



Validation of the particle finite element method (PFEM) for simulation of free surface flows

Validation
of the PFEM

A. Larese, R. Rossi, E. Oñate and S.R. Idelsohn

*International Center for Numerical Methods in Engineering (CIMNE),
Universidad Politécnica de Cataluña, Barcelona, Spain*

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Abstract

Purpose – The purpose of this paper is to evaluate the possibilities of the particle finite element method for simulation of free surface flows.

Design/methodology/approach – A numerical simulation of a number of examples for which experimental data are available is performed. The simulations are run using the same scale as the experiment in order to minimize errors due to scale effects. Some examples are chosen from the civil engineering field: a study of the flow over a flip bucket is analyzed for both 2D and 3D models, and the flow under a planar sluice gate is studied in 2D. Other examples, such as a 2D and 3D “dam break” with an obstacle are taken from the smooth particle hydrodynamics literature.

Findings – Different scenarios are simulated by changing the boundary conditions for reproducing flows with the desired characteristics. Different mesh sizes are considered for evaluating their influence on the final solution.

Originality/value – Details of the input data for all the examples studied are given. The aim is to identify benchmark problems for future comparisons between different numerical approaches for free surface flows.

Keywords Flow, Simulation, Fluid dynamics, Finite element analysis

Paper type Research paper

1. Introduction

The availability of sufficient computer power, together with the maturity of the tools for CFD analysis, opens the way to the simulation of flow problems of increasing complexity. Between the many practical applications, the simulation of free-surface flows represents a particularly interesting problem. The challenge is in this case connected both to the inherent difficulty in the simulation of a highly unsteady flow and to the rapid variation of the shape of the “fluid body.” This second feature is particularly demanding for the fluid simulation as it requires the constant (and automatic) redefinition of the boundary conditions.

Different methods have been devised over the years to deal with this challenge.

A first category of algorithms is based on the idea of tracking the evolution of a free surface defined with the help of a smooth distance function (level set) (Osher and Fedkiw, 2001), or of a scalar value representing the quantity of fluid in a given area.

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This is the basis of the volume of fluid (VOF) technique. This scalar function is convected according to the flow velocity field once a suitable discretization of the space is provided. This allows using existing Eulerian codes and this justifies the success of the VOF method in the CFD community. This formulation permits to deal naturally with separation (or reattachment) of parts of the fluid domain; nevertheless some concerns remain particularly on the imposition of the Dirichlet boundary conditions on the free surface. Even if all the advantages of Eulerian methods on fixed meshes can be retained, the VOF approach tends to introduce some diffusion in the position of sharp interfaces (see for examples Zalesak's circle benchmark (Osher and Fedkiw, 2003)).

An alternative formulation, known as smooth particle hydrodynamics (SPH) allows the Lagrangian simulation of a number of particles through the use of a simple meshless technology (Roubtsova and Kahawita, 2006; Bonet *et al.*, 2006). This technique, which is rising an increasing interest in the scientific community due to its simplicity and computational efficiency, faces however some severe drawbacks. As a first disadvantage it is generally difficult to construct a meshless discretization, which satisfies the partition of unit property for arbitrary distribution of points. Recent advances in the field lead however to the definition of methods like the moving least squares flavor of SPH, which allows to construct shape functions that automatically satisfy the partition of unity constraint (Brownlee *et al.*, 2007; Cola, 2002; Dilts, 1999) at the expense of some computational efficiency. Secondly, its application is appealing as long as a explicit formulation for the fluid can be used, which makes it unattractive (although possible) for truly incompressible flows.

Other approaches, like the particle in cell method widely used in computational physics, rely on the use of a background Eulerian mesh on the top of which "particles" are convected in a Lagrangian way. This method has been used in a wide range of fluid solid interaction problems: some examples of application to incompressible flows exist (Li and Liu, 2002).

The possibility exists to blend the advantages of "Particle" methods with finite element methods (FEM) in order to resolve the open issues. The PFEM (from now on) achieves this result by convecting in a Lagrangian way the fluid "particles" while redefining at the beginning of each step a new mesh (Oñate *et al.*, 2004; Idelsohn *et al.*, 2006). This allows to reproduce very accurately the convection of the nodes and to impose the Dirichlet conditions in a natural way. Further, all of the convergence results can be inherited from the FE technique which guarantees the reliability of the computational predictions (Oñate *et al.*, 2004).

The PFEM treats the mesh nodes in the fluid domain as particles which can freely move and even separate from the main fluid domain representing, for instance, the effect of water drops. A FE mesh connects the nodes defining the discretized domain where the governing equations are solved in the standard FEM fashion. The PFEM is the natural evolution of recent work of the authors for the solution of FSI problems using Lagrangian FE and meshless methods (Aubry *et al.*, 2005; Idelsohn *et al.*, 2006, 2004; Oñate *et al.*, 2004).

An obvious advantage of the Lagrangian formulation is that the convective terms disappear from the fluid equations. The difficulty is however transferred to the problem of adequately (and efficiently) moving the mesh nodes. We use an innovative mesh regeneration procedure blending elements of different shapes using an extended Delaunay tessellation (Idelsohn *et al.*, 2006, 2004).

The need to properly treat the incompressibility condition in the fluid still remains in the Lagrangian formulation. The use of standard FE interpolations may lead to a volumetric locking defect unless some precautions are taken (Donea and Huerta, 2003; Zienkiewicz *et al.*, 2005). In our work, the stabilization via a finite calculus (FIC) procedure has been chosen (Oñate, 2000). Applications of the FIC method for incompressible flow analysis using linear triangles and tetrahedra are reported in Oñate *et al.* (2004), Oñate and Idelsohn (1998) and Oñate and García (2001).

The objective of this work is to show the ability of the PFEM for reproducing real experiments comparing experimental values with numerical results. Different examples of civil engineering situation are considered. The flux over a flip bucket, the under seal flow, the dam break with an obstacle are the experiments that are described and reproduced with the PFEM. The position of the free surface and the velocity and pressure fields are the parameters chosen for the experimental-numerical comparisons. The influence of the mesh size and of the viscosity effects are also investigated in some detail.

2. Overview of the PFEM

Let us consider a domain containing both fluid and solid subdomains. The moving particles interact with the solid boundaries thereby inducing the deformation of the solid which in turn affects the flow motion making the problem is fully coupled.

In the PFEM, both the fluid and the solid domains are modeled using an updated Lagrangian formulation. That is, all variables in the fluid and solid domains are assumed to be known in the current configuration at time t . The new set of variables in both domains are sought for in the next or updated configuration at time $t + \Delta t$. The FEM is used to solve the continuum equations in both domains. Hence, a mesh discretizing these domains must be generated in order to solve the governing equations for both the fluid and solid problems in the standard FEM fashion. We note that, the nodes discretizing the fluid and solid domains are viewed as material particles which motion is tracked during the transient solution. This is useful to model the separation of fluid particles from the main fluid domain and to follow their subsequent motion as individual particles with a known density, an initial acceleration and velocity and subject to gravity forces.

It is important to note that each particle is a material point characterized by the density of the solid or fluid domain to which it belongs. The mass of a given domain is obtained by integrating the density at the different material points over the domain.

The quality of the numerical solution depends on the discretization chosen as in the standard FEM. Adaptive mesh refinement techniques can be used to improve the solution in zones where large motions of the fluid or the structure occur.

2.1 Basic steps of the PFEM

For clarity purposes we will define the collection or cloud of nodes (C) pertaining to the fluid and solid domains, the volume (V) defining the analysis domain for the fluid and the solid and the mesh (M) discretizing both domains.

A typical solution with the PFEM involves the following steps:

- (1) The starting point at each time step is the cloud of points in the fluid and solid domains. For instance, nC denotes the cloud at time $t = t_n$ (Figure 1).

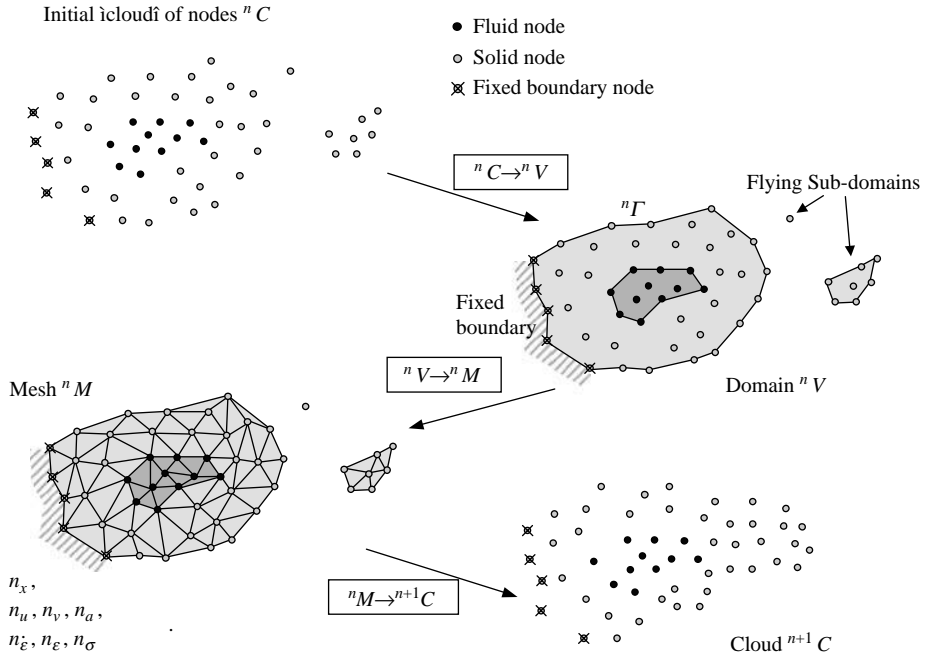
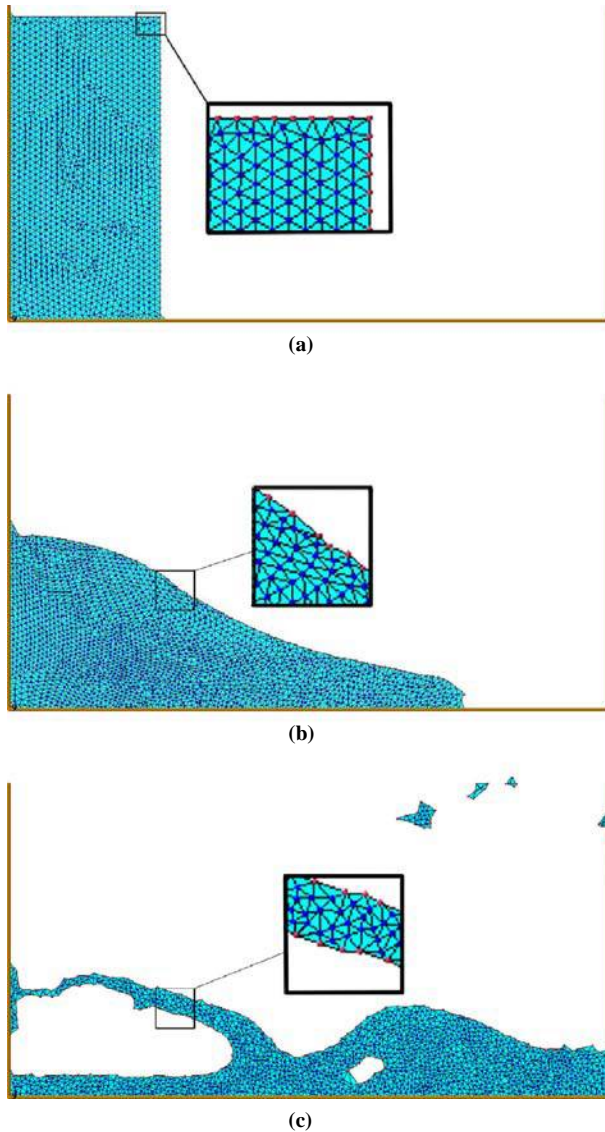


Figure 1. Sequence of steps to update a “cloud” of nodes from time n ($t = t_n$) to time $n + 1$ ($t = t_n + \Delta t$)

- (2) Identify the boundaries for both the fluid and solid domains defining the analysis domain ${}^n V$ in the fluid and the solid. This is an essential step as some boundaries (such as the free surface in fluids) may be severely distorted during the solution process including separation and re-entering of nodes. The Alpha shape method (Edelsbruner and Mücke, 1994) is used for the boundary definition.
- (3) Discretize the fluid and solid domains with a FE mesh ${}^n M$. In our work, we use an innovative mesh generation scheme based on the extended Delaunay tessellation (Idelsohn *et al.*, 2006, 2004; Oñate and Idelsohn, 1998).
- (4) Solve the coupled Lagrangian equations of motion for the fluid and the solid domains. Compute the relevant state variables in both domains at the next (updated) configuration for $t + \Delta t$: velocities, pressure and viscous stresses in the fluid and displacements, stresses and strains in the solid.
- (5) Move the mesh nodes to a new position ${}^{n+1} C$ where $n + 1$ denotes the time $t_n + \Delta t$, in terms of the time increment size. This step is typically a consequence of the solution process of Step 4.
- (6) Go back to Step 1 and repeat the solution process for the next time step.

Figure 2 shows a typical example of a PFEM solution in 2D. The pictures correspond to the analysis of the problem of breakage of a water column (Oñate *et al.*, 2004, 2006). Figure 2(a) shows the initial grid of three-noded triangles discretizing the fluid domain



Note: Figures show the evolution of the FE mesh discretizing the water domain at different instances

Figure 2.
Breakage of a water
column analyzed with the
PFEM

and the solid walls. Figure 2(a) and (c) show the mesh for the solution at two later times. Details of the PFEM can be found in Oñate *et al.* (2004), Idelsohn *et al.* (2006, 2004) and Aubry *et al.* (2005).

The following sections describe a suite of benchmark tests chosen for the experimental validation of the PFEM.

3. Flip bucket

Flip buckets are energy dissipators used at the end of ski jump spillway of large dams; their purpose is to throw the water well clear off the dam. The jet of a ski jump spillway leaves horizontally whereas the jet of a flip bucket is deflected upwards to induce disintegration in the air particles. Particular care should be taken in the construction of the dissipation pool, which is the impact zone. Moreover, the spray produced can cause damage to the surroundings and may adversely affect nearby electrical installations.

Some examples of existing flip buckets are here shown in Figure 3(a) and (b).

3.1 Experimental setting

Both 2D and 3D models are considered in order to reproduce the experimental setup developed by Juon and Hager (2000) at the Zurich University. The original aim of their investigation was to propose a simple theory for the behavior of a flux over flip buckets. This implied the derivations of fitting curves from the experimental data which are taken for comparison with the PFEM results (Figure 4).

