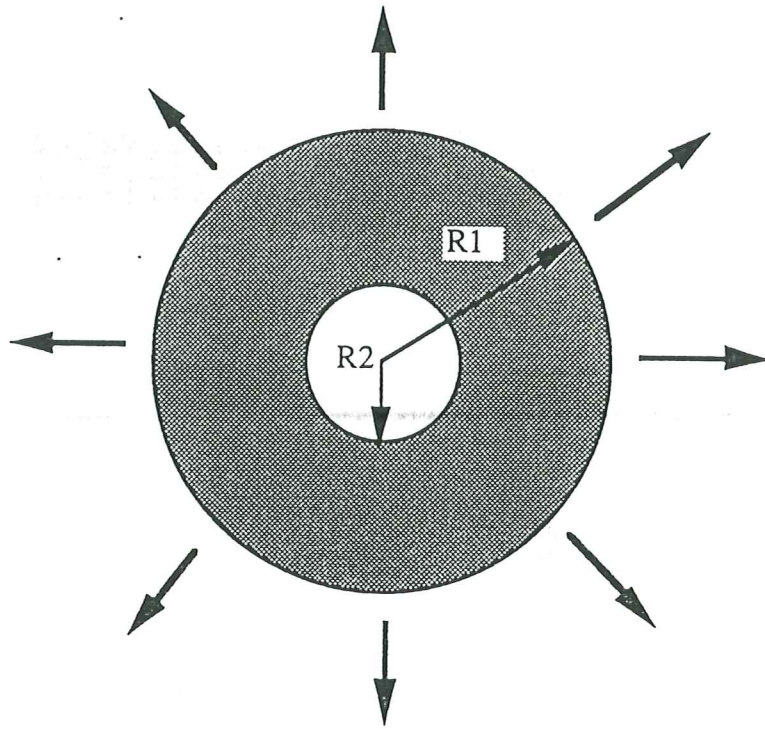


fig 8

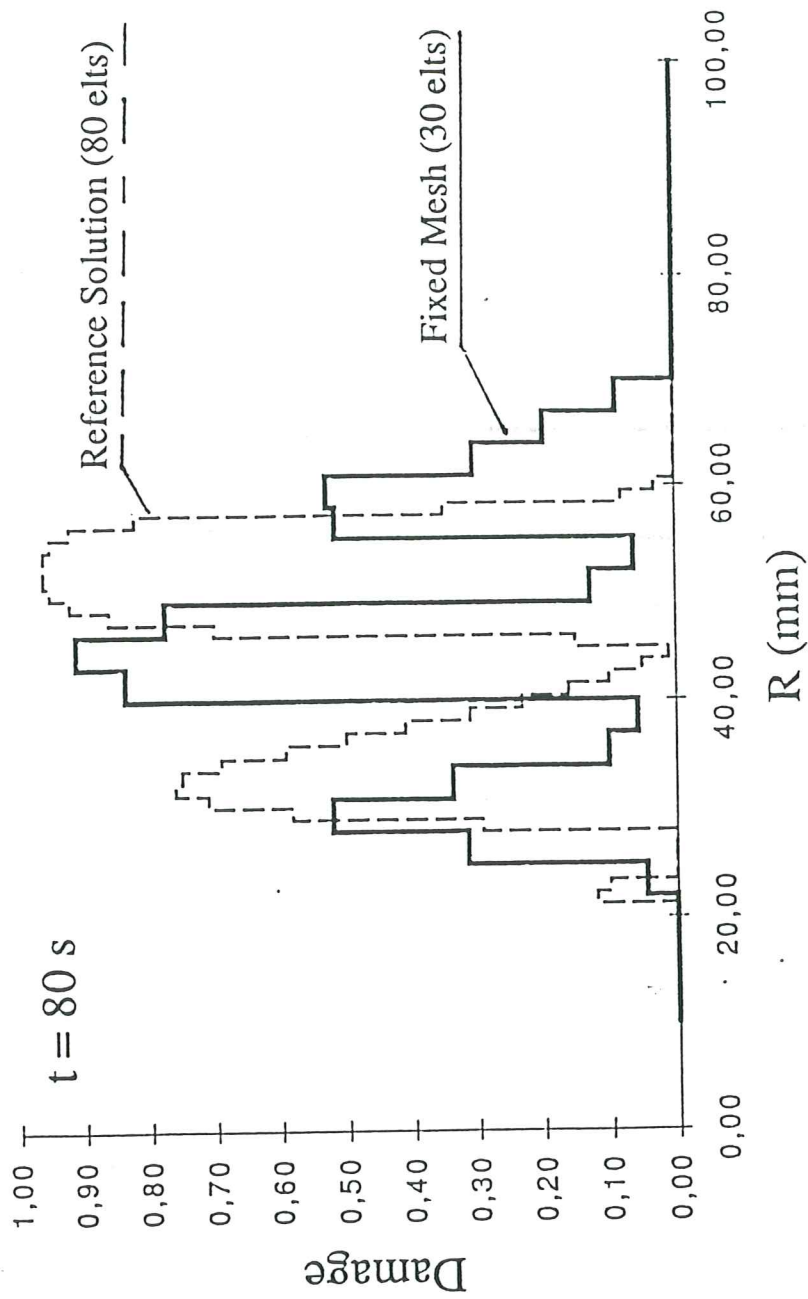


Fig. 9 : Wave Propagation in a Hollow Sphere : Comparison of the Damage Profile Between the Reference Solution (80 elements) and a Fixed Mesh Solution with 30 elements.

