A DIFFUSED INTERFACE IMMERSED BOUNDARY–LATTICE BOLTZMANN METHOD FOR SIMULATION OF STENOSIS

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Abstract. The immersed boundary method has attracted growing interest in CFD research community due to its simplicity in dealing with moving boundaries. In the diffused interface immersed boundary method, a discrete delta function is introduced to account for the boundary effects on the fluid, which causes the diffusion of the boundary interface. Therefore, the diffused interface immersed boundary method requires higher grid resolution in the vicinity of the immersed boundaries to get a better representation of the boundary. A strategy for the application of diffused interface immersed boundary-lattice Boltzmann method is introduced in the simulation of stenosis. The developed numerical method has been examined in the simulation of stenosis. Results show that the current solver is able to accurately predict the velocity profile within the stenosis.

1 INTRODUCTION

Computational fluid dynamics become an important tool for improving our understanding the effect of normal physiological and pathological behaviors in the arterial system [1]. Stenosis, in ideal circumstances, is modeled as a straight rigid pipe with a constriction. A range of studies have been conducted based on this model, including experimental study by Ahmed & Giddens [2], two-dimensional (2D) numerical studies by Tian et al. [3] and Huang et al. [4] and three-dimensional (3D) numerical studies by Pal et al. [5] and Varghese et al. [6].

In the diffused interface immersed boundary method (IBM), a discrete delta function is introduced to account for the boundary effects on the fluid, which causes the diffusion of the boundary interface. Thus, the diffused interface IBM is first order scheme [7] and requires higher grid resolution in the vicinity of the immersed boundary to get a better representation of the boundary. Liu et al. [8] used the diffused interface IBM for the simulation of flapping foil. Wang et al. [9, 10] adopted the diffused interface IBM for the simulation of heat transfer and fluid-structure interaction of a flapping filament. Pal et al. [5] used a second order IBM in the simulation of turbulent flow in an idealized stenotic blood vessel. However, there are few studies on the diffused interface IBM in the simulation of blood flow. In this work, a strategy for the application of diffused interface immersed boundary-lattice Boltzmann method (IB-LBM) is introduced here for the simulation of stenosis.

2 GOVERNING EQUATIONS

The unsteady incompressible flow is governed by the continuity and Navier-Stokes equations:

$$\nabla \cdot \boldsymbol{u} = 0, \quad \frac{\partial \boldsymbol{u}}{\partial t} + \boldsymbol{u} \cdot \nabla \boldsymbol{u} = -\frac{1}{\rho} \nabla p + \nu \nabla^2 \boldsymbol{u} + \boldsymbol{f}, \tag{1}$$

where **u** is the fluid velocity, ρ is the constant density, and p and v are respectively the pressure and kinematic viscosity, and **f** is the body force.

3 NUMERICAL METHOD

The D3Q19 lattice Boltzmann method (LBM) with multi-relaxation-time (MRT) model is adopted for the fluid dynamics. The IBM is used to realize the no slip and no penetration boundary conditions of the rigid tube wall. In the LBM, the macroscopic dynamics of the fluid are the result of the statistical behaviour of the particles, which are described by the distribution function $g_i(\mathbf{x}, t)$. The evolution of this distribution function is according to

$$g_i(\mathbf{x} + \mathbf{e}_i \Delta t, t + \Delta t) - g_i(\mathbf{x}, t) = \Omega_i(\mathbf{x}, t) + \Delta t G_i,$$
⁽²⁾

where $g_i(\mathbf{x},t)$ is the distribution function for particles with velocity \mathbf{e}_i at position \mathbf{x} and time t, Δt is the time increment, $\Omega_i(\mathbf{x},t)$ is the collision operator, and G_i is the forcing term accounting for the body force \mathbf{f} . The D3Q19 model [11] is used on a square lattice. The MRT collision model is adopted and is given by [12]:

$$\Omega_i = -(\boldsymbol{M}^{-1}\boldsymbol{S}\boldsymbol{M})_{ij}[g_i(\boldsymbol{x},t) - g_i^{eq}(\boldsymbol{x},t)], \qquad (3)$$

where \boldsymbol{M} is a $q \times q$ transform matrix, and q is the number of particle speed. For D3Q19 model, \boldsymbol{M} is a 19 × 19 matrix. $\boldsymbol{S} = diag(\tau_0, \tau_1, \dots, \tau_{q-1})^{-1}$ is a non-negative diagonal 19 × 19 relaxation matrix. The determination of \boldsymbol{S} in three-dimensional model can be found in [11]. The equilibrium distribution function g_i^{eq} is defined as