The experiments were conducted in a rectangular 7 m long channel; its base was 0.499 m and its height was 0.7 m. It was divided into two different zones:



Figure 3.
Examples of existing
dams with flip buckets

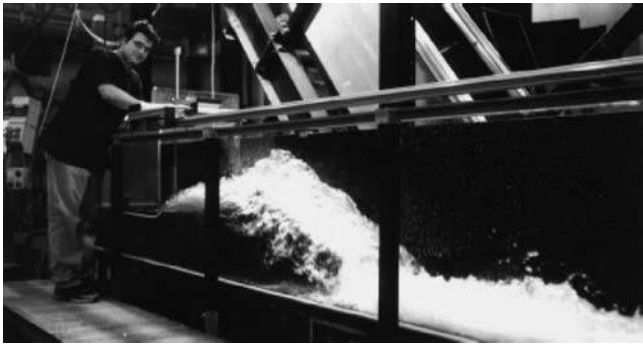


Figure 4.
Flip bucket. Experimental
set-up taken from

- (1) the upper part includes a 1 m long approach channel and the invert; and
- (2) the downstream part, thus conserving the width of the channel, simulating the dissipation pool.

The discharge was controlled by a jet-box that was regulated to obtain the designed flow velocity and flow depth h_0 ; in our work we have taken a constant value of $h_0 = 0.05$ m. R varies from 0.20 to 0.25 m and the invert angle was $\beta = 30^\circ$. The upper point of the flip bucket was placed at $w + h_s$ above the downstream channel with $h_s = 0.25$ m and $w = (1 - \cos\beta)R$ (bucket elevation) as shown in Figures 4 and 5. The free surface profile and of the upper and lower nappe profiles for the jet were identified using a point gauge to ± 0.5 and ± 0.1 mm, respectively.

The second part of the experiment included a 3D analysis: a deflector was placed in the channel. It was regulated starting from $\varphi = 0^\circ$ (the 2D case) until $\varphi = 30^\circ$. The effect of the deflector is the creation of a shock wave.

3.2 Flip bucket experimental results

Juon and Hager (2000) obtained a theoretical description of flow over flip buckets by the extrapolation of fitting functions starting from experimental data. These functions are the starting point for the validation tests. A brief overview follows of all the equations used in our work. More details on the experiment can be found in Juon and Hager (2000).

The gathering of experimental data starts from prescribing different inflow depths and discharge values. In order to avoid scale effects, a minimum depth of 4 cm is imposed. Discharges are characterized by the Froude number that can vary between 3 and 7.

On the base of Juon and Hager (2000) results, it is possible to trace a qualitative jet trajectory from the take-off point and of the pressure head distribution along the upstream channel. This is given by:

$$z = z_0 + \tan \alpha_j x - \frac{gx^2}{2V_j^2 \cos^2 \alpha_j}. \quad (1)$$

The empirical data follow well the parabola of a mass point shown in equation (1) where α_j is the take-off angle, V_j is the take-off velocity that can be considered equal to the velocity at the entry V_0 for flow conditions without scale effects (Juon and

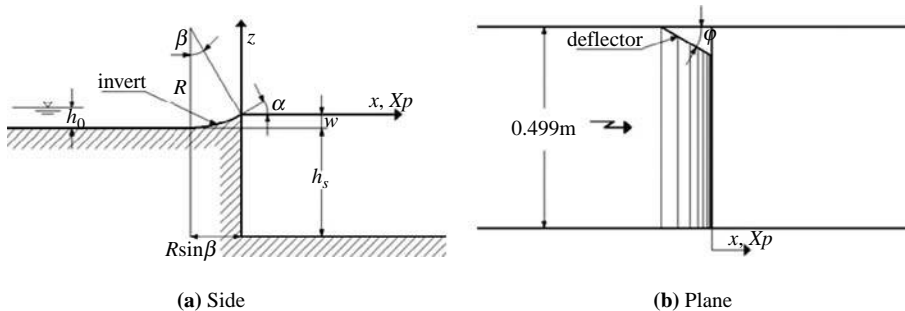


Figure 5.
Flip bucket. Schematic
representation of
experimental set-up

Hager, 2000). This equation emanates from the classical equation of parabolic motion:

$$z = z_0 + \tan \alpha_j x - \frac{1}{2} g t^2; \quad (2)$$

where, the time is computed as:

$$t = \frac{x}{V_j \cos \alpha_j};$$

where, $V_j \cos \alpha_j$ is the steady velocity along the x axis.

Figure 6 shows the comparison between experimental data in the upper and lower profile of the jet trajectory, obtained for different radii (R), inflow depth (h_0), and Froude numbers Fr and the parabola of a mass point in non-dimensional terms. The theoretical equation is in this case:

$$Z_0 = \tan \alpha_j X - \frac{1}{2} \frac{X^2}{\cos^2 \alpha_j}; \quad (3)$$

where, $X = (x/h_0 Fr_0^2)$ and $Z_0 = (z_0 - h_0)/(z_m - h_0)$ (Figure 7).

An important detail is that the take-off angle α_j is significantly smaller than the invert angle $\beta = 30^\circ$ (Heller and Hager, 2005). Their ratio can be calculated in terms of the ratio between the depth of the water h_0 and the flip bucket radius R (Heller and Hager, 2005):

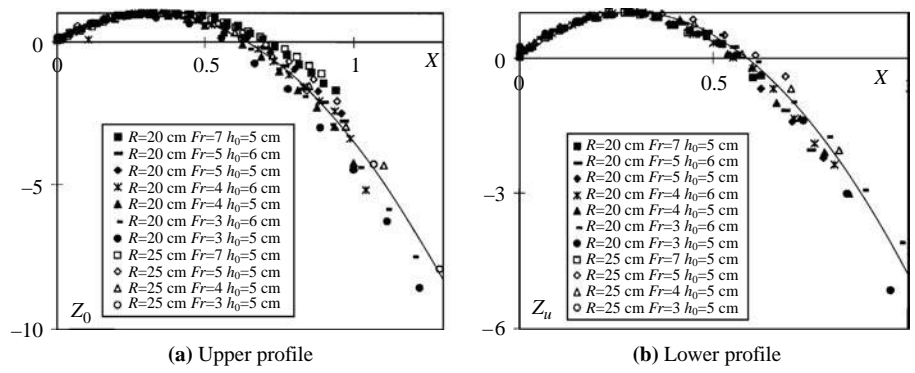


Figure 6.
Flip bucket.
Non-dimensional jet
trajectory obtained as the
average of experimental
results taken from

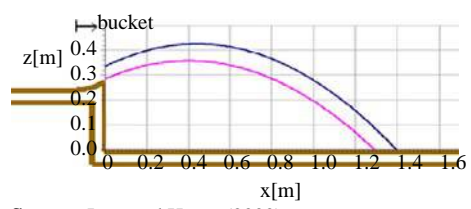


Figure 7.
Flip bucket. Jet trajectory
obtained as the average of
experimental results

Source: Juon and Hager (2000)

$$\frac{\alpha_0}{\beta} \left(\frac{70^0}{\beta} \right)^{1/6} = \frac{1}{2} \left[1 + \exp \left(-8 \left(\frac{h_0}{R} \right)^2 \right) \right]; \quad \text{for } 0 \leq \left(\frac{h_0}{R} \right). \quad (4)$$

The attention is focused on the pressure (h_P) that develops along the approach channel and along the invert. Pressure has to be constant and equal to $h_P = h_0 = 0.05$ m, and it has to be equal to the sum of a static pressure head (h_0) plus a dynamic pressure head on the flip bucket. The dynamic part has to be different than zero only in the invert. A normalized pressure parameter used for the experimental-numerical comparisons is defined as (Juon and Hager, 2000):

$$H_P = \frac{h_P - h_0}{h_{PM} - h_0};$$

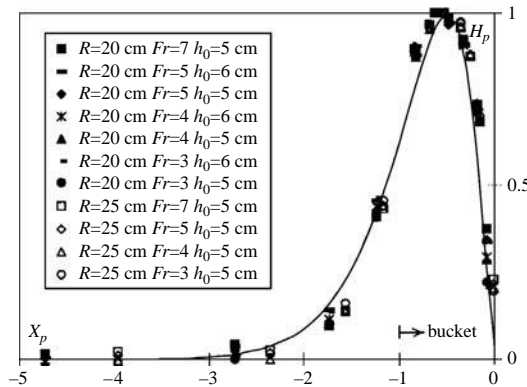
where, the abscissa origin, $x = 0$, is located at the take-off point and $R \sin \beta$ is the flip-bucket length, h_{PM} is the maximum pressure head plotted along the normalized streamline coordinate $X_P = x/R \cdot \sin \beta$ (Figure 8). This head is calculated as $h_{PM}/h_0 = (h_0/R)F_0^2$ assuming a potential vortex model. This assumption is correct if the bend number $B = (h_0/R)^{0.5}F_0 \leq 1.5$. The evolution of the experimental pressure head H_P as reported in Juon and Hager (2000) is:

$$H_P = [-2X_P \cdot \exp(1 + 2X_P)]^{2/5}, \quad (5)$$

The accuracy of the 2D computational results is analyzed considering:

- refinement of the meshes; and
- change of the Froude number (Fr) of the discharges.

In the second part of the experiment, we analyzed the effect of a deflector of variable angle placed at the bucket entry. A shock wave is generated. The highest level the jet can achieve should be twice the maximum nappe height without deflector, and its planar contraction should occupy all the channel in function of the angle.



Note: $R = 20\text{--}25$ cm; $Fr = 3\text{--}7$; $h_0 = 5\text{--}6$ cm

Source: Juon and Hager (2000)

Figure 8.
Flip bucket.
Non-dimensional pressure
curve obtained as the
average of experimental
results

The shock-wave outline was described in Juon and Hager (2000) with the profile of a standard spillway (Figure 9):

$$Z_L = \frac{A(X_L + B) \cdot \ln[A(X_L + B)]}{C} - D. \quad (6)$$

For the specific case of $A = 0.14$, $B = 2.7$, $C = 0.023$ and $D = 16$:

$$X_L = \frac{x}{h_0 F_0}; \quad (7)$$

$$Z_L = \frac{(z_L - z_{LM})}{h_0 \sin^{3/8} \varphi}. \quad (8)$$

With Z_{LM} being the maximum nappe elevation that, similarly as the maximum horizontal distance x_{LM} , depends only on the Froude number as:

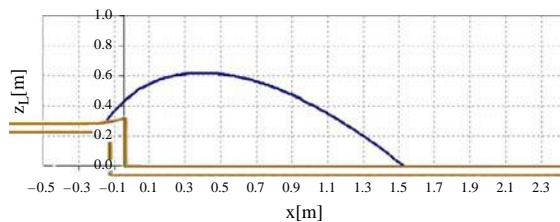
$$\frac{x_{LM}}{h_0 F_0} = 0.05 F_0^2; \quad (9)$$

$$\frac{z_{LM}}{h_0 \sin^{3/4} \varphi} = 0.45 F_0^2. \quad (10)$$

3.3 Flip bucket: PFEM results for the 2D model

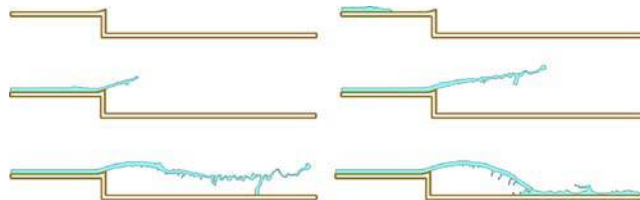
Using the GiD (2006), the pre- and post-processing system a simple 2D model was built (Figure 10). It reproduced the experimental setup. The approach channel was only 1 m long as it is proved that flip bucket effects can be considered negligible 0.5 m far away from the beginning of the invert. The downstream channel is 1.5 m for the lowest Froude numbers such as 3 and 4, and 2.5 m for $Fr = 5$ and 7.

Figure 9.
Flip bucket. Shock-wave trajectory obtained as the average of experimental results



Source: Juon and Hager (2000)

Figure 10.
Flip bucket. PFEM results for the development of the jet



A no-slip boundary condition is imposed at the walls. Water is the considered fluid for all the analyzed experiments. Its physical characteristics are:

- density: $1,000 \text{ kg/m}^3$;
- dynamic viscosity: 10^{-3} N s/m^2 ; and
- inflow velocity: variable x -component in terms of Fr (Table I).

Basically, two different meshes were considered for each value of Fr , while keeping constant the depth of the inflow discharge equal to $h_0 = 0.05 \text{ m}$ and the flip bucket radius equal to $R = 0.25 \text{ m}$. The initial element size for the first mesh is 0.01 m while for the second one is 0.005 m . The PFEM models initially have, respectively, 831 and 1,659 linear elements. A shorter model was sufficient in the case of the slowest discharges ($Fr = 3$ and 4) as the jet touched the channel after less than a meter. The initial meshes in this case have, respectively, 631 and 1,259 linear elements.

All the triangular elements are created at each time step to simulate the entry of the fluid; the number of the elements increase of some 100 percent in few seconds of simulations.

3.3.1 Jet trajectory. Few seconds of analysis are necessary to achieve a steady state. Pressure over the invert and jet trajectory are not influenced by the development of the downstream conditions.

As expected, the accuracy of the output jet trajectory improves when the mesh is refined (Figures 11-15).

Model	Fr	Q (l/s)	v_{in} (m/s)
A	3	52.41	2.1
B	4	69.88	2.8
C	5	87.35	3.5
D	7	122.29	4.9

Table I.
Flip bucket. Discharge
data for the four models

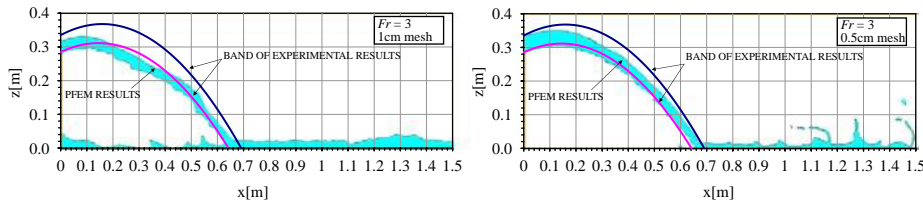


Figure 11.
Flip bucket. $Fr = 3$.
Comparison between
PFEM results and the
band of experimental
values for the jet trajectory

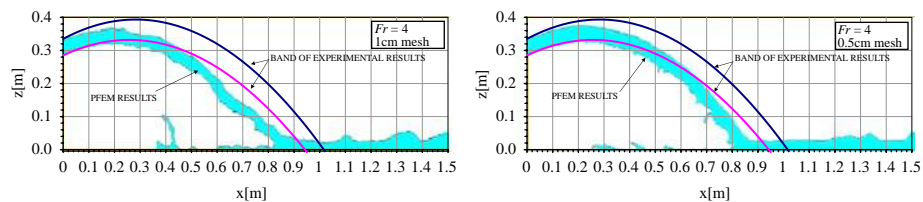


Figure 12.
Flip bucket. $Fr = 4$.
Comparison between
PFEM results and the
band of experimental
values for the jet trajectory

The accuracy of the PFEM results is also higher with the increasing for the velocity. This can be explained by the reduced importance of the viscosity effects which cannot be resolved on the coarse meshes used. The biggest discrepancy occurs for $Fr = 3$ and the 1 cm mesh as shown in Figure 16(b). Experimental results are quite different from the computational ones, once the steady state is reached. The flow seems not to have the energy to keep the jet active. The accuracy of PFEM results increases when using a finer element size. This is clearly seen in Figure 16(a) where results for the 0.5 cm mesh are shown (Table II).

3.3.2 Pressure results. Looking at the output of the bottom pressure distribution and comparing the PFEM results with the graphs interpolating the experimental data, the differences were not significant Figures 17-19. Along the approach channel, as already said, the pressure head (H_p) should be constant and equal to the depth of the flow.