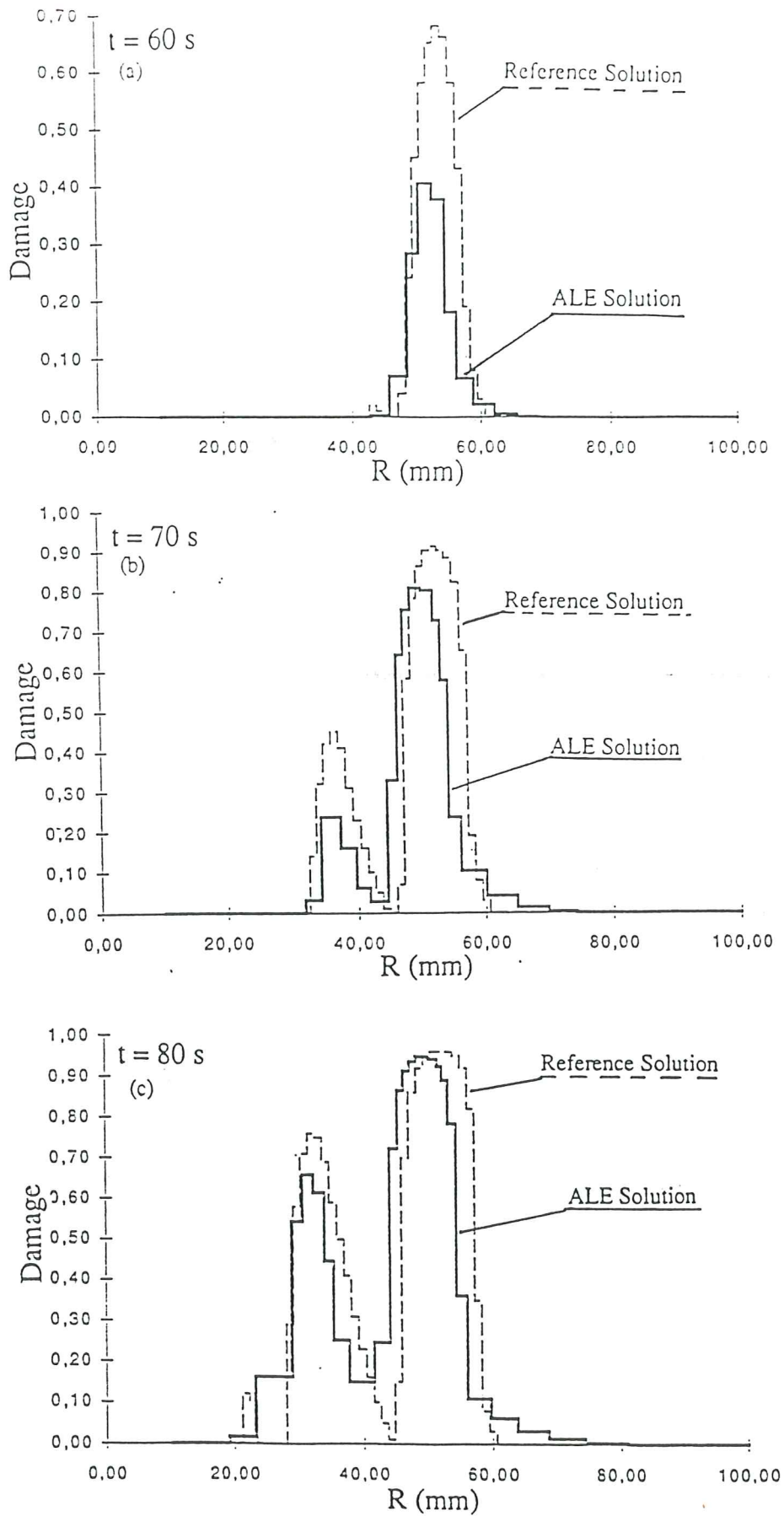


Fig. 10 : Wave Propagation in a Hollow Sphere : Comparison of the Damage Profile Between the Reference Solution (80 elements) and the ALE Solution with 30 elements at 60s (a), 70s (b), and 80s (c).