$$g_i^{eq} = \rho \omega_i \left[1 + \frac{\boldsymbol{e}_i \cdot \boldsymbol{u}}{c_s^2} + \frac{\boldsymbol{u} \boldsymbol{u} : (\boldsymbol{e}_i \boldsymbol{e}_i - c_s^2 \boldsymbol{I})}{2c_s^4} \right],\tag{4}$$

where $c_s = \Delta x/(\sqrt{3}\Delta t)$ is the speed of sound, Δx is the lattice spacing, I is the unit tensor, and the weighting factors ω_i are given by $\omega_0 = 1/3$, $\omega_{1-6} = 1/18$ and $\omega_{7-18} = 1/36$. The velocity u, mass density ρ and pressure p can be obtained according to

$$\boldsymbol{\rho} = \sum_{i} g_{i}, \quad p = \boldsymbol{\rho} c_{s}^{2}, \quad \boldsymbol{u} = (\sum_{i} \mathbf{e}_{i} g_{i} + \frac{1}{2} \boldsymbol{f} \Delta t) / \boldsymbol{\rho}, \tag{5}$$

The force scheme proposed in [14] is adopted to determine G_i

$$G_i = [\boldsymbol{M}^{-1}(\boldsymbol{I} - \boldsymbol{S}/2)\boldsymbol{M}]_{ij}F_i,$$
(6)

$$F_{i} = \left(1 - \frac{1}{2\tau}\right)\omega_{i}\left[\frac{\boldsymbol{e}_{i} - \boldsymbol{u}}{c_{s}^{2}} + \frac{\boldsymbol{e} \cdot \boldsymbol{u}}{c_{s}^{4}}\boldsymbol{e}_{i}\right] \cdot \boldsymbol{f},$$
(7)

where τ is the non-dimensional relaxation time.

In the present study, an diffused interface IBM is adopted to handle the no slip and no penetration of the rigid tube walls. A comprehensive overview of this method was given by Kang et al. [15]. In this method, a body force f is added in the Navier-Stokes equation to mimic a boundary condition according to,

$$\boldsymbol{f}(\boldsymbol{x},t) = -\int \boldsymbol{F}_{ib}(s,t)\delta(\boldsymbol{x}-\boldsymbol{X}(s,t))d\boldsymbol{A},$$
(8)

$$\boldsymbol{F}_{ib}(s,t) = 2(\boldsymbol{U}_{ib}(s,t) - \boldsymbol{U}(s,t)), \qquad (9)$$

$$\boldsymbol{U}_{ib}(s,t) = \int \boldsymbol{u}(x,t) \delta(\boldsymbol{X}(s,t) - \boldsymbol{x}) d\boldsymbol{x}, \qquad (10)$$

where $F_{ib}(s,t)$ is the Lagrangian force density, dA is the element surface area of the immersed boundary, $\delta(\mathbf{x} - \mathbf{X}(s, t))$ is Dirac's delta function, \mathbf{x} is the coordinate of the fluid lattice nodes, \mathbf{X} is the coordinate of the tube wall, $U_{ib}(s,t)$ is the immersed boundary velocity obtained by interpolation at the immersed boundary, and $\boldsymbol{U}(s,t) = 0$ for a rigid tube wall. The 4-point discrete delta function $\delta_h(\boldsymbol{x})$ is used to approximate the Dirac delta function [13]:

$$\delta_h(\mathbf{x}) = \frac{1}{\Delta x \Delta y \Delta z} \phi(\frac{x}{\Delta x}) \phi(\frac{y}{\Delta y}) \phi(\frac{z}{\Delta z}), \tag{11}$$

$$\phi(r) = \frac{1}{8} \left(3 - 2|r| + \sqrt{1 + 4|r| - 4r^2} \right) \quad 0 \le |r| \le 1, \tag{12}$$

$$\frac{1}{8} \left(5 - 2|r| + \sqrt{-7 + 12|r| - 4r^2} \right) \quad 1 \le |r| \le 2,$$

$$0 \quad |r| > 2.$$
(13)

$$|r| > 2. \tag{14}$$

3.1 Summary of the IB-LBM algorithm

The implementation of the iterative IB-LBM algorithm is summarized as follows:

- 1. Initialize the computation parameters;
- 2. Stream the distribution function to obtain g_i ;
- 3. Compute the macroscopic variables: density ρ and the uncorrected velocity u using

$$\boldsymbol{\rho} = \sum_{i} g_{i}, \quad \boldsymbol{u} = \frac{1}{\rho} \sum_{i} \boldsymbol{e}_{i} g_{i}; \tag{15}$$

- 4. Set iteration counter m to 0;
- 5. Interpolate the immersed boundary velocity \boldsymbol{U}_{ib}^m using equation (10);
- 6. Compute the Lagrangian force density $\boldsymbol{F}_{ib}^{m}(s,t)$ using equation (9);
- 7. Spread $F_{ib}^m(s,t)$ to the Eulerian lattice to obtain $f^m(x)$;
- 8. Correct Eulerian velocity near to the immersed boundary according to

$$\boldsymbol{u}^{m+1}(\boldsymbol{x}) = \boldsymbol{u}^m(\boldsymbol{x}) + \frac{\boldsymbol{f}^m(\boldsymbol{x})dt}{2\rho(\boldsymbol{x})};$$
(16)



Figure 1: Side and front views of the stenosis geometry.