Accuracy improves using a finer mesh, as expected. As for the analysis of trajectory, also in pressure case, the model with a discharge with $Fr = 3$ gave the less accurate results (Figure 17). Unfortunately, mesh refinement does not improve the solution. If the coarse mesh gives a lower level of pressure, the finer mesh overestimates the maximum pressure head by a 10 percent.

Better accuracy is achieved for a higher velocity both for the coarse and the fine meshes as shown in Figures 18 and 19. Smaller oscillations and increasing of accuracy can be noticed using a finer mesh for a fixed velocity.

Figure 13.
Flip bucket. $Fr = 5$.
Comparison between
PFEM results and the
band of experimental
values for the jet trajectory

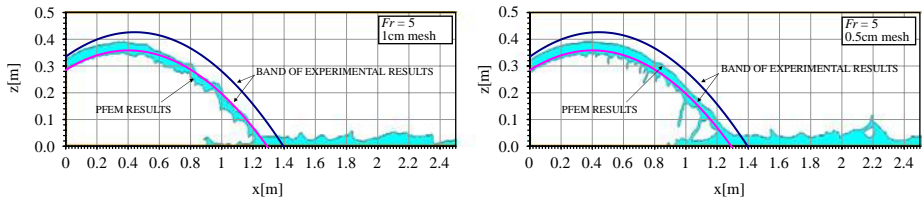


Figure 14.
Flip bucket. $Fr = 7$.
Comparison between
PFEM results and the
band of experimental
values for the jet trajectory

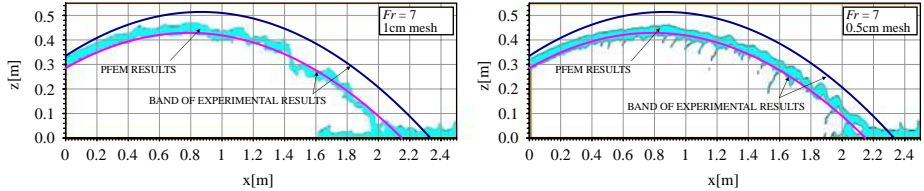
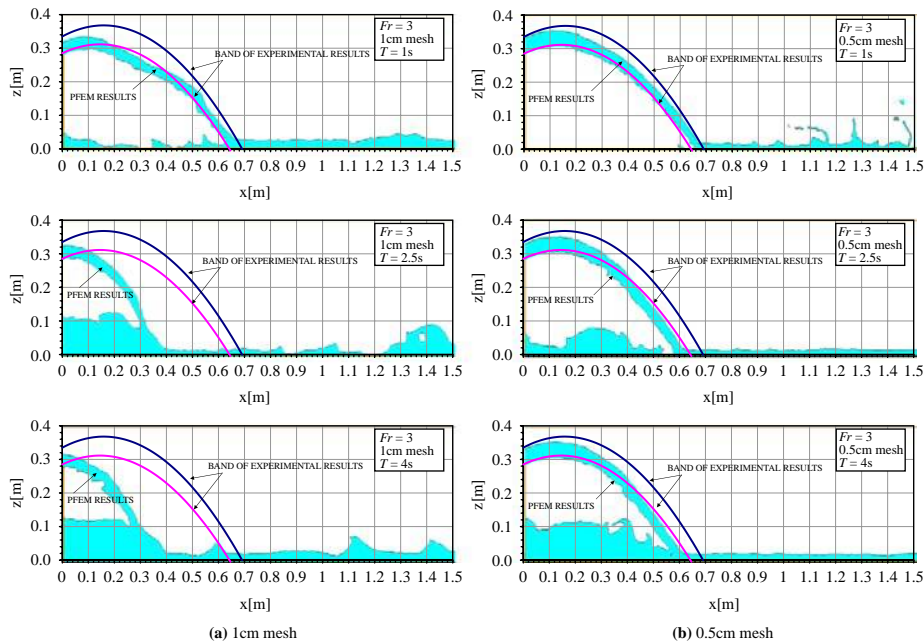


Figure 15.
Flip bucket flow.
Side view



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Figure 16.
Flip bucket. $Fr = 3$. Jet
development for different
mesh sizes

	Range of X coordinate of the jet touching point		
	Analytical solution (m)	1 cm mesh	0.5 cm mesh
$Fr = 3$	0.642-0.691	0.315-0.345 m (50 percent)	0.595-0.645 m (7 percent)
$Fr = 5$	0.642-0.691	1.292-1.396 m (10 percent)	1.276-1.395 m (0.5 percent)
$Fr = 7$	0.642-0.691	1.975-2.055 m (7 percent)	2.185-2.295 m (6 percent)

Note: Figures in parentheses represent “average error”

Table II.
Flip bucket. Accuracy of
the PFEM in reproducing
the jet impact point

3.4 Flip bucket 3D model

The second part of Juon and Hager (2000) experiment considered the effect of a variable invert angle (β can vary from 0° to 30°). This creates a restriction of the channel over the flip-bucket (Figure 20). A 3D model is built using surfaces that reproduce the geometry of the entire experimental setting (21).

The plane and side development of this wave is the output analyzed in this section. The main problem with this model is the presence of an incoming fluid. The initial mesh has some 72,000 three-noded triangular elements. This number increases after very few seconds of processing, arriving at 1.5 million tetrahedra elements. Calculation time is too long to permit the implementation of all the cases of the 2D analysis. Hence, only one model for each Froude number was built with a constant mesh dimension of $\Delta x = 0.01$ m.

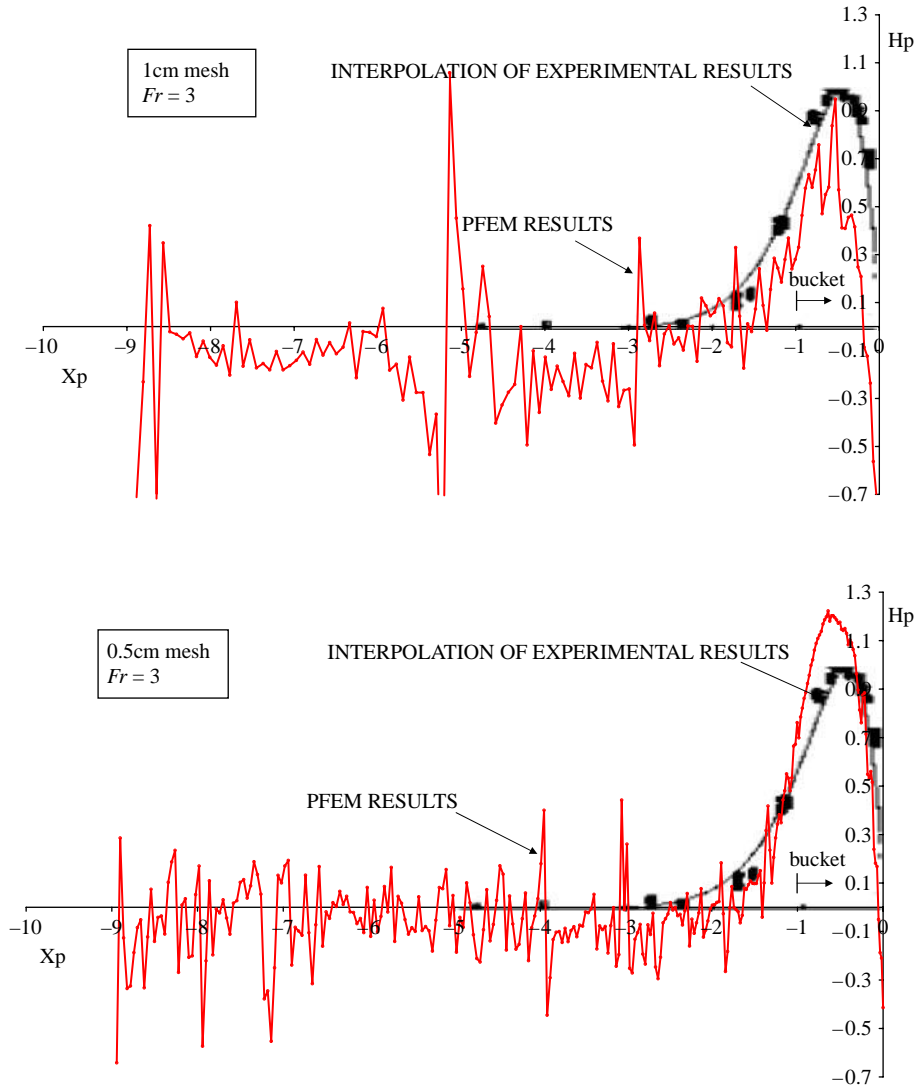


Figure 17.
Flip bucket.
Non-dimensional pressure
head: $Fr = 3$ after 4 s

Note: Reproduced from the only available original

Two sides of the shock wave were analyzed: the X - Y and the X - Z trajectory, as mentioned in Section 2. It was difficult to extrapolate clear images from the PFEM results as the development of such a wave it is not isolated from the flow but it is part of it.

Figures 20-24 show that, the side development of the shock wave is well reproduced and it also agrees well with the experimental data. It is more difficult to visualize the good simulation of the planar trajectory that only is qualitatively compared (Figure 25). In any case, the expected behavior is confirmed.

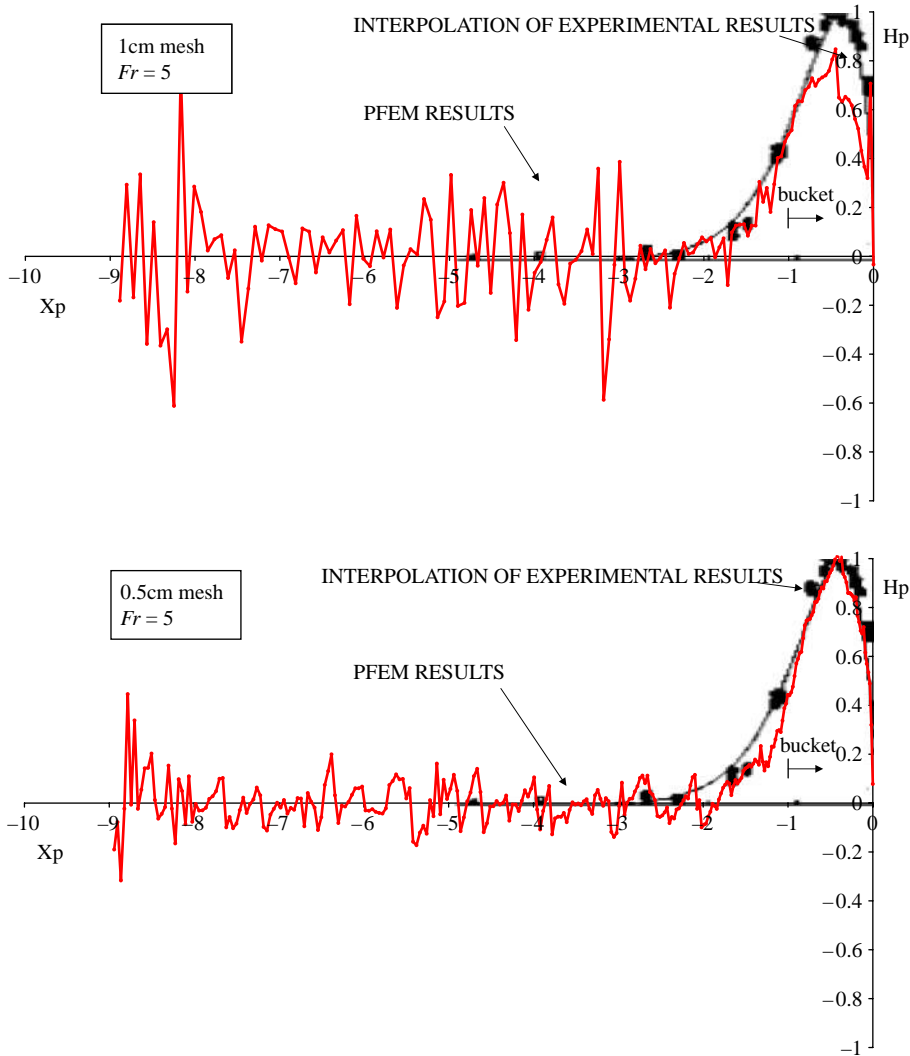


Figure 18.
Flip bucket.
Non-dimensional pressure
head: $Fr = 5$ after 4 s

Note: Reproduced from the only available original

4. Sluice gate

One of the typical and mostly used discharge regulators is a sliding gate which controls the outflow of water. Examples of this kind of structure include the gates that are at the two ends of a chamber and the gates used to garrison the discharge channel of a dam. In this case, a simple planar sluice gate is the object of the analysis.

The specific discharge of the under seal flow is governed by the classical equation:

$$q = a \cdot C_c \sqrt{2gh}; \quad (11)$$

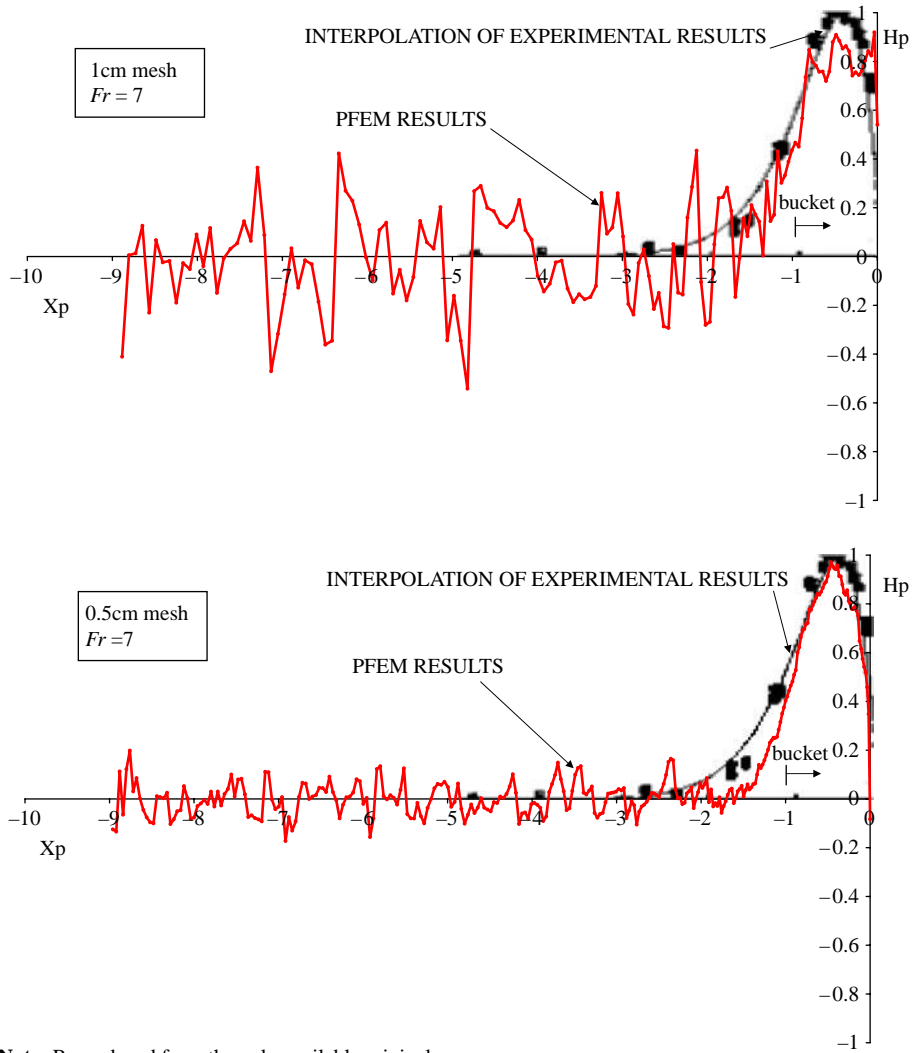


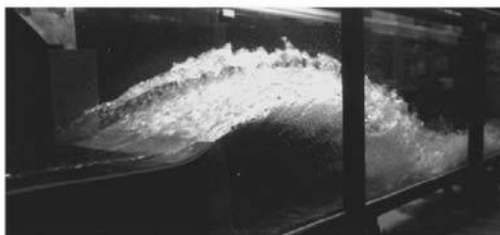
Figure 19.
Flip bucket.
Non-dimensional pressure
head: $Fr = 7$ after 4 s

Note: Reproduced from the only available original

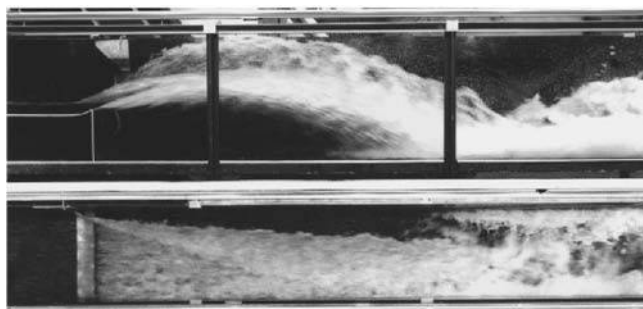
where a is the height of the sliding gate from the bottom of the channel, $C_c = 0.611$ is the contraction coefficient and h is the depth of upstream water.

