# FREE SURFACE FLOW IN VERTICAL TAYLOR-COUETTE SYSTE 

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#### Abstract

Vertical Taylor-Couette flow with a free surface at its top has been examined by numerical and experimental approaches. Being compared with the ideal case with infinite length cylinders, this system has effects of the finite lengths, the gravitational acceleration and the surface tension. The inner cylinder rotates and the outer cylinder and the bottom wall are stationary. The numerical approach uses the VOF method to model the free surface. In the experiment, the flow patterns are observed from the top and the meridional views, and the displacement of the free surface is measured. The flow modes are classified by the number of vortices appearing in the meridional plane and the radial flow directions at the top and the bottom. The power spectra of the displacement of the free surface and the bulk energy are numerically evaluated and the bulk energy tends to give favorable agreement with the spectra of the displacement obtained experimentally. The transition from the secondary mode flow appearing at higher Reynolds number to the primary mode flow at lower Reynolds number is examined and comparison between the numerical and experimental results are made.


## 1 INTRODUCTION

Flows with free surfaces appear in fluid machinery, chemical reactors and ocean currents and they have scientific and practical interests. In this paper, we numerically and experimentally study the free surface flow between vertical rotating cylinders. Some characteristics of the free surface flow around a rotating cylinder were the targets of the study of viscous fingerings [1], [2]. After the pioneering and milestone study conducted about one hundred years ago [3], a series of modified flows with no main axial flow has been carried out on the flows between rotating cylinders with finite lengths [4].

The examination of air-liquid and liquid-liquid flow between rotating cylinders is interesting and important, and not a small number of studies have been performed. For the airliquid flows in horizontal cylinders, Hubacz et al. [5] and Dluska et al. [6] experimentally classified the flow patterns including stratified and slug flows and observed the interfacial
area. The Peixinho et al. [7] measured the deformation of the free surface around a rotating inner cylinder in the flow at low liquid to air ratio and compared with the result of a numerical simulation considering the surface tension. They concluded that the free surface profile was well formulated by the capillary number. Baier et al. [8], Zhu et al. [9] and Nakase et al. [10] experimentally studied the two-liquids flow between horizontal cylinders. They examined the instability of the liquid-liquid interface and found the axially banded flow and radially separated flow. The mixture of water and kerosene flow in vertical rotating cylinders was studied by Sathe et al. [11]. They conducted the experimental and numerical study and compared the unsteady profile development of the interface and the dispersion process. Broadbent et al. [12] investigated the flow of the mixture of silicon oil and water in vertical rotating micro cylinders with free surface at the top and found the flow patterns similar to those in the horizontal cylinders [8]. However, they did not concern with the behavior of the free surface. Taylor-Couette system with a vertical axis and a free surface at the top has been studied in the case of the flows with or without an entrainment of the air. In the rotating flow with the entrainment, the bubbles were trapped near the boundaries of the vortices appearing in the meridional plane and the torque required to maintain the rotation would be reduced [13], [14]. Linek et al. [15] investigated the stability of flows with no entrainment. They determined the axial wave number at a given liquid height and referred to the Eckhaus boundary to investigate the stability. They also observed the reduction of the number of vortices in the decelerating flow. Deng et al. [16] estimated the profiles of the free surface by experimental observations and theoretical predictions and found the increase in the axial velocity component when the flow became wavy. Vertical Taylor-Couette system with a free surface was also utilize to understand the rotating flow with free surface [17] and the azimuthally periodic flow with Rossby wave between the cylinders with a low inner to outer radius ratio [18].