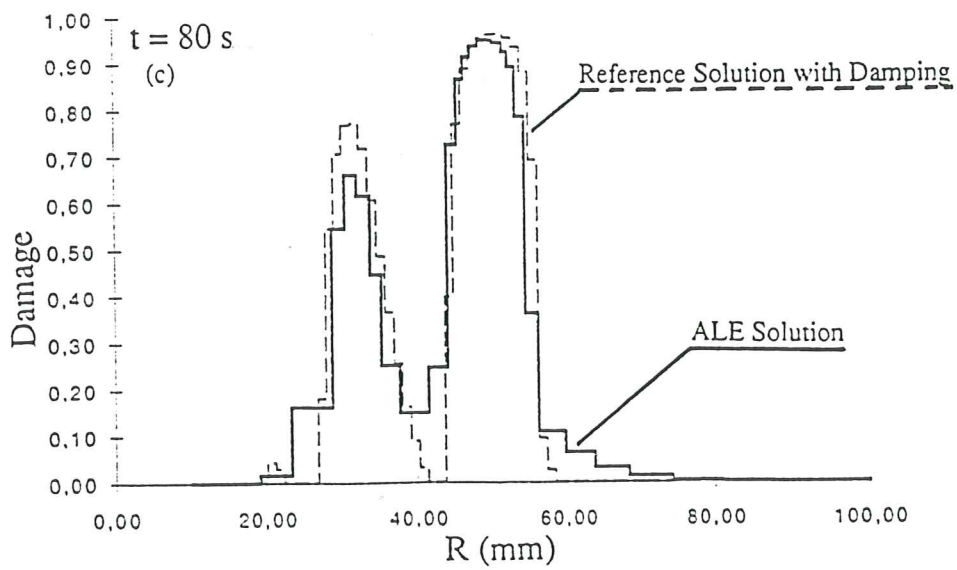
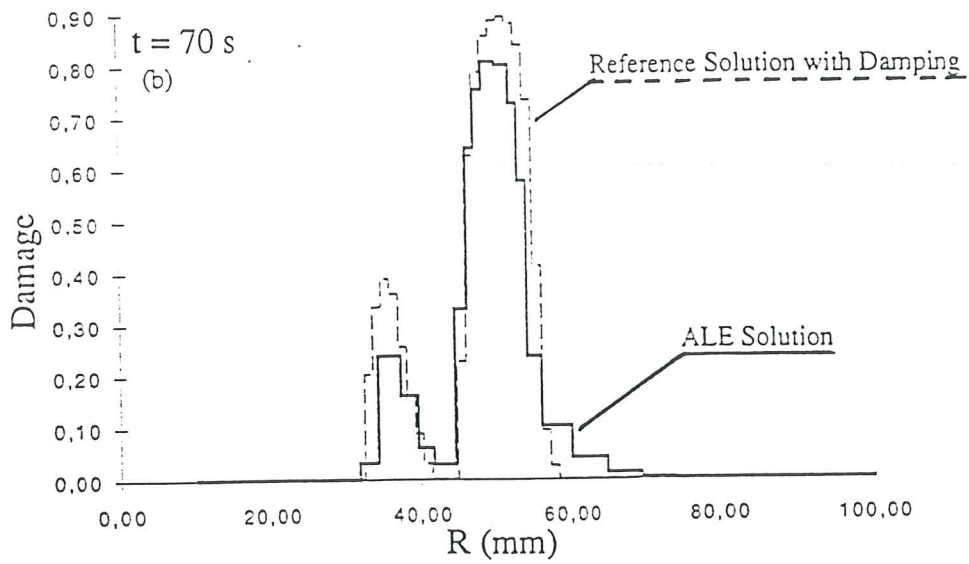
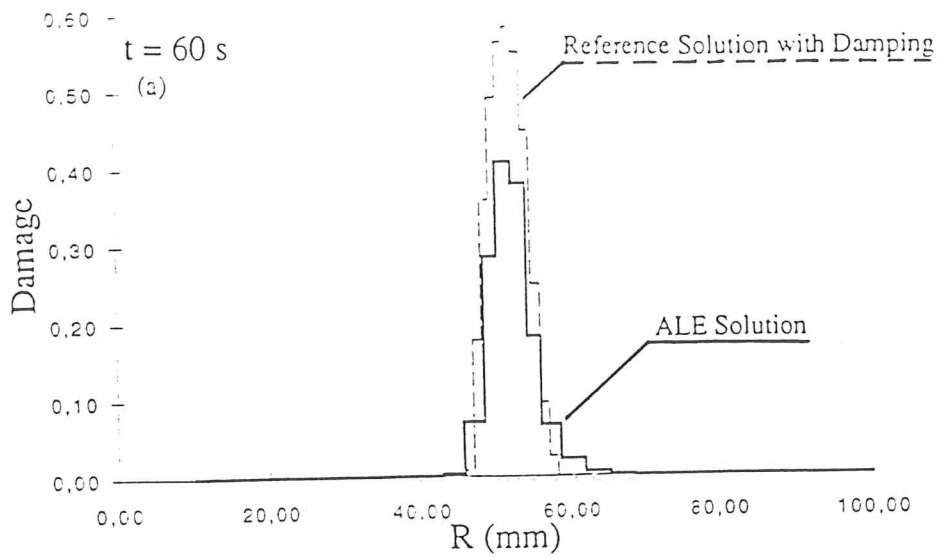