The behavior of the under seal discharge and of the free surface contraction is analytically described and experimentally proved once the upstream condition and the geometrical data are given.

Next a step is placed at the end of the downstream channel in order to creating a slow discharge that, clashing with the fast under seal flow, generates an hydraulic jump.



(a) Lateral views



(b) Lateral and upper views

Figure 20.
Flip bucket. 3D
experiment

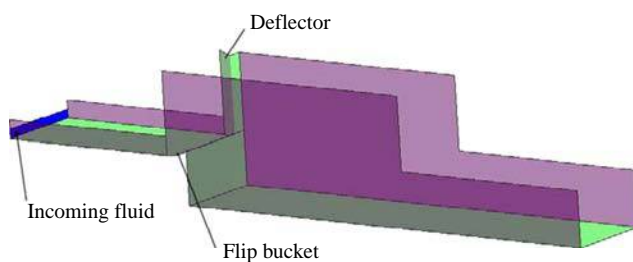


Figure 21.
3D flip bucket. Geometry

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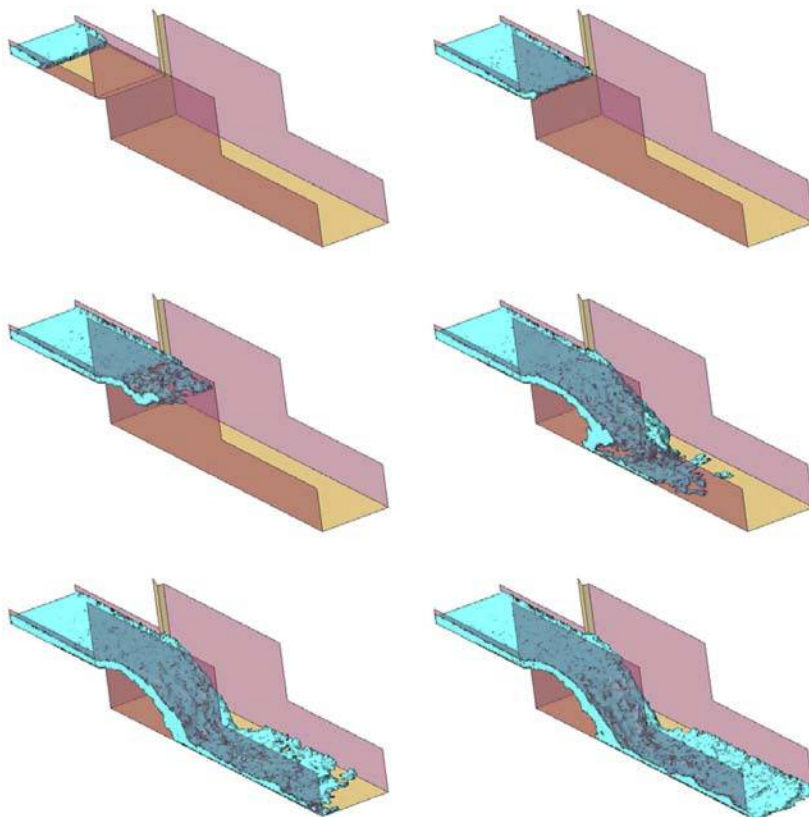


Figure 22.
3D flip bucket. PFEM
results

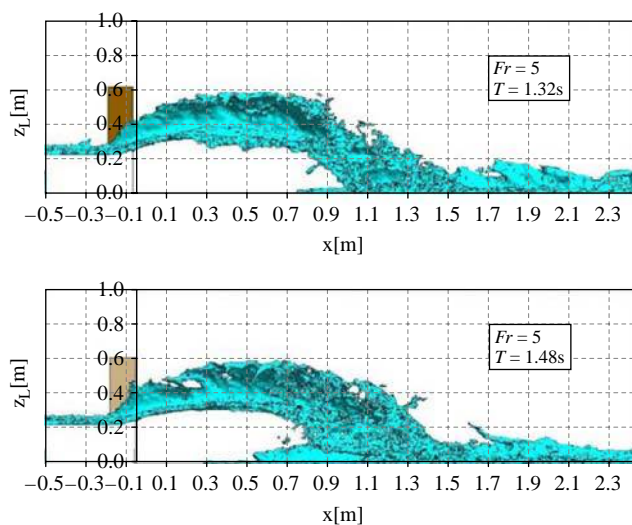


Figure 23.
3D flip bucket. Side view
of PFEM results. The
curve shows the average
of experimental values

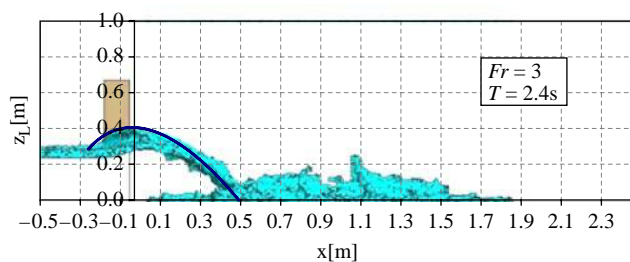
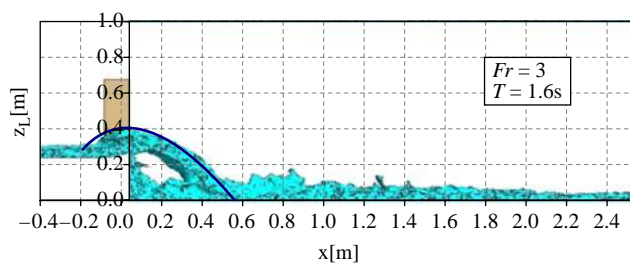


Figure 24.
3D flip bucket. Side view
of PFEM results. The
curve shows the average
of experimental values

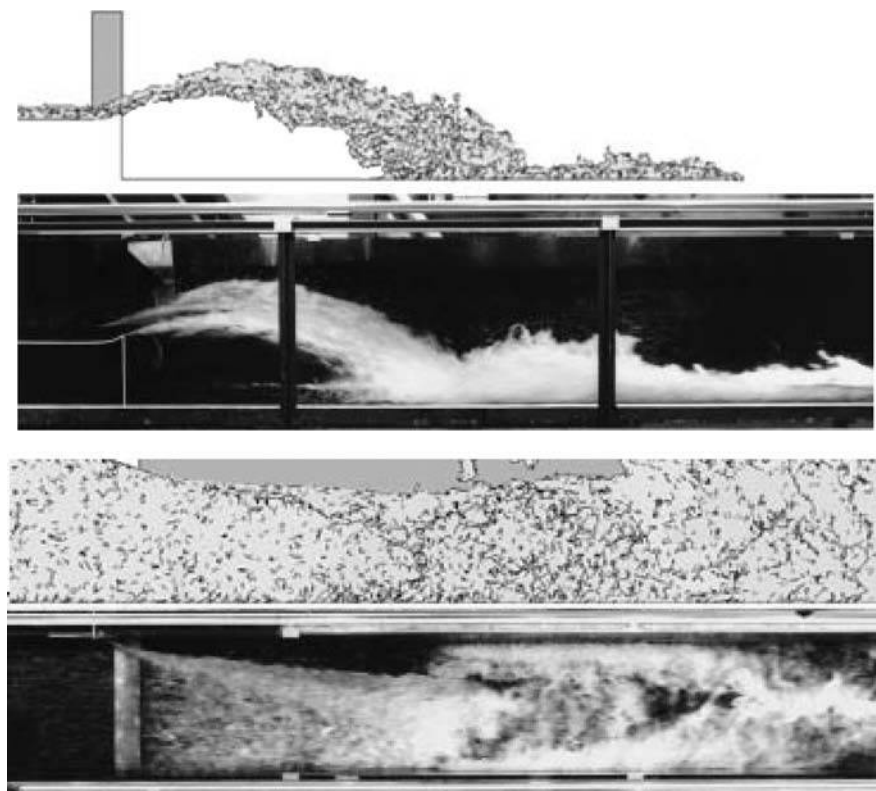


Figure 25.
3D flip bucket.
Comparison between the
PFEM results with
the experimental
photographs. Model with a
20° deflector

4.1 Experimental setting

The experimental data used for the comparison are taken from the laboratory test carried out at the Hydraulic section of the Faculty of Civil Engineering of the University of Padua, Italy (Cola, 2002).

The experimental equipment is schematically shown in Figure 26 and it is composed of a plexiglas rectangular channel, its length is 1 m and its width is 0.3 m. This channel leans on a beam that can be regulated for simulating different slopes. The inlet of the flow is controlled from the level of an upstream surge tank (V_c in Figure 26). Downstream the channel there is a second reservoir where water falls in. An electrical pump permits the passage of water from the downstream tank to the upstream one; therefore a closed circuit with constant discharge is created. At the end of the channel there is a flap blade that can be regulated for generating the downstream conditions wanted. About 16 pressure intakes are connected with some piezometers, seven of them are inserted in the sluice gate while the rest is placed in the channel bottom.

4.2 2D PFEM model

A 2D approach is sufficient to reproduce the phenomenon of the under seal flow. The kind of gate chosen can influence the contraction condition of the flow. In this case a 1 cm thin planar gate is set (Figure 27). Kirchhoff showed that if the gate is lifted up a distance which is much smaller than the gate width, the contraction coefficient C_c can be calculated as (Ghetti, 1984):

$$C_c = \frac{\pi}{2 + \pi} = 0.611.$$

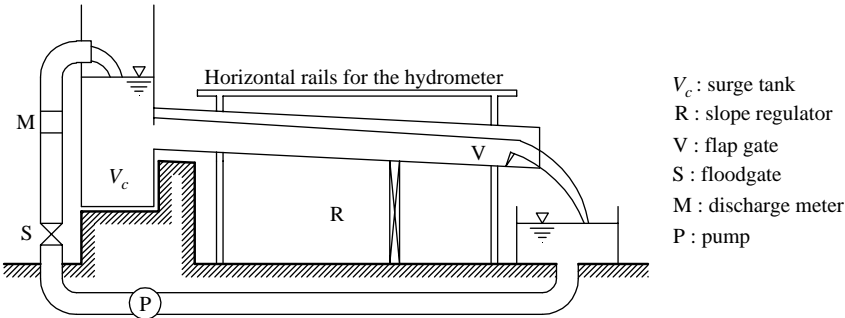


Figure 26.
Schematic representation
of the experimental device

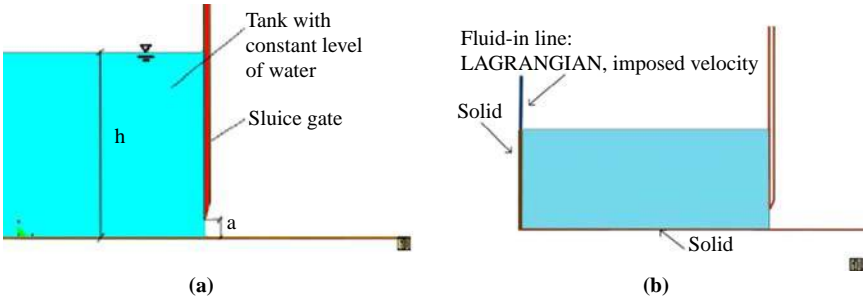


Figure 27.
Sluice gate model. A detail

Different numerical models were built for reproducing a system that represents the real setting of an upstream tank with a constant level of water. The volume of liquid in the upstream reservoir is already present at the beginning of the simulation. A balance of the inflow and the outflow in the reservoir is ensured by a continuous generation of fluid: the same discharge coming out under gravity force is brought in. The inflow is again modeled in a Lagrangian way, which originates a perturbation of the level of the reservoir. In fact, the inlet has to be inserted over the maximum level of water in the tank, otherwise the presence of fluid would made impossible the pushing forward of the entering flow (Figure 27).