Taylor-Couette system with finite length cylinders has multiple solutions [19], and it shows some flow modes. Let the inner cylinder rotate, while the outer cylinder is stationary. The primary modes appear when the inner cylinder accelerates in quasi-steady state and the vortices in the meridional plane appear via a pitchfork bifurcation. The sudden acceleration and/or the higher Reynolds number make secondary modes with vortex numbers and flow directions different from those of the primary modes. Nakamura et al. [20] used the cylinders with both stationary ends and experimentally obtained the secondary mode flows. They also observed the transition processes of the well-developed secondary mode flows to the primary mode flows during the slow deceleration of the inner cylinder. The aspect ratio they adopted, which is defined by the fraction of the cylinder length to the gap width between two cylinders, is from 3.0 to 7.0. Watanabe et al. [21] numerically investigated the bifurcation result in [20] and obtained good agreement. Toya et al. [22] carried out their experimental study of the bifurcation between the secondary and primary modes in the vertical asymmetric TaylorCouette system with a top free surface and a stationary bottom wall. For their results, Watanabe et al. [23] simulated the system with an effect of surface tension at the small aspect ratio up to about 4.0 and evaluated the bifurcation diagram of the secondary and the primary modes. A new interest about the vertical Taylor-Couette system with free surface is the slant of the axis, and the effect of the tilt angle on the flow stability was examined [24], [25].

In this paper, we extend the study [23] and numerically investigate the vertical TaylorCouette system with a free surface at the top. We also conduct some experiments with our
coworker who is one of the authors of study [22] and draw a comparison. Then, we investigate the developing process of the free surface flows, the flow modes, the unsteady behaviors of the well-developed flows and the transition from the secondary modes at higher Reynolds numbers to the primary modes.

## 2 FORMULATION

### 2.1 Flows in Taylor-Couette system with vertical axis and free surface

Taylor-Couette system considered in this paper consists of two vertical concentric cylinders. The inner cylinder with radius $r_{i}$ rotates and the outer cylinder with radius $r_{o}$ remains stationary. The bottom end of the cylinders is fixed wall. The lengths of the cylinders are long enough to let the working fluid have a free surface between the cylinders and the top and bottom conditions are asymmetric. The reference velocity is the circumferential velocity of the inner cylinder $U$ and the reference length is the cylinders' gap width $d=r_{o}-r_{i}$, and they define the Reynolds number $R e$. The height of the quiescent fluid is $L$ and the aspect ratio $\Gamma$ is given by $L / d$.

The flow mode is specified by the number of vortices and the flow directions at the top and the bottom in the meridional plane. The rotation of the inner cylinder brings a centrifugal effect. Consequently, the flow on the top free surface is radial outflow. On the other hand, the flow in a viscous Ekman layer on the bottom is retarded and the flow direction is radially inward near the bottom. Therefore, the outflow at the top and the inflow at the bottom are the normal flow, and then the top vortex with outflow on the free surface and the bottom vortex with inflow near the bottom wall are the normal vortices. The flow direction opposite to the direction of the normal flow is anomalous and the anomalous flow is brought by an anomalous vortex. The mode of the flow with a normal top vortex and a normal bottom vortex is a normal mode. Otherwise, the flow mode is anomalous. When the normal mode flow has $n$ vortices in the meridional plane, the mode is the normal $n$-vortex mode and it is denoted by $\mathrm{N} n$. The anomalous mode flow with $n$ vortices is the anomalous $n$-vortex mode $\mathrm{A} n$, and $\mathrm{Au} n$ means the mode of $\mathrm{A} n$ with an upper top anomalous vortex.

Other than the normal and anomalous modes, the vortex flows are classified into the primary mode and the secondary mode. The primary mode represents the first vortex flow which appears after a quasi-steady acceleration of Couette flow. Even in the same geometrical flow field as that of the primary mode, the secondary mode flows have numbers of vortices and/or rotating directions different from those of the primary mode flow and they usually emerge at higher Reynolds numbers.

### 2.2 Experimental and numerical methods

The experimental setup by Toya [22] is used to measure and observe the flow. The radii of the inner and the outer cylinders are 20 mm and 30 mm , respectively, and the radius ratio $\eta$ is 0.667 . The inner cylinder rotates and the outer cylinder and the bottom wall are fixed. The working fluid is silicon oil and its kinematic viscosity, density and surface tension are $9.54 \times 10^{-5} \mathrm{~m}^{2} / \mathrm{s}, 935.0 \mathrm{~kg} / \mathrm{m}^{3}$ and $2.07 \times 10^{-3} \mathrm{~N} / \mathrm{m}$, respectively. A slit light source and a small quantity of aluminum flakes are used to visualize the flow in the meridional plane.