- 9. Update iteration counter m = m + 1;
- 10. Repeat steps 5-9 until the immersed boundary velocity \boldsymbol{U}_{ib}^m converges to the velocity of the tube wall $\boldsymbol{U}(s,t)$;
- 11. Calculate g_i^{eq} using equation (4);
- 12. Perform the collision step with the total Eulerian body force:

$$\boldsymbol{f}(\boldsymbol{x}) = \sum_{m=1}^{m_{max}} \boldsymbol{f}^m(\boldsymbol{x}); \tag{17}$$

13. Go to step 2 for next time-step.

Open Multi-Processing (OpenMP) parallel computing strategy has been incorporated into the code to accelerate the computation. The computations are performed on a workstation of Intel Xeon CPU E5-2650 2.3GHz.

4 RESULTS

The fluid flow through a 75% area reduction stenosis is simulated. The stenosed tube is generated by a cosine function of the axial coordinate x:

$$S(x) = \frac{1}{2}D[1 - s_o(1 + \cos(2\pi(x - x_o)/L))], y = S(x)\cos\theta, \quad z = S(x)\sin\theta,$$
(18)

where *D* is the diameter of the non-constricted tube, L = 2D is the length of the stenosis, for a 75% area reduction stenosis $s_o = 0.25$ and x_o is the central location of the stenosis. Figure 1 shows the side and front view of the stenosis. A steady Hagen-Poiseuille velocity profile is imposed at the inlet, and the initial flow field is initialized as:

$$\frac{u}{U_0} = 2\left(1 - r^2\right), \quad \frac{v}{U_0} = 0, \quad \frac{w}{U_0} = 0, \tag{19}$$

where *u*, *v* and *w* are the velocity in *x*-, *y*- and *z*-directions, respectively, U_0 is the averaged velocity at the inlet and $r = \sqrt{y^2 + z^2}$ is the radial distance from the tube centreline. The computational domain is a rectangular parallelopiped ($x \in [-3D, 17D]$, $y \in [-0.6D, 0.6D]$ and $z \in [-0.6D, 0.6D]$). The grid size of the fluid and the tube are 0.02D and 0.01D, respectively. The Reynolds number Re = 500 is defined



Figure 2: Axial velocity profiles at five axial locations (x = 1D, x = 2.5D, x = 4D, x = 5D, and x = 6D) downstream of the stenosis. Red solid: present IB-LBM; black circle: direct numerical simulation by Varghese et al. [6]. The tube wall is extended outwards by one grid point to compensate the diffusion of the boundaries caused by the diffused interface IBM.

based on the diameter D and the averaged velocity U_0 at the inlet. The tube wall is rigid with the no-slip boundary condition and a constant pressure is specified at the outlet. The tube wall is extended outwards by one grid point to compensate the diffusion of the boundaries caused by the diffused interface IBM.

As shown in figure 2, the axial velocity profiles at five axial locations downstream of the stenosis shows good agreement with the direct numerical simulation results by Varghese et al. [6]. The predicted pressure, streamwise velocity and vorticity contours are shown in figure 3. The flow is laminar and symmetric with a jet formed downstream of the stenosis. Figure 3(a) shows that the velocity increase significantly due to the constriction of the tube wall, which results in a low pressure area is observed at the most constricted part of the stenosis shown in figure 3(b). The vorticity contours in figure 3(c) show that the flow is symmetric and there is no vortex shedding downstream of the stenosis.

5 CONCLUSIONS

A strategy for the application of diffused interface IB-LBM is firstly introduced in the simulation of stenosis. The diffused interface IB-LBM solver is able to accurately predict the velocity profile within the stenosis.

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(c) Streamwise vorticity contours.

Figure 3: Pressure, streamwise velocity and vorticity contours in the Z-X symmetry plane of the 75% area reduction stenosis. The velocity is non-dimensionalized by the inlet averaged velocity U_0 . The pressure here is a relative pressure to the outlet pressure and is non-dimensionalized by ρU_0^2 . The vorticity is non-dimensionalized by U_0/D .

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