The value of the discharge was one of the input data: for the first two models shown in Figure 28 it is $q = 103.3 \text{ l/sm}$ ($Q = 31 \text{ l/s}$). Once the height (y) of the inlet is set and knowing the value of the specific discharge (q) coming out from the sluice gate, the initial velocity is given by:

$$v = \frac{q}{y};$$

Models 1 and 2 (Figures 29 and 30) have been used to verify the good behavior of the under seal flow using the PFEM. Three parameters have been analyzed:

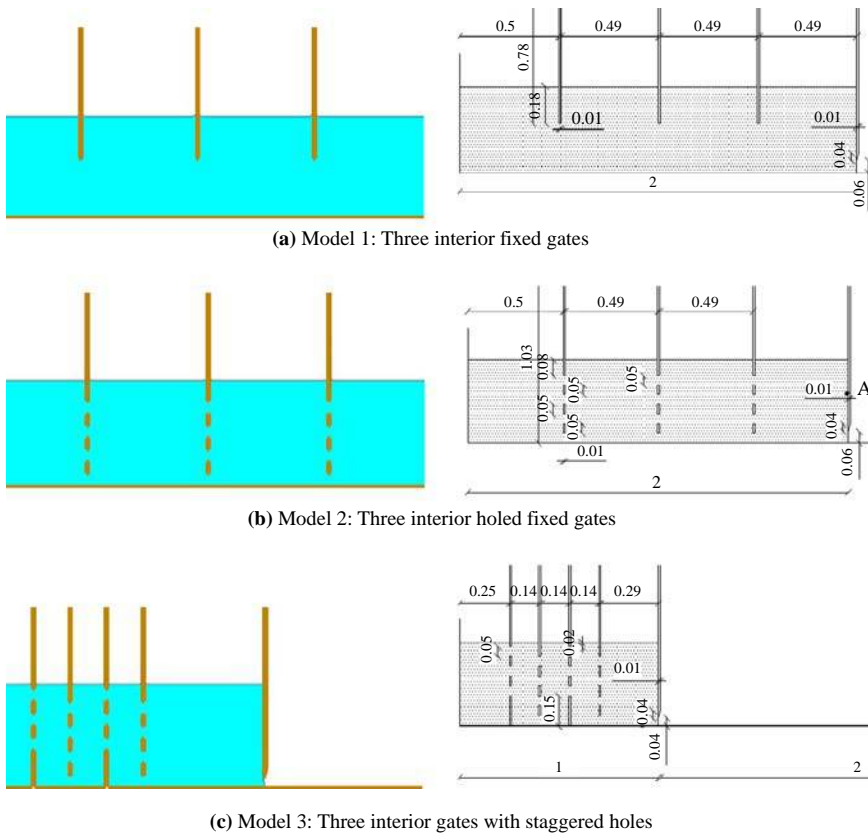


Figure 28.
Sluice gate. Geometry of
the three models analyzed

Figure 29.
Sluice gate. Model 1.
Simulation of the flow at
different instances

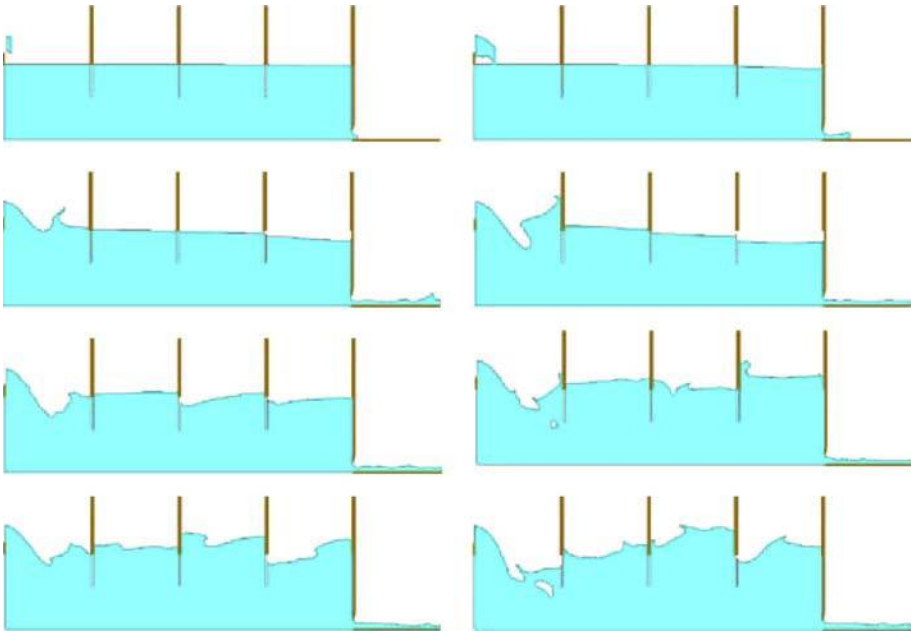
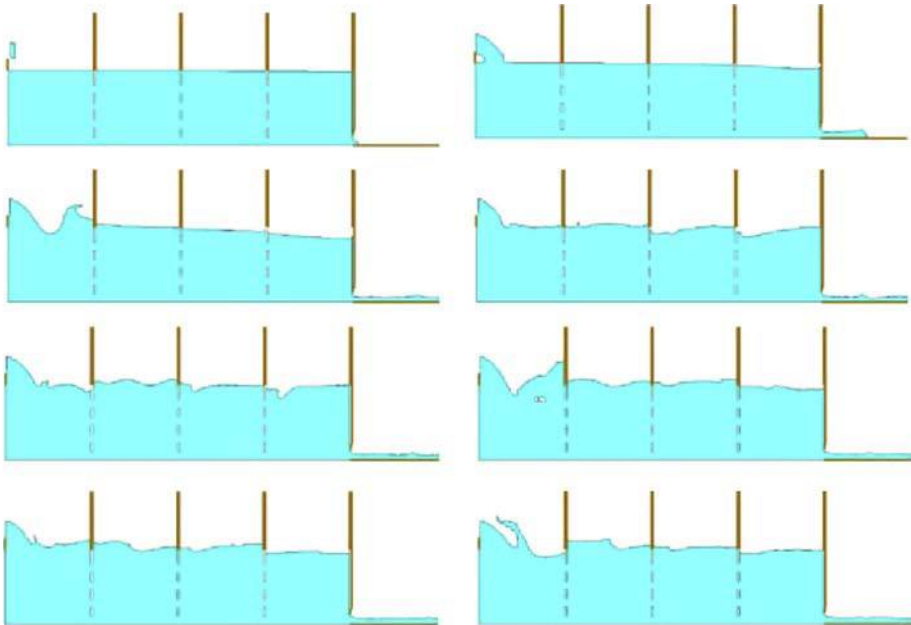


Figure 30.
Sluice gate. Model 2.
Simulation of the flow at
different instances



- (1) the pressure along the gate;
- (2) the outgoing discharge; and
- (3) the free surface of the downstream water.

The initial dimension of the mesh is $\Delta_x = 1$ cm. The depth of the water in the tank, following equation (11), is $h = 0.433$ m.

A third model (Model 3 in Figure 28) has been used to verify the possibility of reproducing a localized phenomenon such as the hydraulic jump generated by the clash of an upstream fast discharge with a downstream slow discharge. In Model 3, as during the experiment, the boundary conditions are created by the simultaneous action of the sluice gate leads to a fast discharge ($Fr^\circ \geq 1$) and of a step at the right side of the model, that generates a transition from fast to slow flow to gain energy (Ghetti, 1984). In this case, the discharge of the model is $Q = 21.1$ l/s and the sluice gate is raised a distance of $a = 4$ cm. The initial dimension of the mesh is $\Delta_x = 0.8$ cm. The depth of water in the tank, following equation (11) is $h = 0.422$ m.

4.3 The pressure along the gate

The measurements performed at the University of Padua were conducted using 7 piezometers on the sluice gate as shown in Figure 31. The value of the capillary migration was taken as 4.43 mm. This was calculated using Jurin equation (Ghetti, 1984).

Assuming an irrotational fluid, the energy which is present into the stream tube can be considered constant. This means that because of the growing of velocity in the zone near the opening of the sluice gate, the pressure head has to drop drastically, compared with the hydrostatic value.

Figures 32 and 33 show the comparison between the experimental results for the pressure head and the computational output, for different instances for Models 1 and 2 shown in Figure 28 in Section 4.2 (Table III).



Figure 31.
Sluice gate. Piezometers

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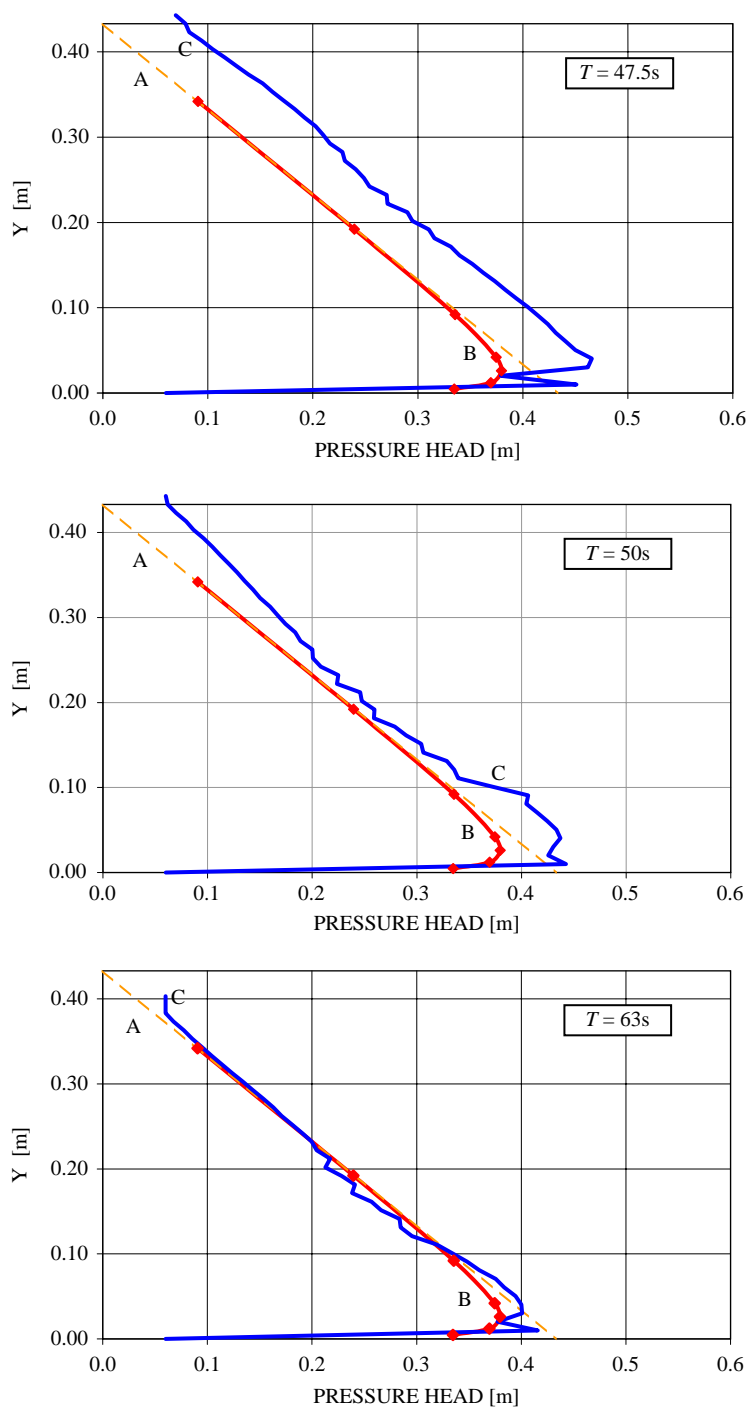


Figure 32.
Sluice gate. Model 1.
Pressure head
distribution. A-hydrostatic
distribution:
B-experimental values,
C-PFEM results

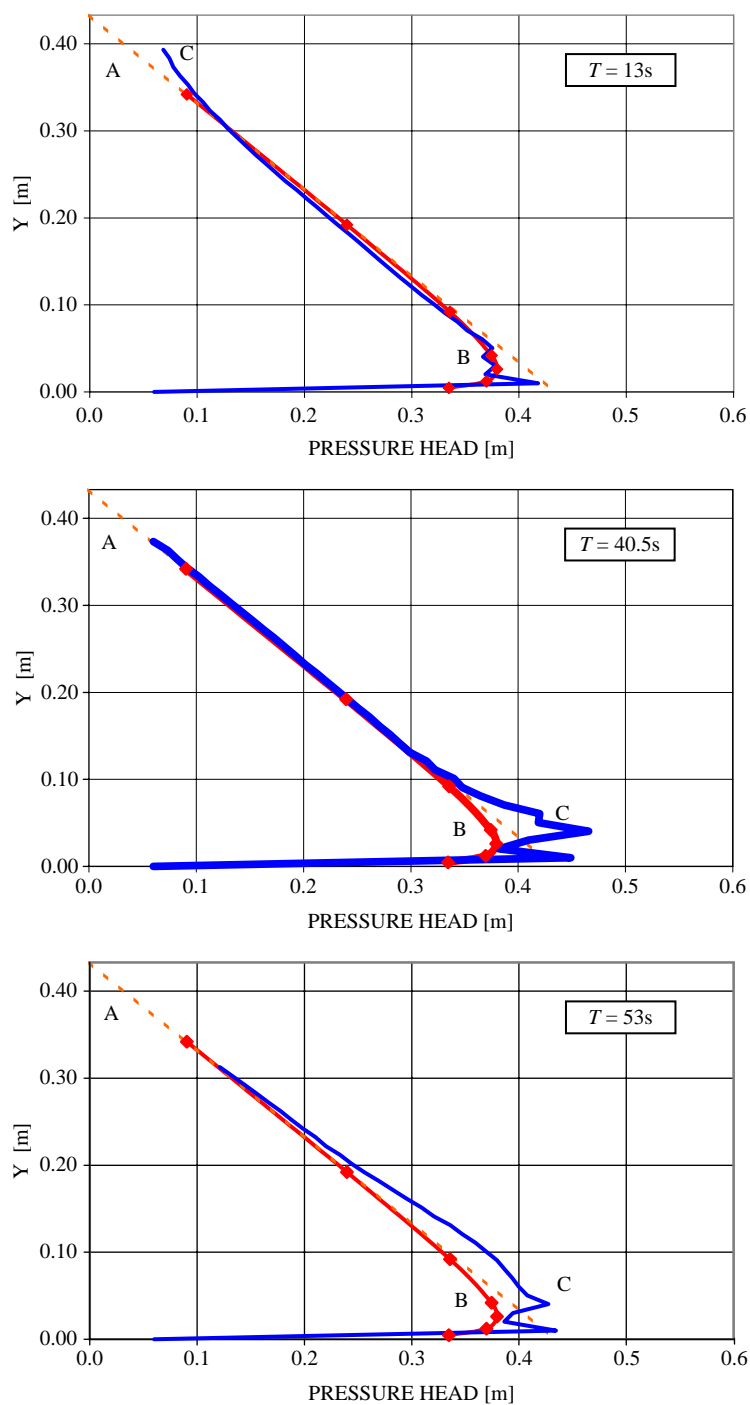


Figure 33.
Sluice gate. Model 2.
Pressure head
distribution. A-hydrostatic
distribution;
B-experimental values,
C-PFEM results

As expected, the model with holed gates (Model 2) gives an upstream depth that is more regular and less subjected to waves due to the constant entrance of fluid. This yields a very good comparison with the experimental results as shown in Figure 33. On the contrary, looking at the results for Model 1 with standard gates, the dynamic effect increases the value of the pressure head as shown in Figure 32(a) and (b).

Figure 34 clearly shows the oscillation of pressure at a point in the middle of the sluice gate. The oscillations are acceptable if compared with the average of experimental values (the horizontal line). The standard deviation is about 11 percent.

4.4 The outing discharge

The discharge of the under seal flow is completely defined if the geometrical characteristics are fixed and the depth of water in the tank is given. Analyzing a single section of the outing flow we would risk to be influenced by local effects. This is the reason why, we have integrated the velocity diagram of more than one vertical section of water for the same time instant. We consider many ideal vertical sections located at different distances from the sluice gate. The error in the discharge evaluation is always lower than 10 percent. Excluding isolated phenomena due to the variation of the water level in the tank, the PFEM results can be considered very accurate (Figure 35).

4.5 The analysis of the free surface of the downstream water

As mentioned in Section 4.2 a thin and planar sluice gate causes a contraction of the free surface of the flow, which depends on a contraction coefficient $C_c = 0.611$ (Figure 35, Gheti, 1984). That is, the depth of water of the under seal flow has to arrive at aC_c , where a is the level of gate lift.