The displacement of the free surface at the midpoint of the cylinders is measured by an ultrasonic sensor with a response time of 0.2 ms and a resolution of 0.2 mm . When the lengths of the cylinders are infinite, the theoretical critical Reynolds number of the onset of the Taylor vortices is 76.4 [26].

The numerical simulation [23] is conducted under the condition of the experiment. The free surface is modelled by the VOF method. The governing equations are the unsteady axisymmetric Navier-Stokes equations, the equation of continuity and the conservation equation of the volume fraction $F$. The effects of the gravitational acceleration and the surface tension estimated from the curvature of the free surface profile are considered. The coordinate system is the cylindrical system with $(r, \theta, z)$ and the velocity components are $u, v$ and $w$. In the numerical formulation, physical quantities are made nondimensional by the reference length $d$ and the reference velocity $U$. Initially, the velocity components are zero in the entire flow region, and the inner cylinder is accelerated at the prescribed Reynolds number. In order to identify the critical Reynolds numbers of the transitions from the secondary modes to the primary modes, the secondary mode flow is decelerated in a quasi-steady state where the Reynolds number is reduced by $2 \%$ during the nondimensional time of 50 .

## 3 RESULTS

### 3.1 Development process and spectrum of unsteady flow

When the inner cylinder at rest begins to rotate, mushroom-type vortices appear on the cylinder surface and they merge or decay after their interactions. Then, a well-developed flow emerges usually after the appearance of some flow patterns. Figure 1 shows the flow development process at $\Gamma=1.6$ and $R e=800$. Each panel depicts the flow pattern in the meridional plane at nondimensional time $t$ and the warm color and the cold color represent the vortices rotating in the clockwise direction and the counterclockwise direction, respectively. At $t=96.0$, three vortices exist in the meridional plane and the rotating directions indicated by the curved arrows are clockwise, counterclockwise and clockwise from the top to the bottom. This flow has radial outflow on the top and radial inflow on the bottom and the flow mode is the normal 3-vortex mode N3. This flow is transient. The upper two vortices decay and the final flow is N 1 with one vortex. The well-developed flow at $\Gamma=$


Figure 1 Development process of flow at $\Gamma=1.6, R e=800$.
2.2 and $R e=617$ presented in Figure 2 has two vortices. The lower vortex has a radial outflow on the bottom wall and it is anomalous, and the flow mode is the anomalous 2-vortex mode A2. This flow is found experimentally and numerically. Some details about the anomalous vortex on stationary walls were denoted by Watanabe et al. [21]. Figure 3 shows the flows at $\Gamma=2.4$. When the Reynolds number is 800 , the flow is the normal 3 -vortex mode N3. Even in the same geometrical flow field, the flow at $R e=1000$ has an anomalous vortex on the bottom wall and the flow mode is A4.

The unsteady behavior of the free surface is one of the interesting phenomena in this kind of flows. Figure 4 represents the experimentally and numerically obtained power spectra of the displacement of the free surface and the numerically predicted power spectra of the


Figure 2 Anomalous 2-vortex mode A2 at $\Gamma=2.2, R e=617$.


$$
\mathrm{Re}=800, \mathrm{~N} 3
$$


$\operatorname{Re}=1000, \mathrm{~A} 4$
Figure 3 Normal and anomalous modes at $\Gamma=2.4$.


Figure 4 Power spectra of experimentally measured displacement, numerically predicted displacement and numerically predicted bulk energy. $\Gamma=5.6$. upper: $R e=1395$, lower: $R e=1550$.
volume-averaged bulk kinetic energy. The numerical displacement is measured at the center point between two cylinders. The upper panels show the results at $R e=1395$ and the rotation frequency of the inner cylinder is 2.65 Hz . The lower results are at $R e=1550$ and the rotation frequency is 2.94 Hz . At the lower Reynolds number, the experimental result shows a simple harmonic motion at 3.05 Hz . The numerical displacement has its first peak at 1.22 Hz and the higher peak at 2.51 Hz , and these peaks do not match the experimental peaks. The spectrum of the energy has its first peak at 2.44 Hz and this almost corresponds to the peak in the experiment. When the Reynolds number is higher, the experimental displacement shows the first peak at 3.42 Hz and the second peak at 1.46 Hz . The numerical displacement has some peaks and the first peak at 1.34 Hz approaches the second peak in the experiment. The first and the second peaks of the energy are 2.68 Hz and 1.34 Hz , respectively, and the profile is similar to that of the experiment. The ultrasonic sensor capturing the displacement measures not a pin-point value but an averaged value in a finite area. This measurement in a finite area may be affected by the unsteady motion giving the kinetic energy.