Fig 11

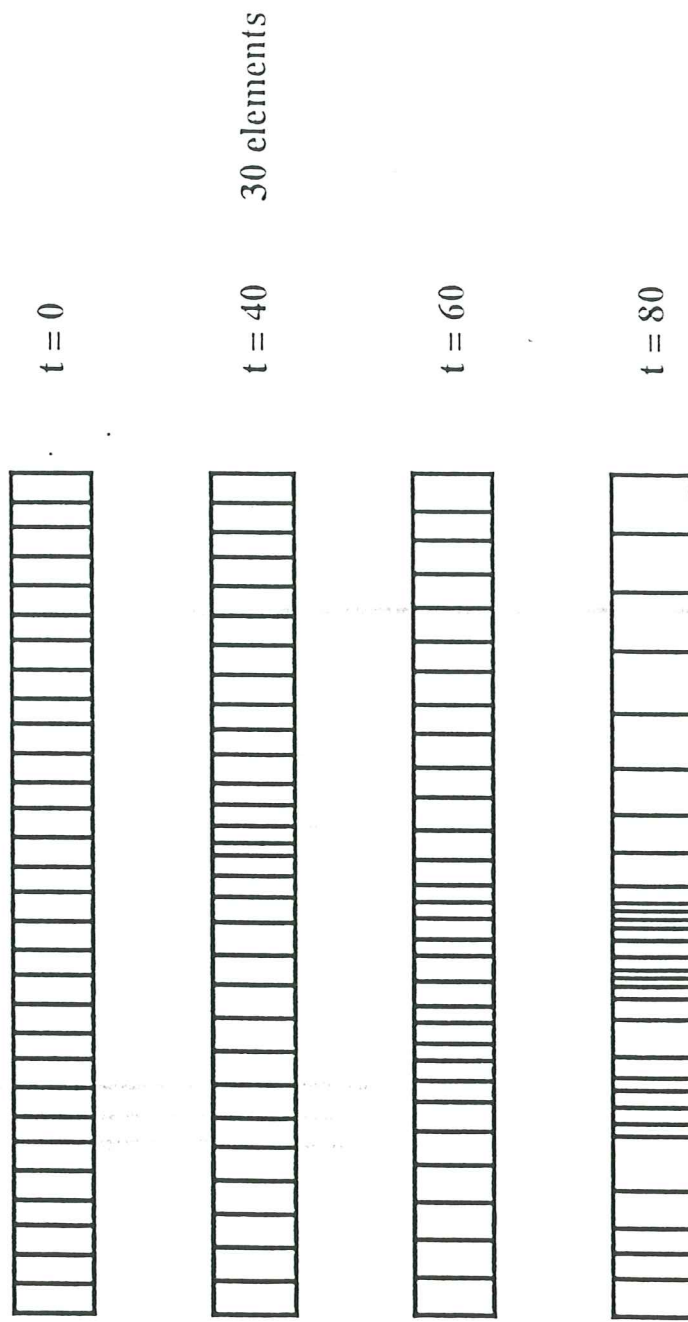


Fig 12 : Wave Propagation in a Hollow Sphere :
Evolution of the mesh in the ALE Solution.