Looking at the oscillations of the free surface, they are more or less of the same order than the dimension of the mesh 34.

Looking at the values of water depth in the interval between 10 and 50 cm from the sluice gate and calculating an average depth, we obtain quite good results in comparison to the standard deviation, as shown in Table IV.

4.6 The hydraulic jump

The hydraulic jump is an interesting phenomenon of energy dissipation generated by the clash of an upstream fast discharge with a downstream slow discharge. Many forms of hydraulic jump exist and basically they are characterized by the Froude number of the upstream discharge. We consider that for a generic section the total thrust is composed of two different parts (Figure 36):

Piezometer	H (cm)	$h = h - 4.43 \text{ mm}$ (cm)	z (cm)	$h - z = P/\gamma$ (cm)
1	43.7	43.3	34.20	9.06
2	43.4	43.0	19.20	23.76
3	42.7	42.3	9.20	33.06
4	41.1	40.7	4.20	36.46
5	39.4	39.0	2.60	36.36
6	35.8	35.4	1.20	34.16
7	30.0	29.6	0.48	29.08

Table III.
Experimental pressure
head for $Q = 31 \text{ l/s}$

- (1) the hydrostatic thrust:

$$M_{\text{hydr}} = \gamma \cdot A \cdot y_G;$$

and

- (2) the dynamic thrust:

$$M_{\text{dyn}} = \rho \cdot Q \cdot v.$$

An hydraulic jump is generated when the total thrust of the upstream discharge is equal to that of the downstream discharge. Once a control volume is defined, the momentum equation, together with the continuity equation (Ghetti, 1984; Cola, 2002), lead to a relationship between the upstream and the downstream flow depth. For a rectangular channel:

$$\frac{y_2}{y_1} = \frac{-1 + \sqrt{1 + 8 \cdot Fr_1^2}}{2}. \quad (12)$$

Figures 37 and 38 compare the development of the free surface with the experimental data for the hydraulic jump.

The kind of hydraulic jump we analyze is a stationary one. The lighter line with dots plots the experimental values, whereas the darker line shows the PFEM results. Note that the computational results are given for much more points than those computed in the experiment. The agreement between experimental and PFEM results is noticeable.

Unfortunately, for a phenomenon like an hydraulic jump steady state and equilibrium between the upstream and downstream channel are not fully obtained after few seconds and only the qualitative behavior can be reproduced and measured.

5. Dambreaking test

Dambreaking tests are quite popular for validation of free surface flow codes. Experimental-numerical validation for such experiments is quite straightforward because of the simplicity of the set up: the collapse of a water column that overpasses a downstream obstacle. No special conditions are needed for the onset of the flow.

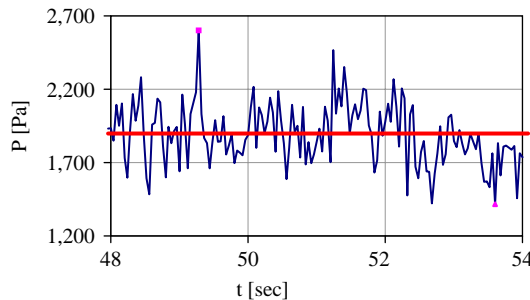


Figure 34. Sluice gate. Model 2. PFEM results for the time evolution of pressure at point A of Figure 28(b) of the sluice gate. The horizontal line shows the average of experimental results

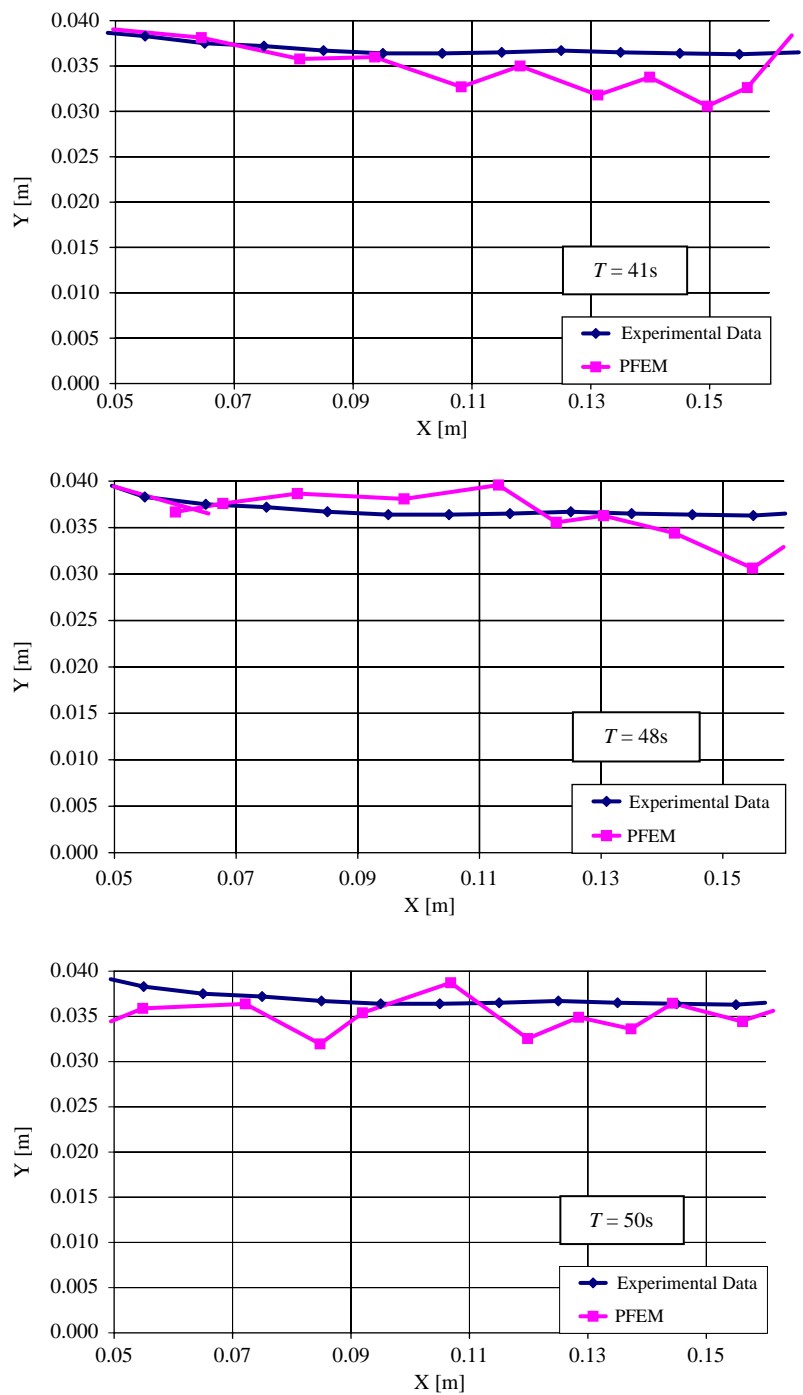


Figure 35.
Sluice gate. Free surface
after the gate showing the
contraction of the flow

In our case, both a 2D and a 3D model are used for analyzing the jet trajectory and the pressure on the obstacle. Experimental data are taken from two different tests available in the literature. Both tests follow a very similar approach with a different geometry for the experimental set up.

5.1 2D experiment

The first experimental data are taken from a test used for validating of a particle method by Koshizuka *et al.* (1995).

The geometry used is shown in Figure 39, where $L = 14.6$ and $h = 2.4$ cm. In the real experiment, the box is made of glass and the water column is supported by a vertical wall which is drawn up in an approximated time of 0.05 s. In the model, a vertical velocity of 5.84 m/s is given to the opening gate assuming a uniform motion of the flow. The initial mesh size is 0.001 m. The model has 50,000 nodes and nearly 100,000 three-noded triangles.

5.1.1 Trajectory comparison. Snap shots of the different instances of the experiment are shown for comparison with the PFEM results at the same times in Figures 40 and 41.

The biggest difference can be noticed for the simulation times equal to 0.4 and 0.5 s, i.e. when the jet touches the downstream wall. The air bubble which is trapped by the jet, is not captured well by the PFEM results. This is due to the fact that air particles are not modeled in our simulation.

5.2 3D experiment

A similar example was subsequently studied in 3D. The results were taken from the work presented at the Spheric Workshop held at the University La Sapienza of Rome in May 2006 (1st Spheric Workshop, 2006). The test is a 3D dambreak

t (s)	Medium depth (m)	Theoretical value (m)	SD
40	0.034	0.035	0.003
45	0.032	0.035	0.004
50	0.036	0.035	0.002

Table IV.
Depth of water between
10 and 50 cm from the
gate at different instances

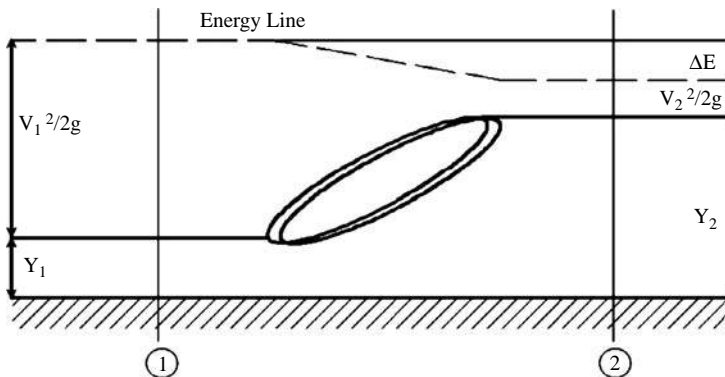


Figure 36.
Hydraulic jump

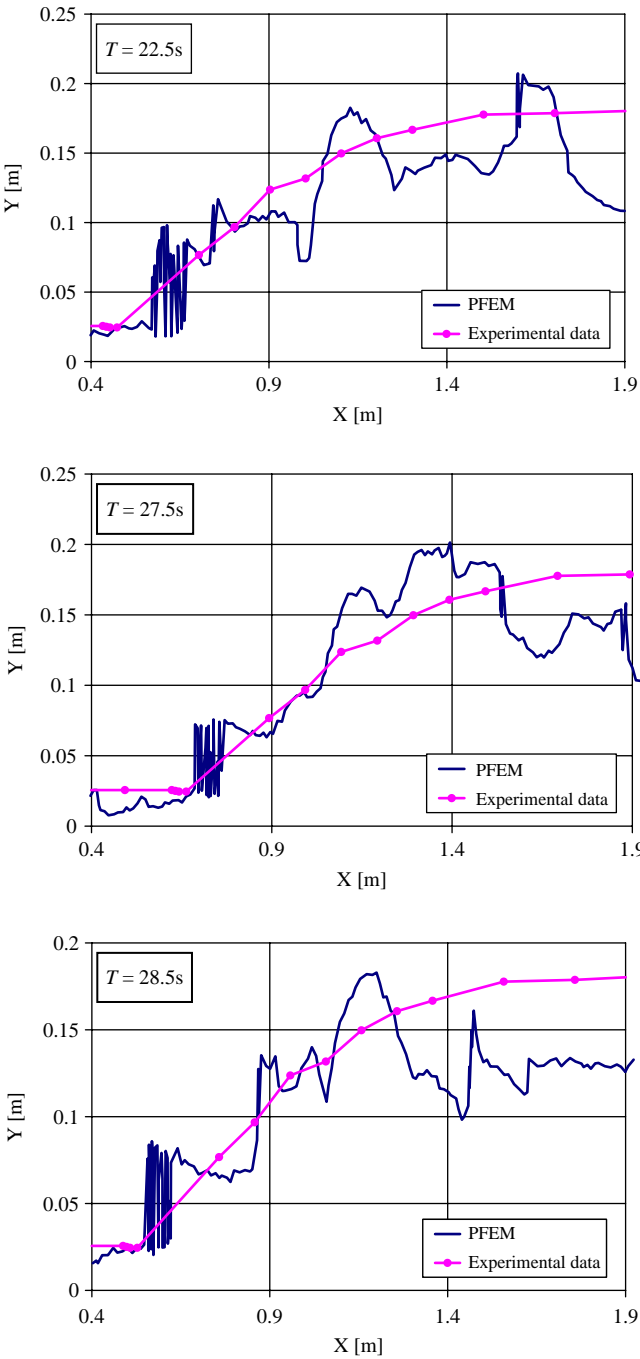


Figure 37.
Hydraulic jump. Free
surface development

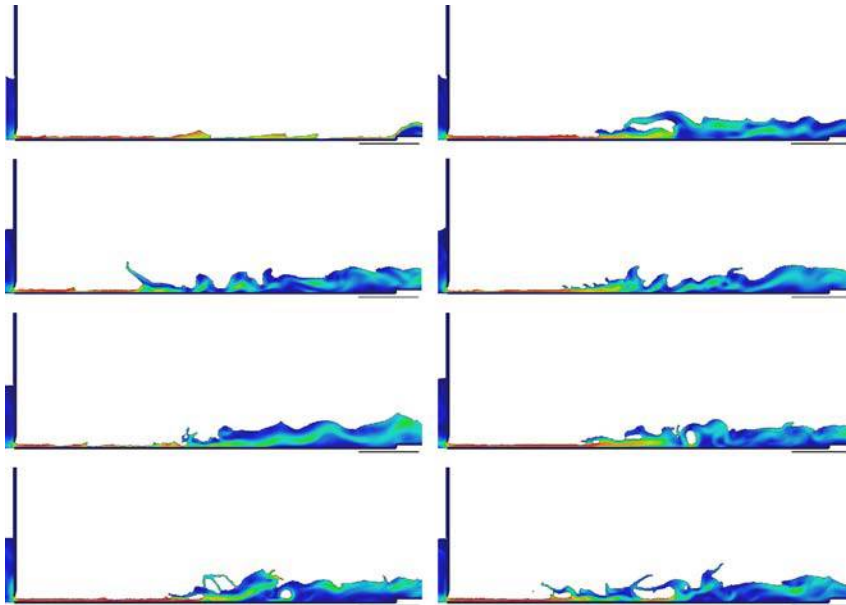


Figure 38.
Hydraulic jump. Colors
indicate the average of the
velocity modules

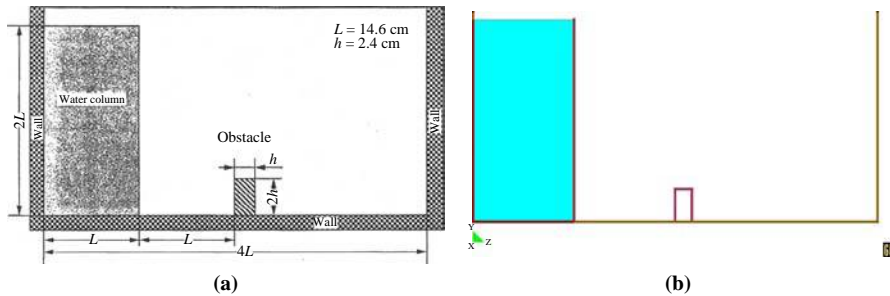


Figure 39.
2D dambreak. Set up: (a)
experimental set up taken
from Koshizuka *et al.*
(1995); (b) PFEM model

represented in Figure 42. The 3D model reproduces the geometry of the experimental set up well described in Kleefsman *et al.* (2005) and follows the parameters of Figures 43-45.