### 3.2 Flow mode transition

As mentioned above, the secondary mode flows tend to appear at higher Reynolds numbers and the primary mode flows emerge at lower Reynolds numbers. In this section, we estimate the critical Reynolds numbers at which the secondary mode flows change to the primary mode flows when the flow is decelerated in quasi-steady states. The parameter space we numerically study has the aspect ratio from 4.5 to 6.0 and the Reynolds number from 100 to


Figure 5 Flow pattern transitions in decelerating flows.
500. The primary mode is N 5 at $4.5 \leq \Gamma \leq 5.2$, and it is Au6 at $5.3 \leq \Gamma \leq 6.0$.

The transition processes of the decelerating vortex flows are shown as velocity vectors in the meridional sections in Figure 5. The color represents the value of the kinetic energy of the radial and axial velocity components. The left panel is the transition from N7 to Au6 at $\Gamma=$ 5.9. The change begins with the decay of the normal vortex at the top. The second vortex from the top grows and it collapses the top vortex, and the anomalous 6 -vortex mode emerges with a radial inflow on the free surface (Au6). In the middle panel, the aspect ratio is 5.5 and the flow before the transition has a top anomalous vortex. With the deceleration, the top vortex weakens and the normal 5 -vortex mode N5 forms. The right panel shows the transition from N7 to N5. The flow modes before and after the transition are normal, and not the top vortex but a pair of vortices in the flow field disappears. In this case, the third and fourth vortices become smaller and diminish. The aspect ratio of the case in the left panel is about 6.0 and the six-vortex mode is expected to be stable. The flow fields in the middle and right panels have the aspect ratio about 5.0 and the normal mode with five vortices is stable.

The experimental and numerical result of the bifurcation diagram of the primary and the secondary modes are represented in Figure 6. When the aspect ratio is 6.0, the experimental and numerical results show the critical Reynolds number of the transition from N7 to Au6 is about 100. The critical Reynolds number of the transition from N7 to N5 is about 200 at $\Gamma=$ 5.0. However, the overlapping range of the aspect ratio of the experimental and numerical results is limited. This is because of the difficulty to find the transition experimentally and numerically. The results of the transition from Au6 to N5 shows a difference. In the experiment, the critical Reynolds number reduces from 160 to 100 as the aspect ratio increases from 5.5 to 6.0 . On the other hand, the numerical result gives almost constant critical Reynolds number. Usually, the anomalous modes a little bit hardly appear when the experimental condition is kept stable. This difference may depend on the imperfection of the experimental setup.


Figure 6 Bifurcation diagram between the primary mode and the secondary mode.

## 5 CONCLUSIONS

- Numerical and experimental study was carried out on the vertical Taylor-Couette system with a free surface at the top. The inner cylinder rotates and the outer cylinder and the bottom wall are stationary. The cylinders are long enough to include the free surface in the annulus of the cylinders. In the experiment, the flow patterns in the meridional section are observed and the displacement of the free surface is measured. The numerical approach is used to predict the unsteady flow development. As has been known, some flow modes appear in a fixed geometrical configuration, which depend on the Reynolds number.
- Though the numerically obtained power spectra of the displacement of the free surface includes its peaks corresponding those in the experimental result, the numerically estimated kinetic energy spectra shows a reasonable agreement with the spectra of the experimentally measured displacement.
- The anomalous modes with a radial outflow on the bottom wall and/or a radial inflow at the top free surface are identified.
- The numerically predicted transition from N7 (normal 7-vortex mode) to Au6 (anomalous 6-vortex mode with an upper anomalous vortex) and the one from N 7 to N5 (normal 5-vortex mode) have favorable agreements with the experimental results. However, the numerical and experimental results show a difference in the transitions from Au6 to N5.


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