5.2.1 Pressure comparison. The experimental value of the pressure measured at eight points in the step as shown in Figure 42 are compared with the PFEM results (Figures 46-49). The simulation is carried out over 6 s of real time. The behavior is well reproduced also if the two nearest points to the angle have some discordance with the pressure level at the wave impact point. The maximum pressure value is always higher in the numerical solution than in the real measurements.

Pressure values at points 1 and 6 are compared for two different PFEM models. As expected, a coarse mesh gives a lower precision. The finer mesh model has 50,000 points and 170,000 four-noded tetrahedra, whereas the coarse mesh has only 14,000 points and 60,000 tetrahedra elements (Figure 50).

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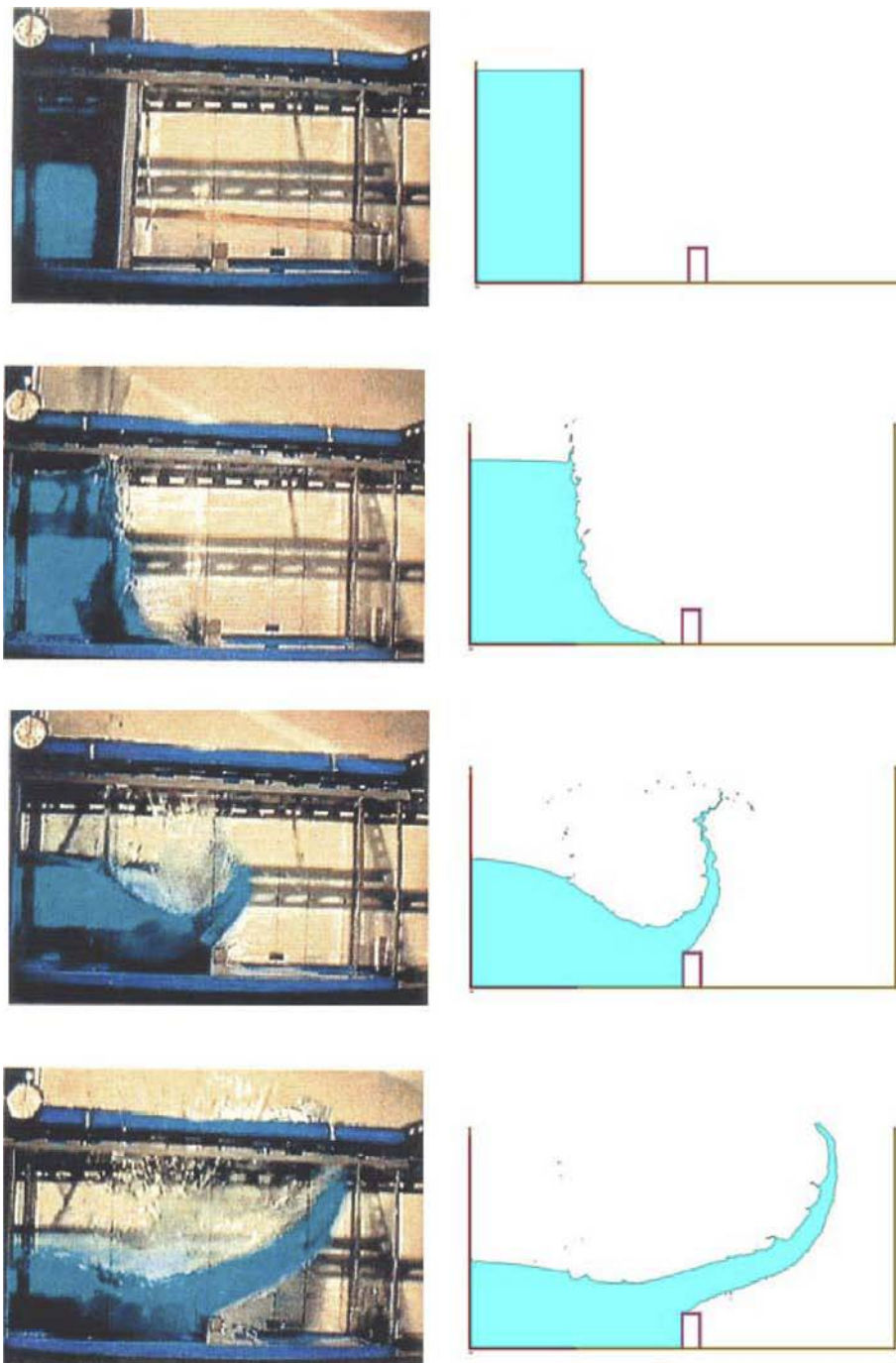


Figure 40.
2D dambreak.
Experimental results on
the left (Koshizuka *et al.*,
1995) compared with
PFEM results at the same
instance

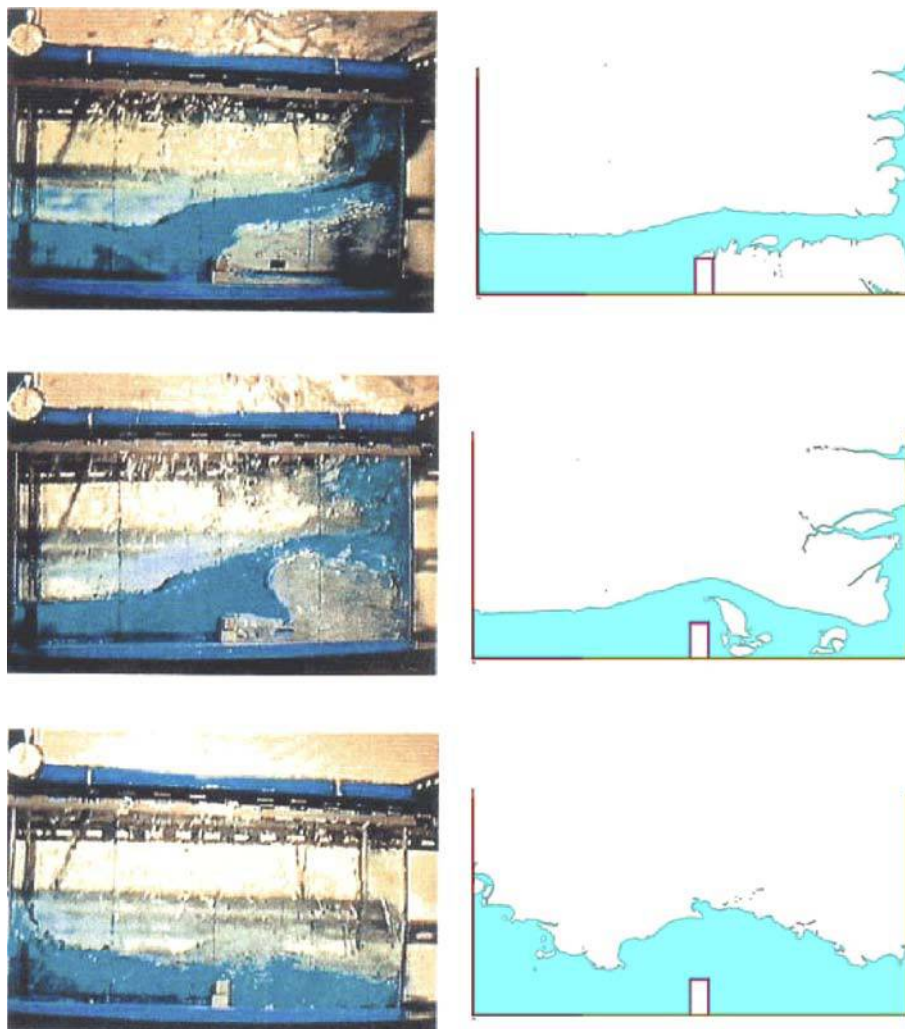
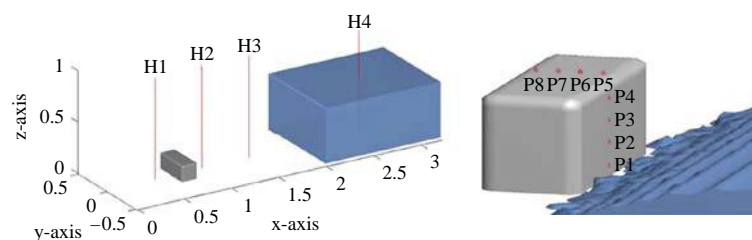


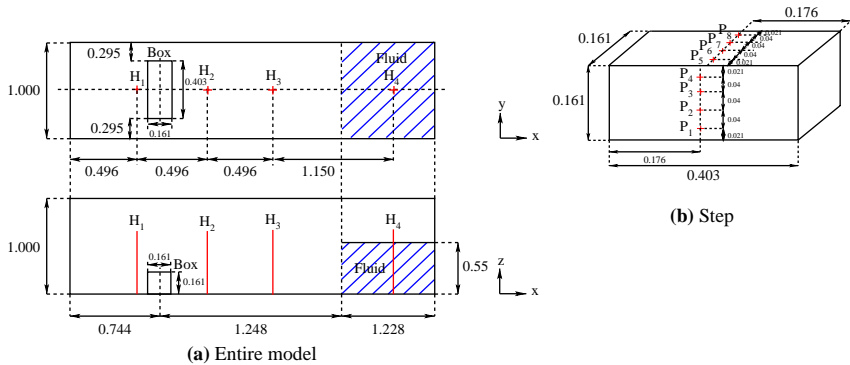
Figure 41.
2D dambreak.
Experimental results on
the left (Koshizuka *et al.*,
1995) compared with
PFEM results at the same
instance



Source: 1st Spheric Workshop (2006)

Figure 42.
Pressure measurement
positions in the dambreak
experiment

Figure 43.
3D dambreak. Geometry
of the experimental set up



Source: 1st Spheric Workshop (2006)

Figure 44.
3D dambreak. PFEM
model

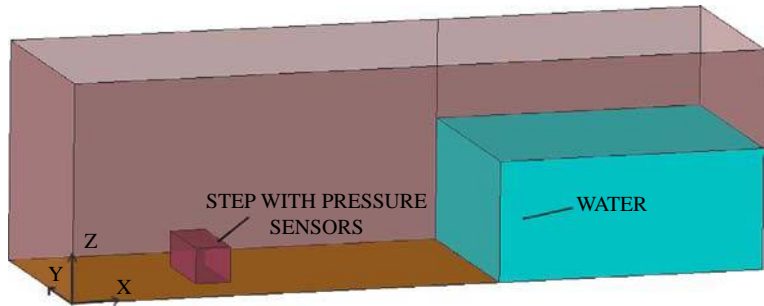
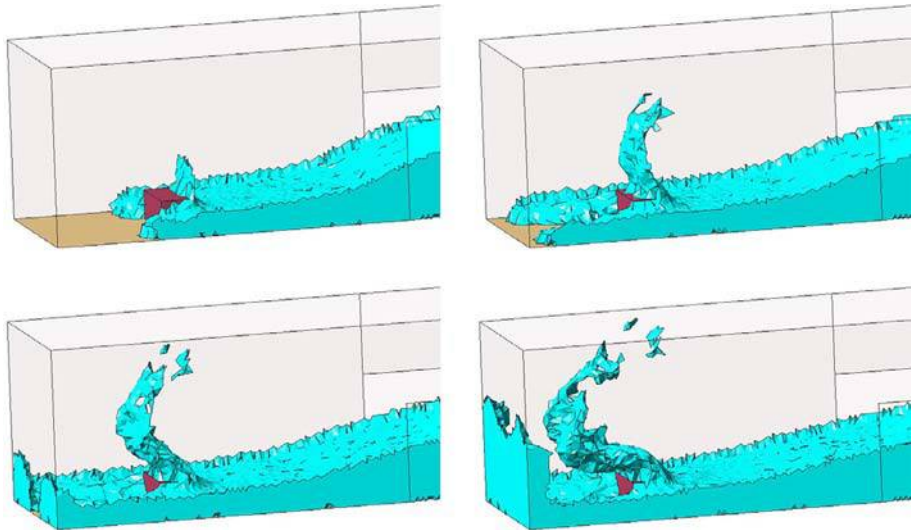


Figure 45.
3D dambreak. PFEM
results for the impact of
the water over the step



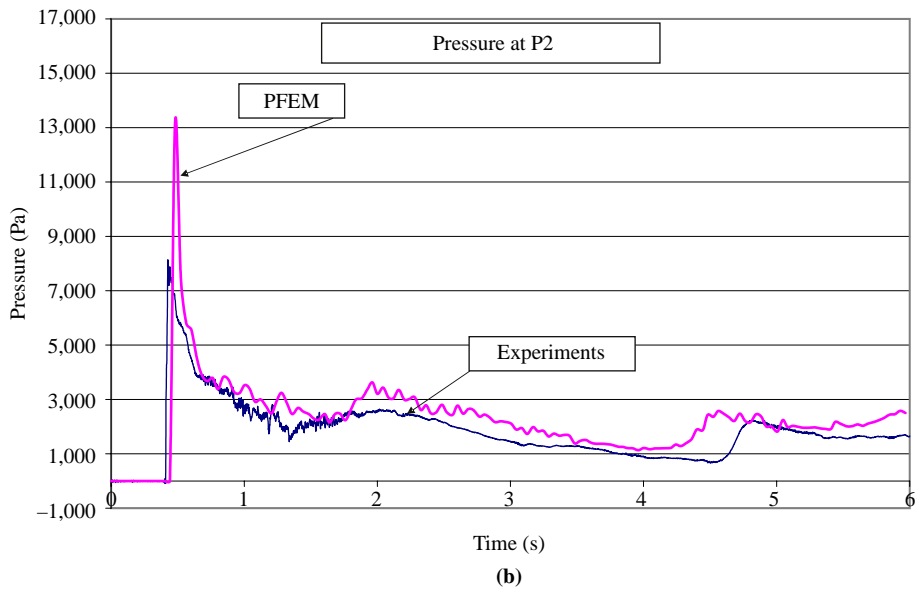
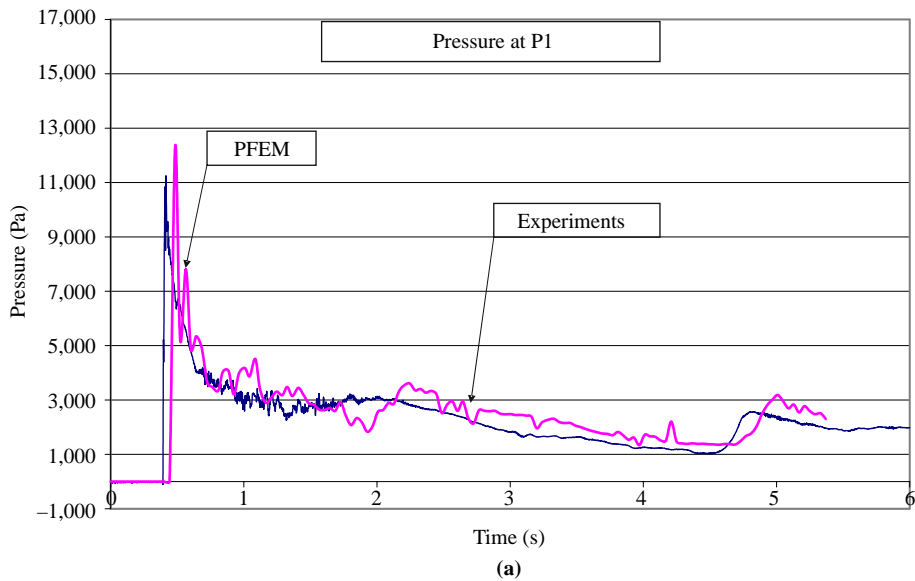


Figure 46.
3D dambreak. Pressure
evolution at points P1 and
P2 of the step

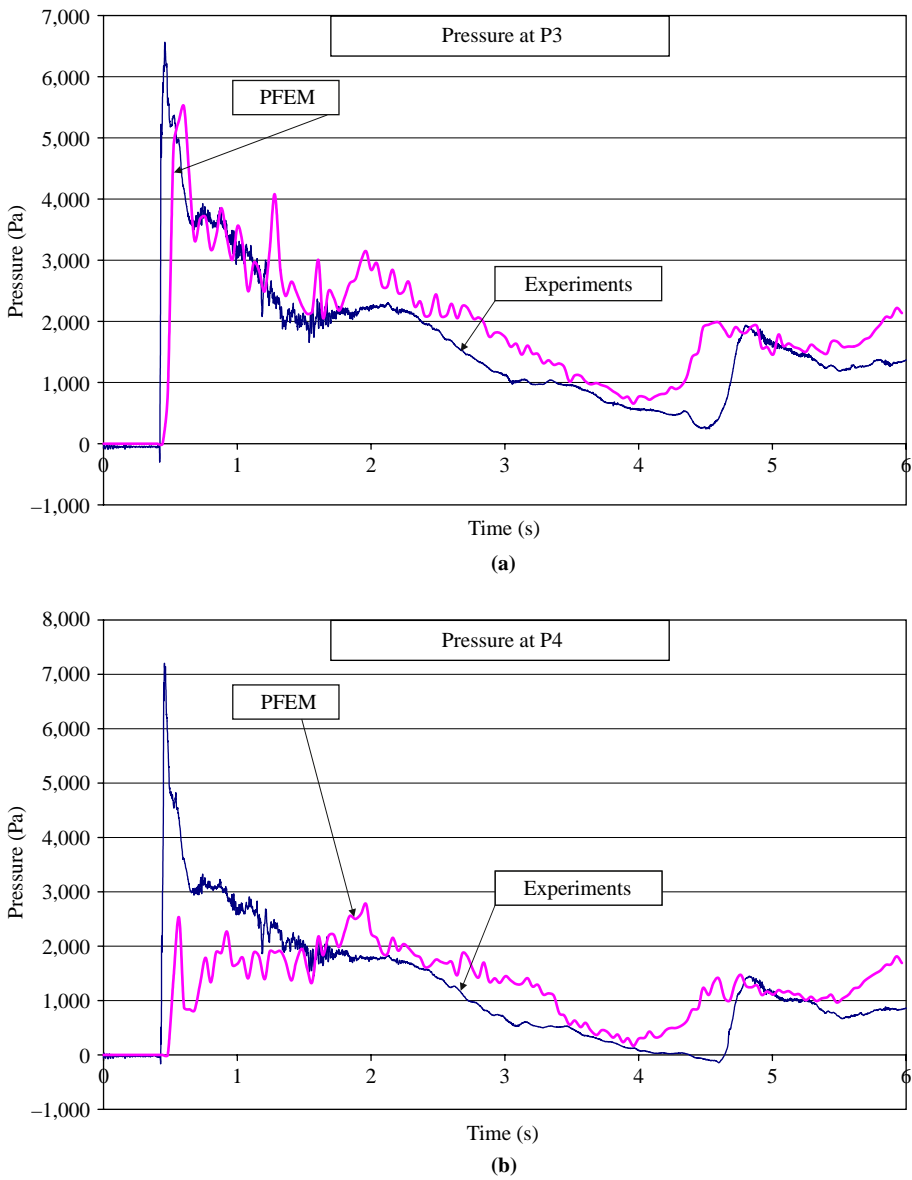


Figure 47.
3D dambreak. Pressure
evolution at points P3 and
P4 of the step

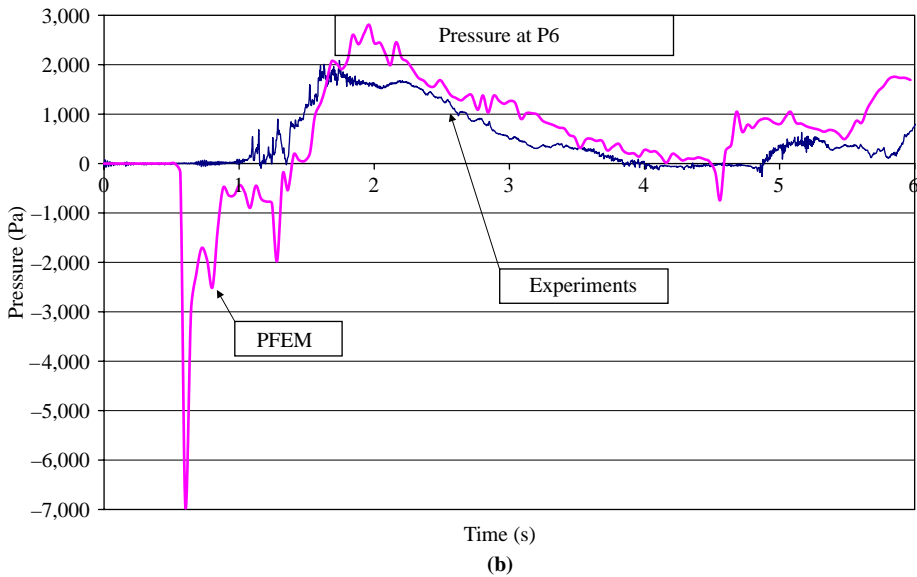
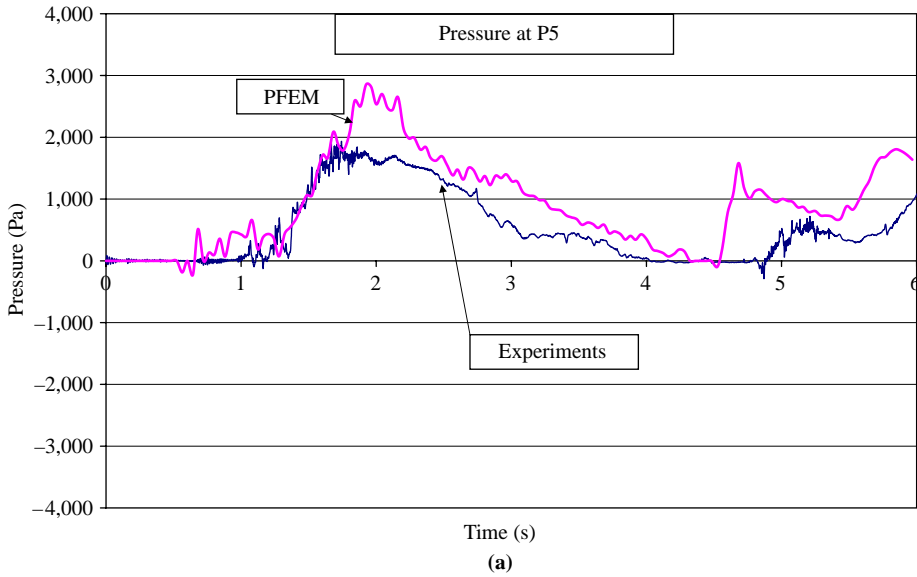


Figure 48.
3D dambreak. Pressure
evolution at points P5 and
P6 of the step

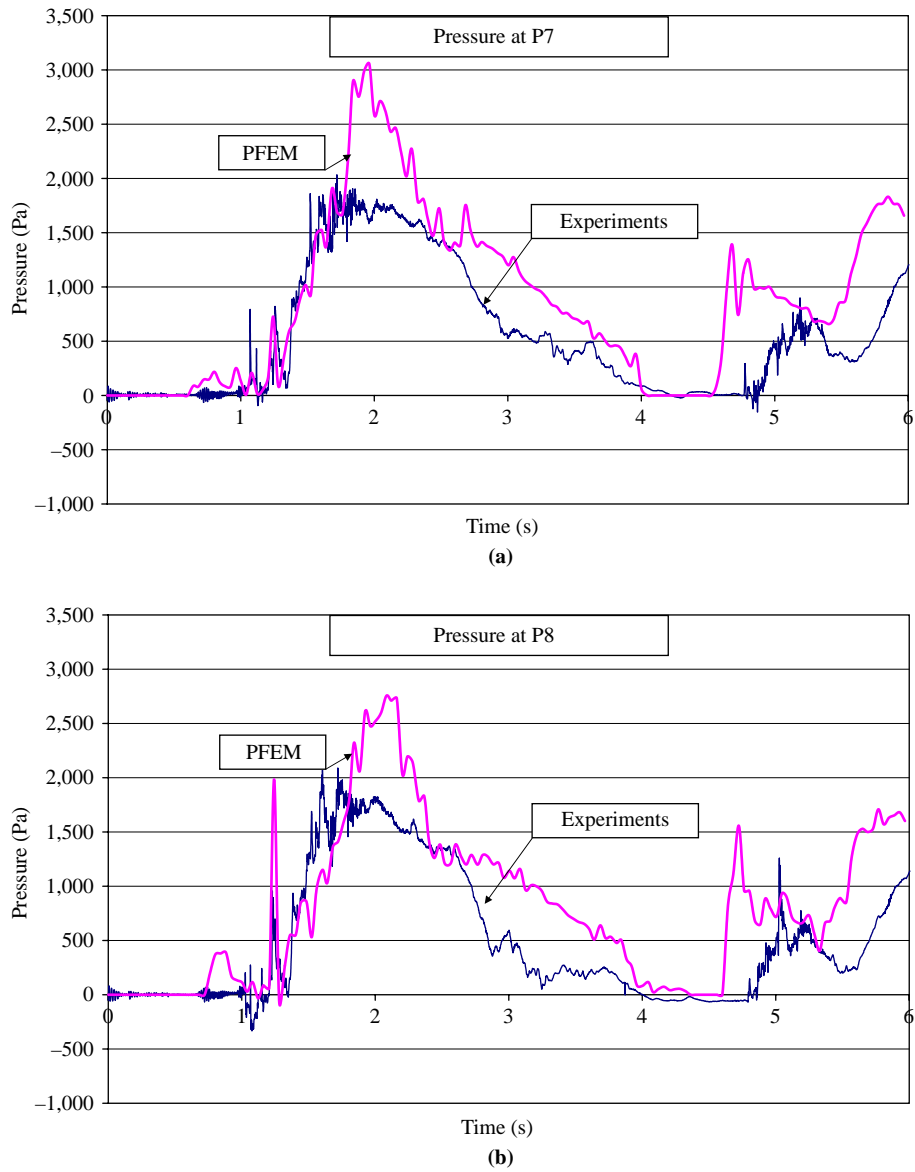


Figure 49.
3D dambreak. Pressure
evolution at points P7 and
P8 of the step

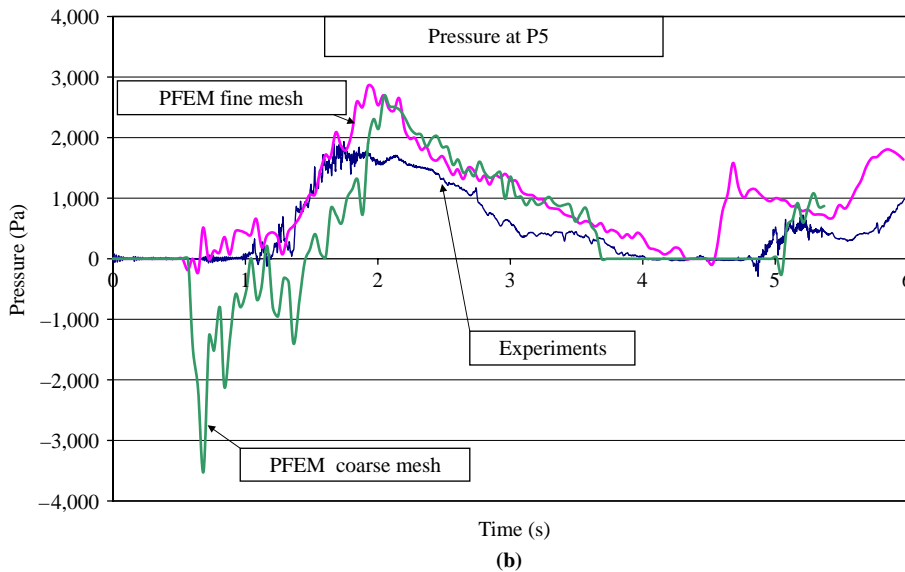
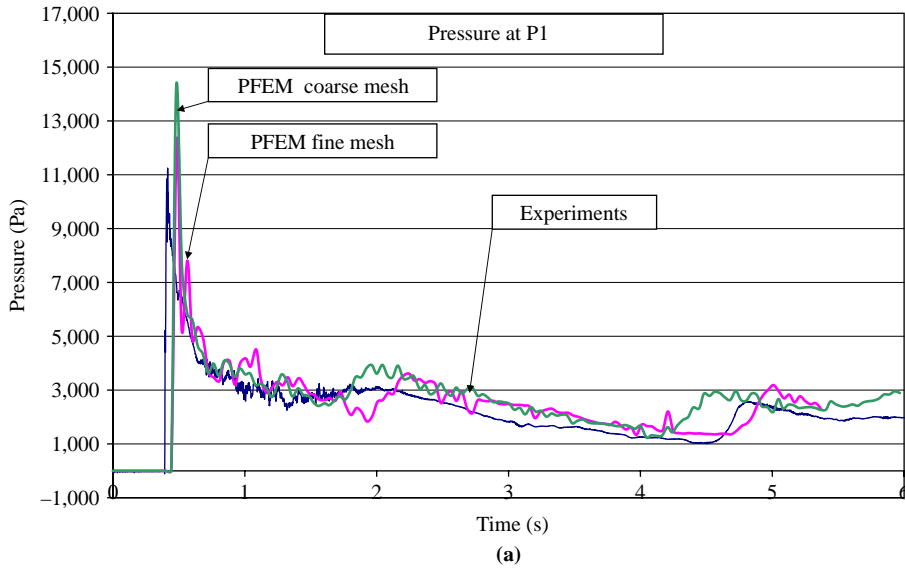


Figure 50.
3D dambreak. Pressure
evolution at P1 and P5.
Experimental data
compared with PFEM
results for two different
meshes

6. Conclusions

The PFEM is a powerful tool for solving free surface flows problems involving large deformation of the fluid domain.

Very good results have been obtained for the main flow parameters of relevance for each problem analyzed (such as the velocity field, the pressure distribution and the free surface position) as shown in the comparison with experimental data.

As expected, the mesh size is an important factor which influences the accuracy of the results. For instance, the mesh used in the examples presented is in all cases too coarse for capturing the real development of the viscous boundary layer.

Accurate solutions of the problems analyzed show the ability of the PFEM for reproducing very complex free surface flows to a high level of reliability.

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Corresponding author

A. Laese can be contacted at: antoldt@cimne.upc.edu

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